

# Proceedings

## NTUA Wind Tunnel Technical Info Day

*These are the presentations of the 1<sup>st</sup> tWEET-IE Technical Info Day*

*The event was addressed to industry and commercial stakeholders interested in the NTUA WT facilities and services*

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### NTUA Wind Tunnel Facility – Capacity and Services

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**Athens, 26<sup>th</sup> June, 2024**

**Venue:** National Technical University of Athens, 2<sup>nd</sup> floor, ANYM Building

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# INVITATION

*In the framework of the Horizon-Europe TWEET-IE project,  
we are pleased to invite you to attend the NTUA Wind Tunnel Technical Info Day.  
We will be presenting our capacity for research and services  
through presentations of past and present case studies.*

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## **TWEET-IE project: Technical Info Day** **NTUA Wind Tunnel Facility – Capacity and Services**

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**Athens, 26<sup>th</sup> June, 2024**

**Venue:** National Technical University of Athens, ANYM Building

<https://maps.app.goo.gl/4sMZtr3nVKXts9bq8>

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# AGENDA

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**Wednesday June 26<sup>th</sup>, 2024**

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**10:00 Welcome and opening**

The TWEET-IE project (Prof. Demetri Bouris)

**10:15 Aerodynamics and Wind Energy**

Wind Tunnel Testing (Asst. Prof. Marinos Manolesos)

Computational Fluid and Structural Dynamics (Assoc. Prof. Vasilis Riziotis)

**11:15 Environmental Flows and Buildings**

Wind Tunnel Testing and Computational Fluid Dynamics (Prof. Demetri Bouris)

**11:45 Coffee Break**

**Discussion – Round Table**

**12:30 Tour of Wind Tunnel Facilities**

Demonstration of wind tunnel testing facility

**13:30 End of Event**



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of Athens



[www.tweet-ie.eu](http://www.tweet-ie.eu)



# Technical Info Day

**NTUA Wind Tunnel Facility – Capacity and Services**

Prof. Demetri Bouris

June 25<sup>th</sup> , NTUA, Athens

**TWEET-IE / Twin Wind tunnels for Energy and the Environment –  
Innovations and Excellence**

**HORIZON-WIDERA-2021-ACCESS-03-01 / PR# 101079125**



- **Twin Wind Tunnels for Energy and the Environment – Innovations and Excellence**
- **Call Topic :** HORIZON-WIDERA-2021-ACCESS-03-01
- **Type of action:** HORIZON Coordination and Support Actions  
i.e. not a research project ... but there is a research component
- **Duration:** 1 November 2022 - 31 October 2025 (36 months)
- **Budget:** 1 498 250.00 €

## Participants:

NTUA, National Technical University of Athens



TUM, Technical University of Munich



KIT, Karlsruhe Institute of Technology



POLIMI, Polytechnic Institute of Milano



TU Delft, Technical University of Delft



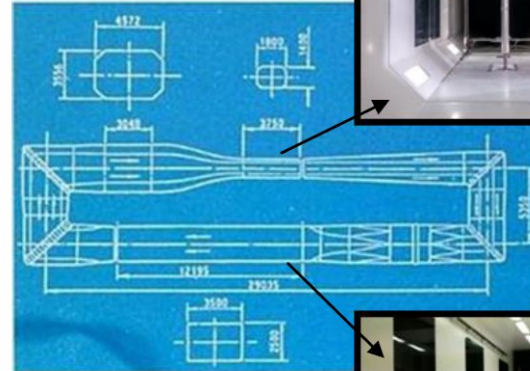
## Project goal :

to re-invent, mobilize and promote the NTUA WT

for EU level research

through shared experience

with top-class Leading Partners.



## **Pillar I. Widening of management and administration capabilities**

- Human and Material Resources
- Certification and Accreditation

## **Pillar II. Widening of technical capacity and competence**

- Physical modeling & measurement of vegetation in urban flows
- Phys. mod. & meas. of wind farms and wake effects.
- Phys. mod. & meas. of airfoil aerodynamic enhancement devices.
- Large scale PIV measurement technology.

## **Pillar III. Widening of outreach, visibility and profile**

- Improving the profile with respect to EU Research needs
- Improving the profile with respect to Services towards Industry

- ✓ SWOT analysis and definition of KPIs for NTUA WT facility assessment
- ✓ Facility Assessment Reports
- ✓ Inauguration of NTUA WT administration and management unit
- Sustainability plan of NTUA WT facility
- ✓ Web page and social media presence (LinkedIn, Facebook, Instagram)
- ✓ Summer schools
- ✓ Research activities
  - ✓ Effects of vegetation on flows in the urban environment
  - ✓ Wake interactions of a cluster of turbines and wake steering techniques
    - Micro devices for enhanced performance of airfoil sections
    - Scale effects in Urban Flows

Definition of KPIs related to:

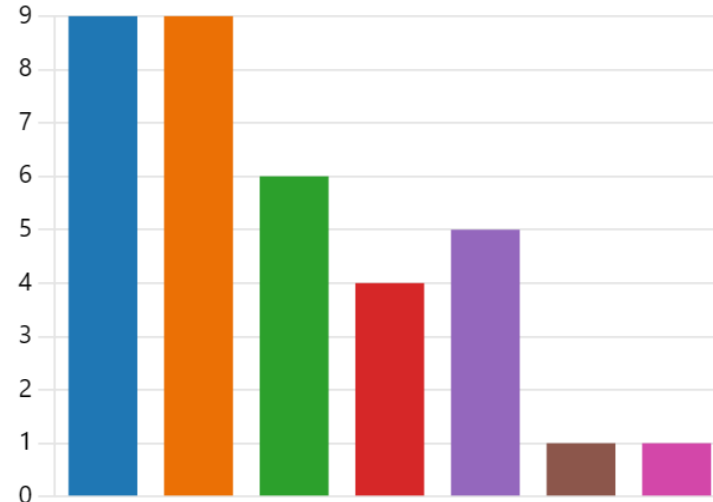
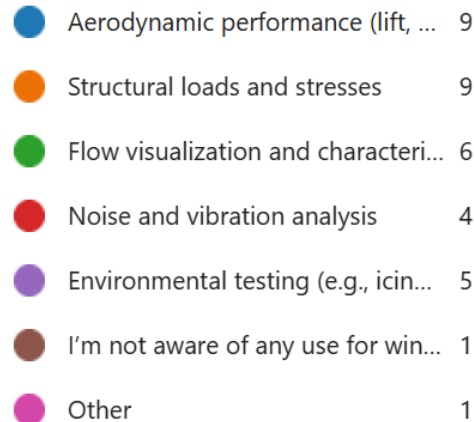
- educational/training activities
- research activities
- improving technical capacity
- improving management skills
- publicity and networking activities

Distinguished as:

- Short term – within the duration of the project
- Longer term – horizon of 2-5 years after the conclusion of the project

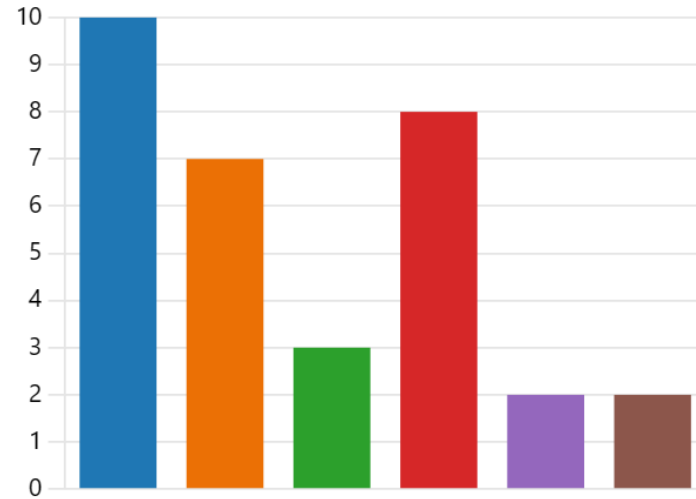
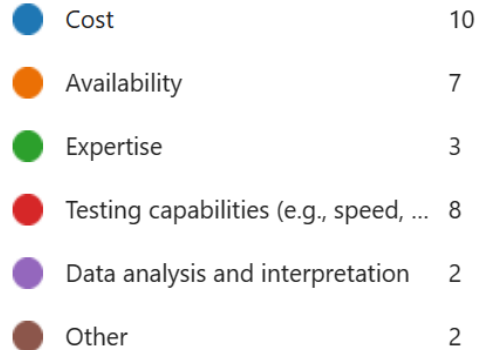
## *What types of wind tunnel testing would be most valuable to your industry?*

- *Aerodynamic performance (lift, drag, downforce, stability)*
- *Structural loads and stresses*
- *Flow visualization and characterization*
- *Noise and vibration analysis*
- *Environmental testing (e.g., icing, rain)*
- *I'm not aware of any use for wind tunnel testing in my industrial activities.*
- *Other (please specify)*



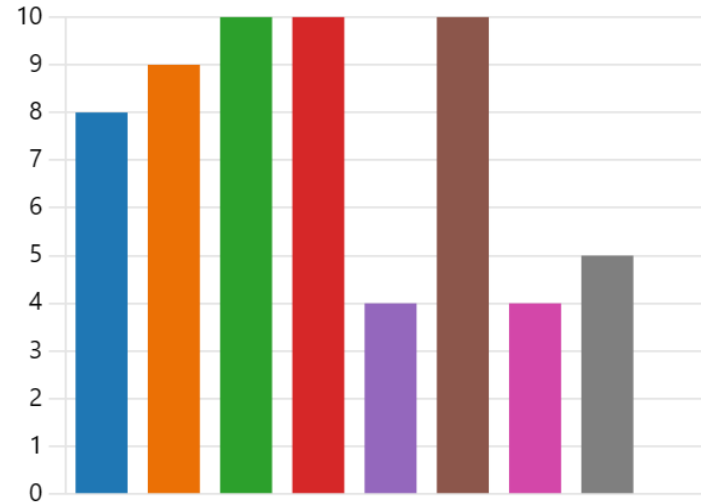


*What do you consider to be the main challenges with wind tunnel testing?*



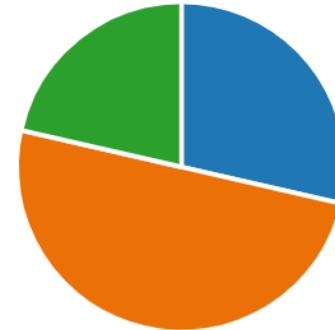
*What are the most important features you (would) look for in a wind tunnel testing facility?*

Specific capabilities (e.g., Reynol...	8
Range of test models/objects ac...	9
Measurement equipment, data ...	10
Expertise of staff	10
Customer service and support	4
Price and availability	10
Confidentiality	4
Communication and Responsive...	5
Other	0

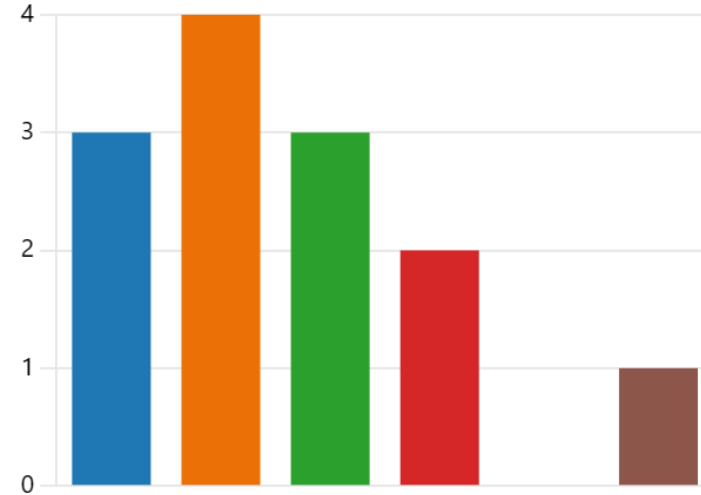


*What do you consider to be a typical project timeline for wind tunnel testing?*

● Less than 1 month	4
● 1-3 months	7
● 3-6 months	3
● More than 6 months	0



*What do you consider to be a typical budget for a wind tunnel test campaign?*



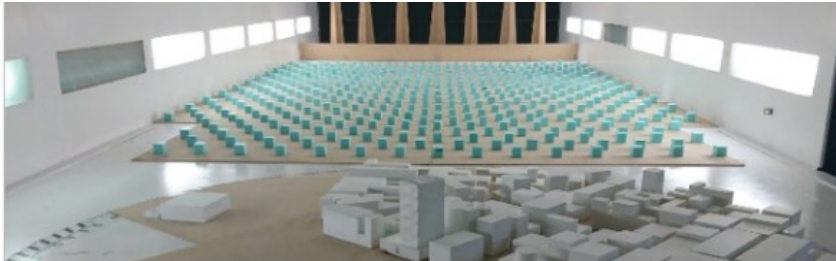
Session	Mon 15/07/2024	Tue 16/07/2024	Wed 17/07/2024	Thu 18/07/2024	Fri 19/07/2024
09:00 - 11:00	Introduction + Lab Tour	Atmospheric Boundary Layer modelling	Wind Turbine Model Design	Airfoil Testing	Hands-on data analysis
11:00 - 13:00	Wind Tunnel Measurement Techniques	Urban Flows	Wind Farm Testing	Wind Tunnel Session	Visit To Wind Farm
14:30 - 18:30	Wind Tunnel Measurement Techniques	Wind Tunnel Session	Wind Tunnel Session	Hands-on data analysis	Visit To Wind Farm



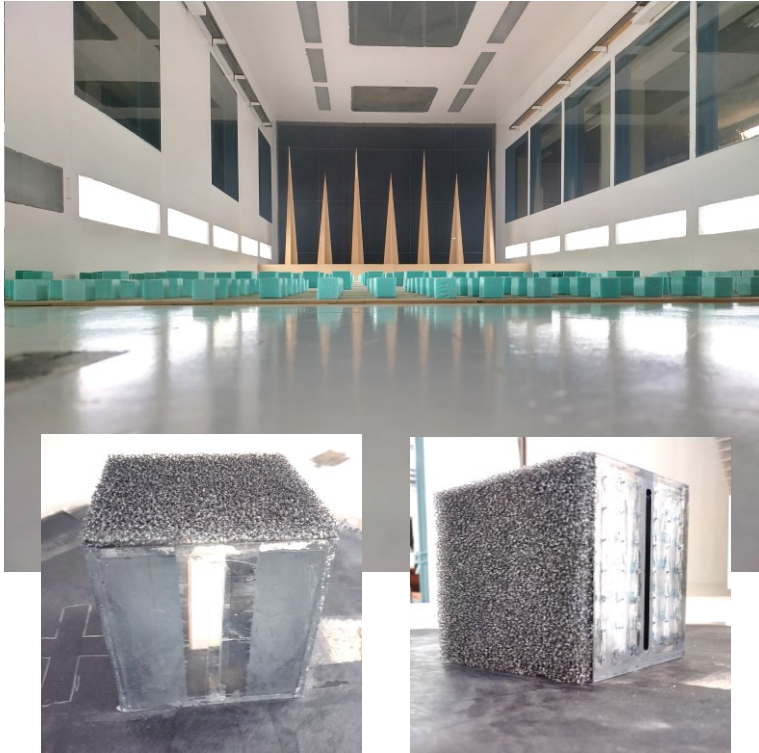
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**POLITECNICO**  
MILANO 1863



Effects of vegetation  
on flows in the urban environment



Wake interactions of a cluster of turbines  
and wake steering techniques



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Technical Info Day

## NTUA Wind Tunnel Facility – Capacity and Services



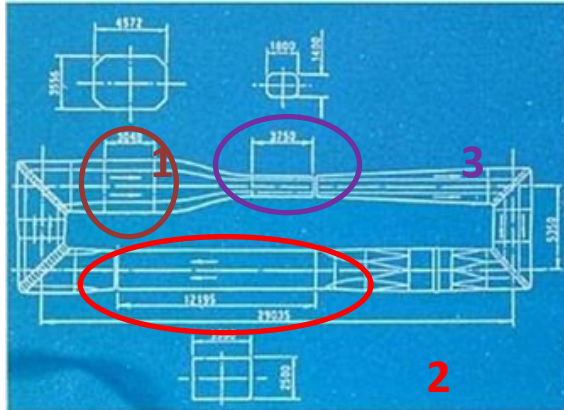
[www.wt.fluid.mech.ntua.gr](http://www.wt.fluid.mech.ntua.gr)



windtunnel.ntua



- Closed circuit
- 7 blade axial fan of 350 hp.
- Subsonic,  
max. speed is 60m/s and  
turbulence level is 0.2%



Total Wind Tunnel Length is 32 m  
3 test sections

Section	Width (m)	Height (m)	Max. speed (m/s)
1	4.5	3.5	9.5
2	3.5	2.5	17.0
3	1.8	1.4	60.0

- Section 1 (**4.5x3.5m**) suitable for small propellers, wind turbine rotors and fans
- Section 2 (**3.5 x 2.5 x 12 m**) suitable for buildings, bridges, wind turbine siting, pollutant dispersion and large model wake studies.  
It is equipped with remotely controlled turn table floor and 3D motor driven traversing mechanism.  
It's the atmospheric boundary layer test section
- Section 3 (**1.8x1.4**) for airfoils, aircraft wings, fuselages, model aircrafts, light and heavy vehicles.  
It's the high speed test section ( up to ~50 m/s)

3 bladed HAWT model (D=2m) in section 1



City model in section 2



Aircraft model in section 3



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# NTUA Wind Tunnel

## Aerodynamics and Wind Energy Applications







## Equipment


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- Force Balances
- Pressure Scanners
- Hot Wire Anemometry
- Stereo Particle Image Velocimetry

✓ **Workshop**

✓ **Prefit Area**





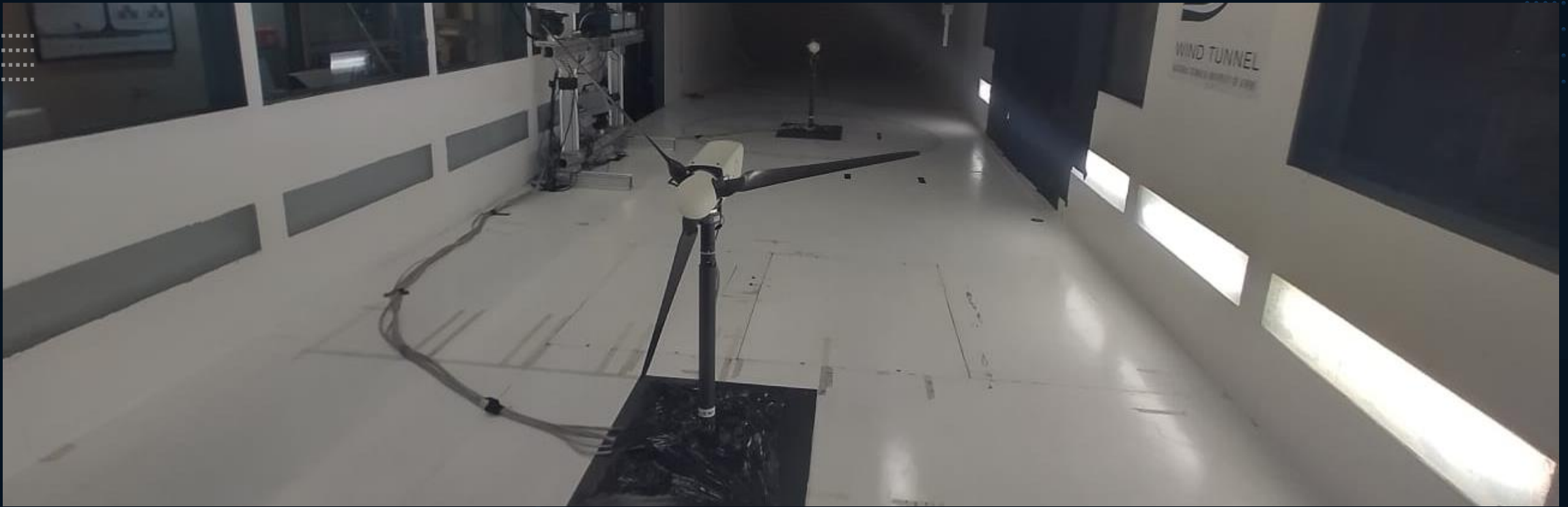
## Large Test Section Applications

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- Small Wind Turbines
- Wind Farm Modelling
- Sports (Cycling)
- Component Testing  
(Tents, Umbrellas, Parachutes)





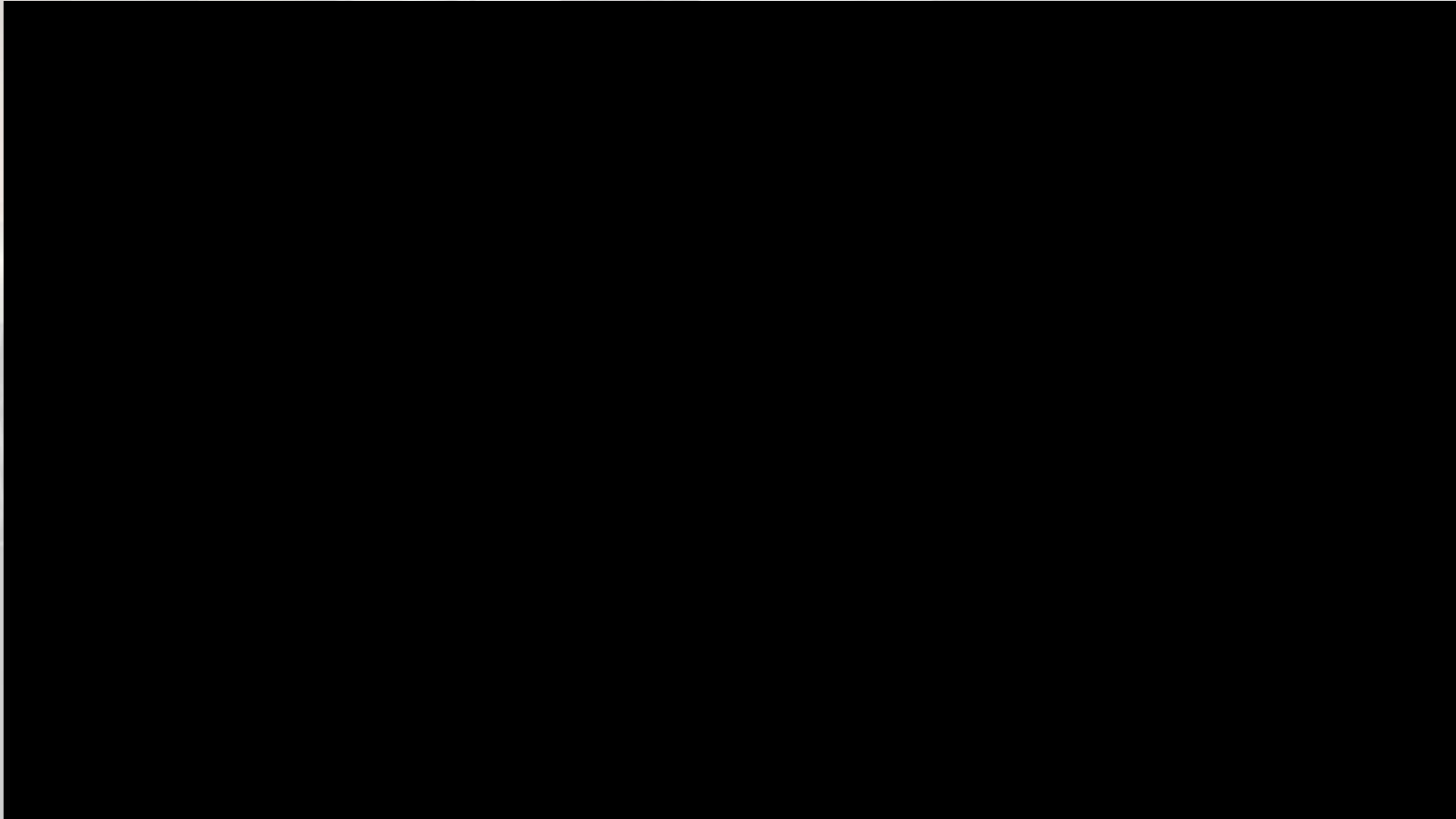


# Horizontal Axis Wind Turbines

- Power
- Loads
- Wake / Blade Aerodynamics
- Component Testing
- Up to  $D=1.1$  m



From two weeks ago..



# Wind Farm modelling

- Perforated disk models
- Different levels of inflow turbulence



# Cycling

- Power
- Drag
- Aerodynamics







## Wind Turbine Nacelle Testing

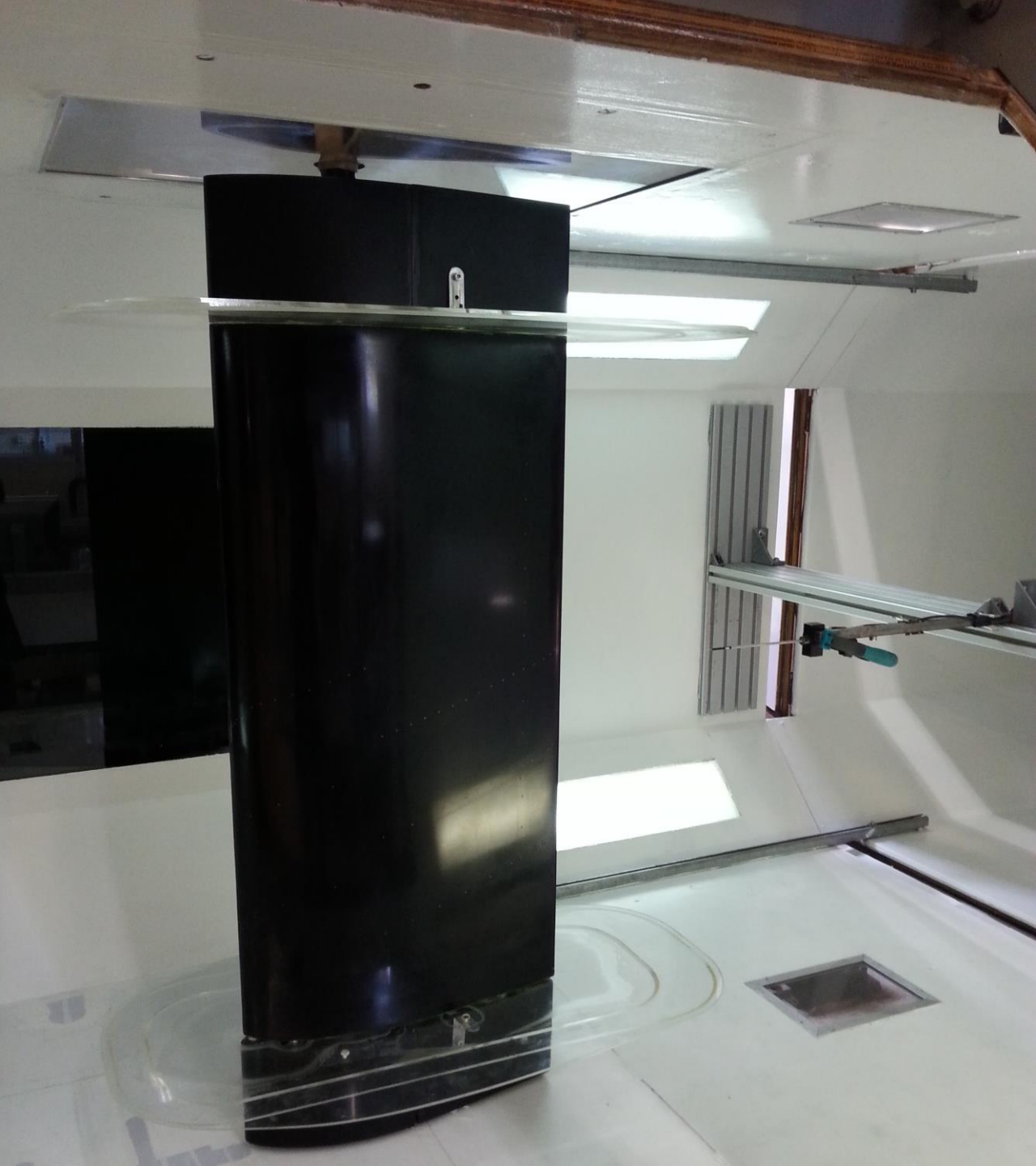
- Forces
- Vibrations
- Aerodynamics

## Small Test Section Applications

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- Airfoil Testing
- Gust Generation
- Flow Control Devices
- Component testing





# Airfoil Testing

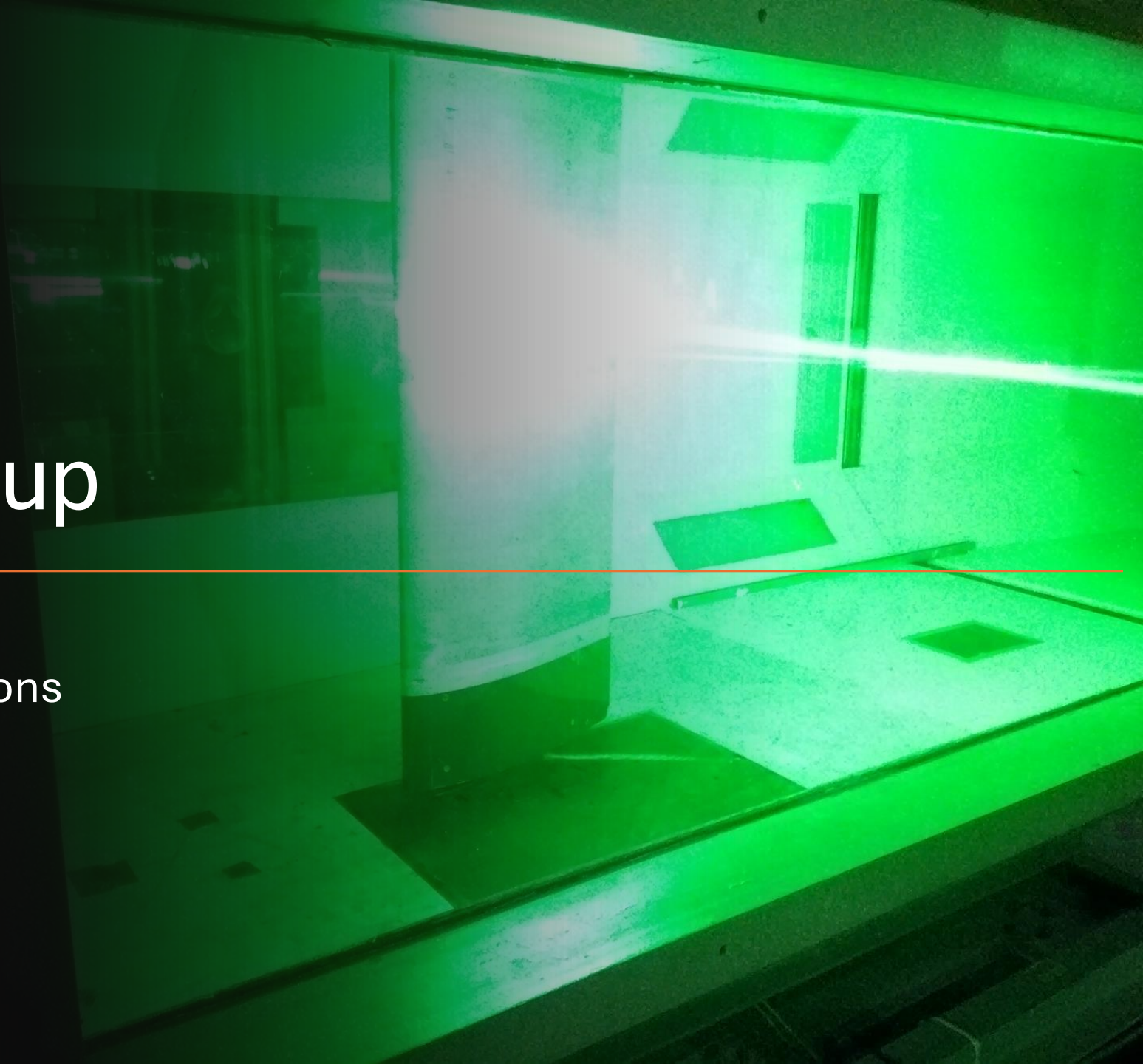
- Wind Turbine Airfoils
- Airfoil Performance
- Flow Control Devices
- Reynolds number up to 1.8M



# Aeroelastic Set up

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Models can pitch or plunge  
under free and/or control motions

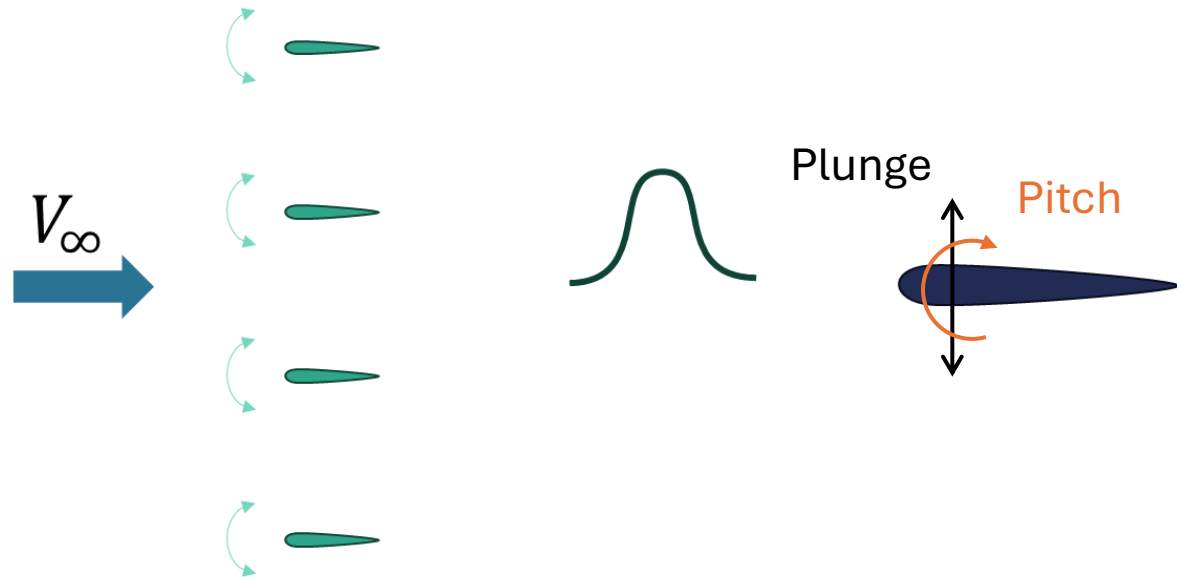


# Aeroelastic Set up



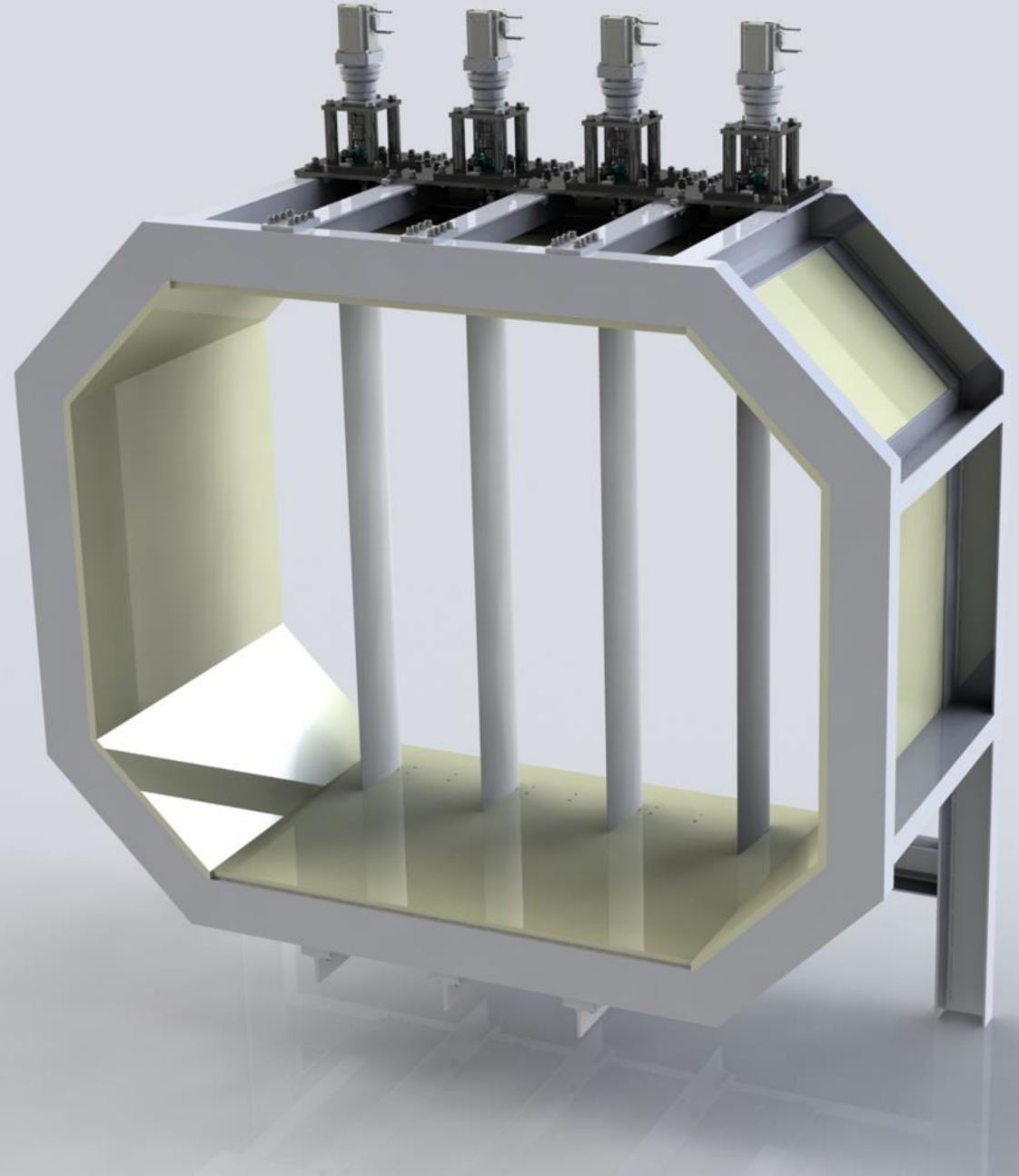


# Gust Generation



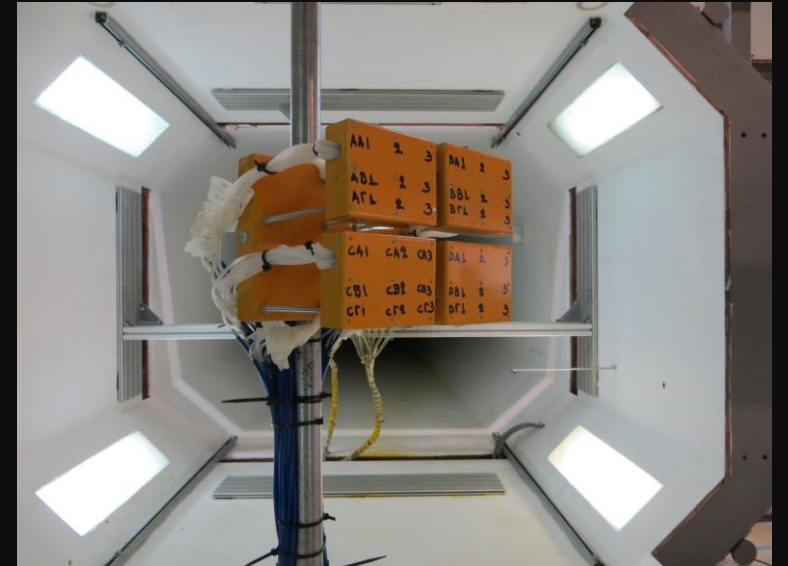
# Gust Generation

One of the few facilities in the world to achieve gusts and Reynolds numbers of this magnitude



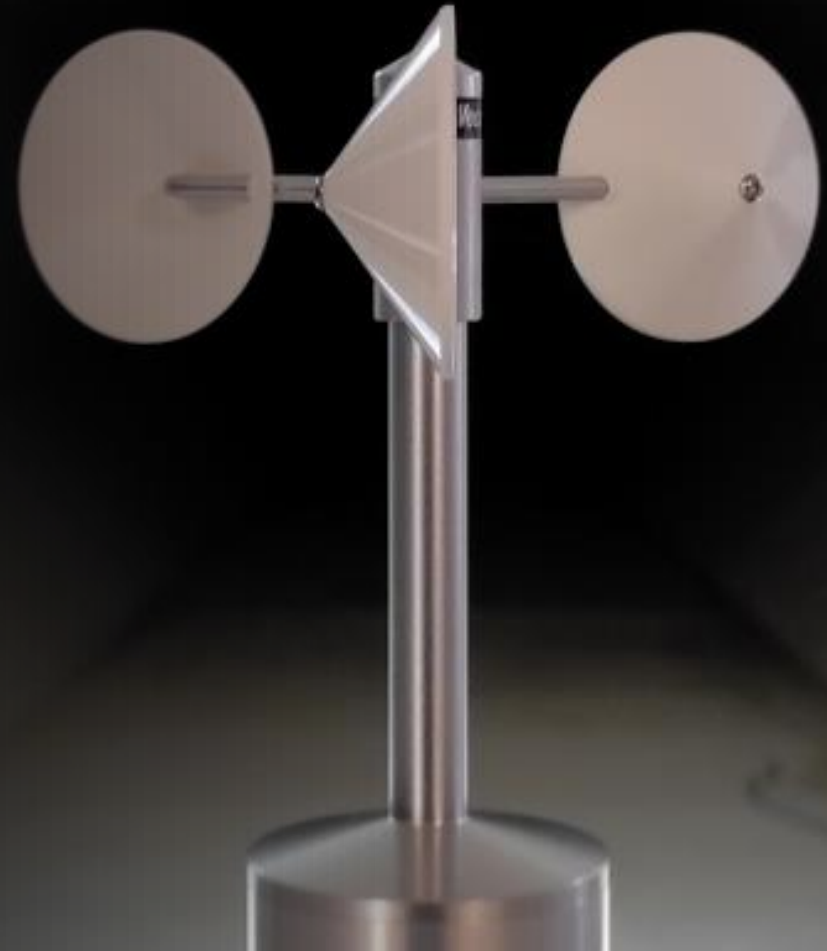
# Component testing

- Survivability
- Forces
- Sealing



# Anemometer Calibrations

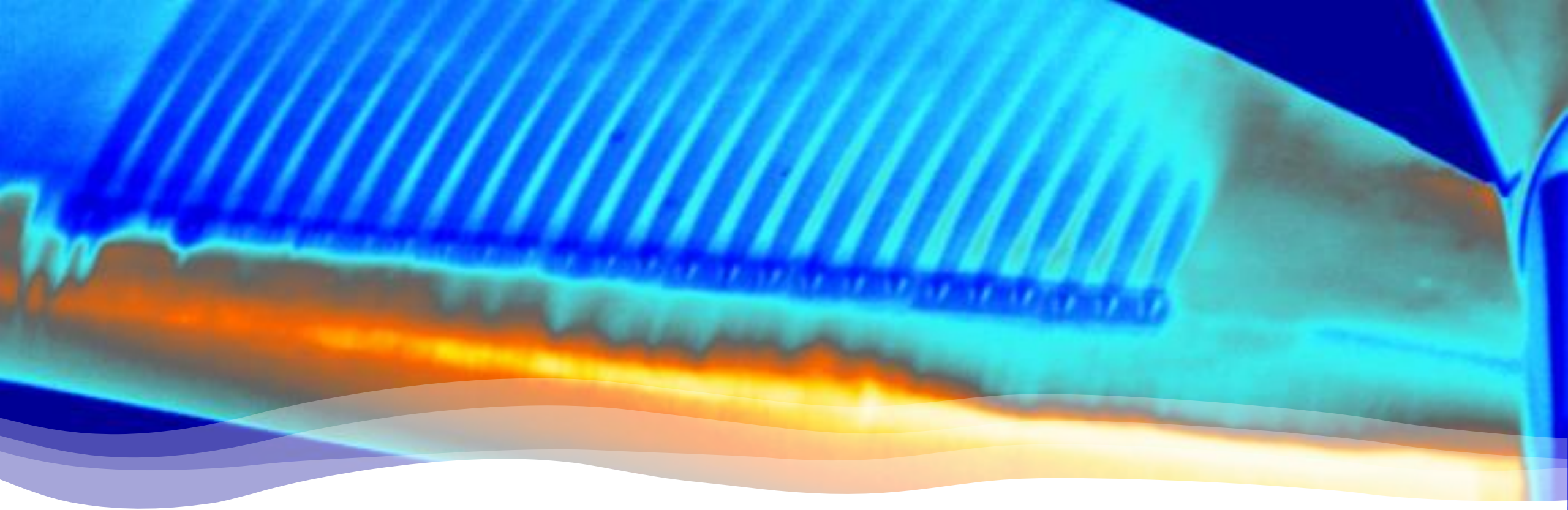
- International Wind Engineering
- MEASNET Certification



# Student Teams

- PROM Racing
- EUROAVIA
- Aiolos





Application:  
**Designing Flow Control Devices on Wind Turbines**



# The project:

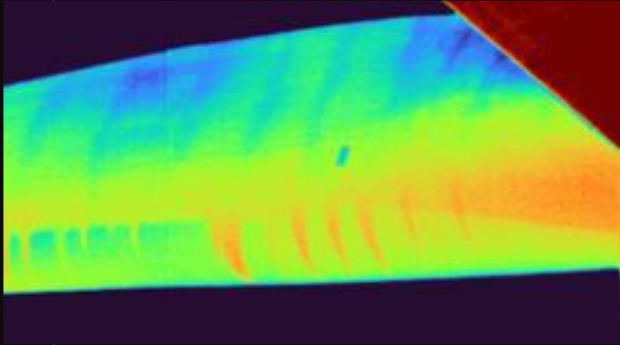
## Demonstration of Enhanced Vortex Generators

### Can we (commercially) design better VGs than the existing ones?

- Improved shape/configuration
- Improved positioning
- Blade specific
- Higher AEP

### Application on two turbines:

- 850kW Vestas V52 ( $D = 52\text{ m}$ )
- 7MW Levenmouth Demonstration Turbine ( $D = 171\text{ m}$ )



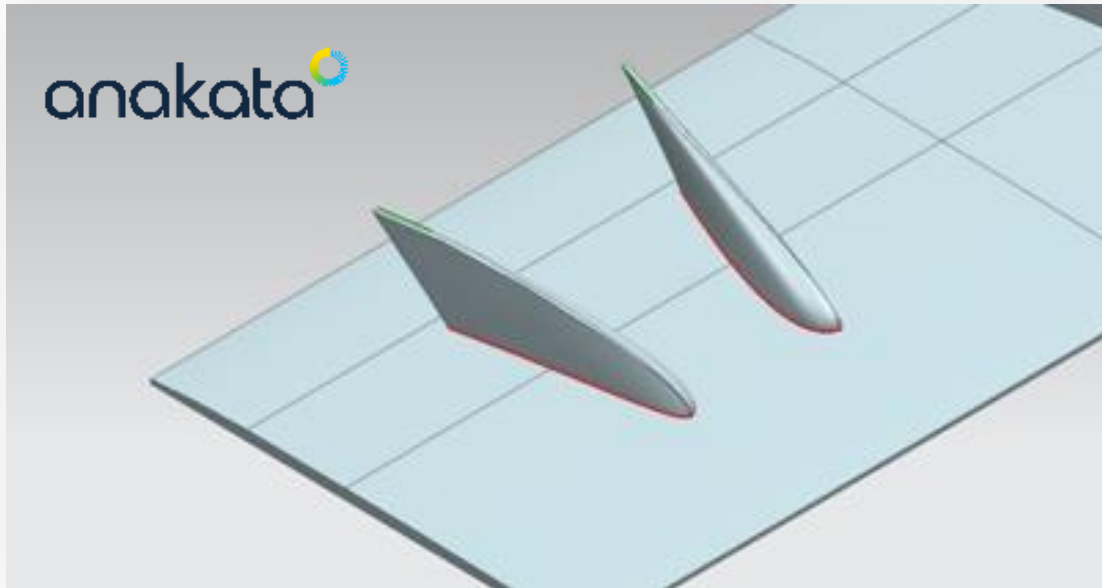
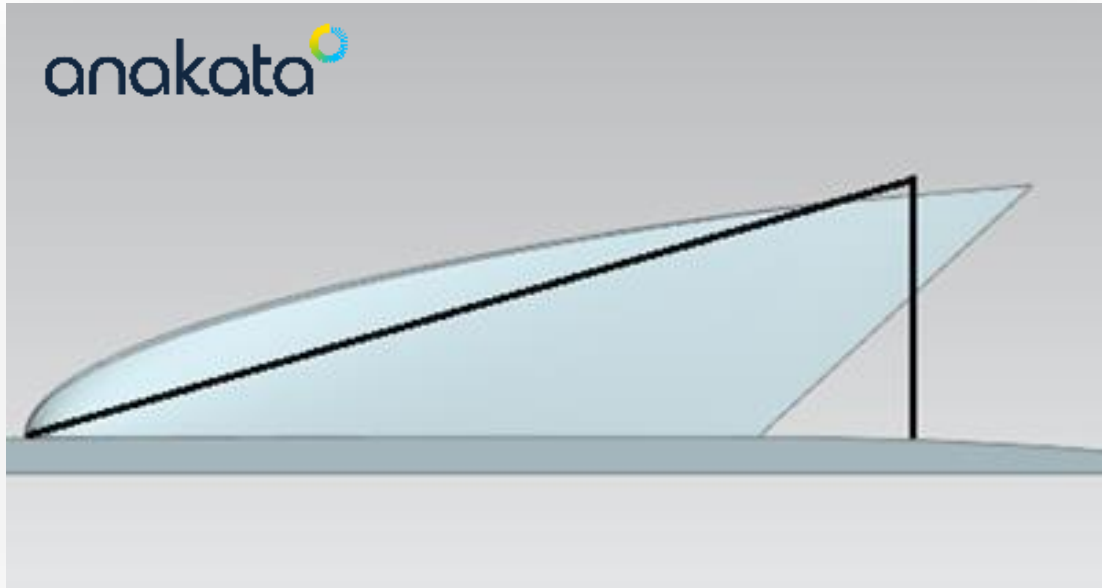
# What we did

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- Wind Tunnel Testing
- RANS CFD simulations
- BEM calculations
- Infrared Thermography
- Field tests



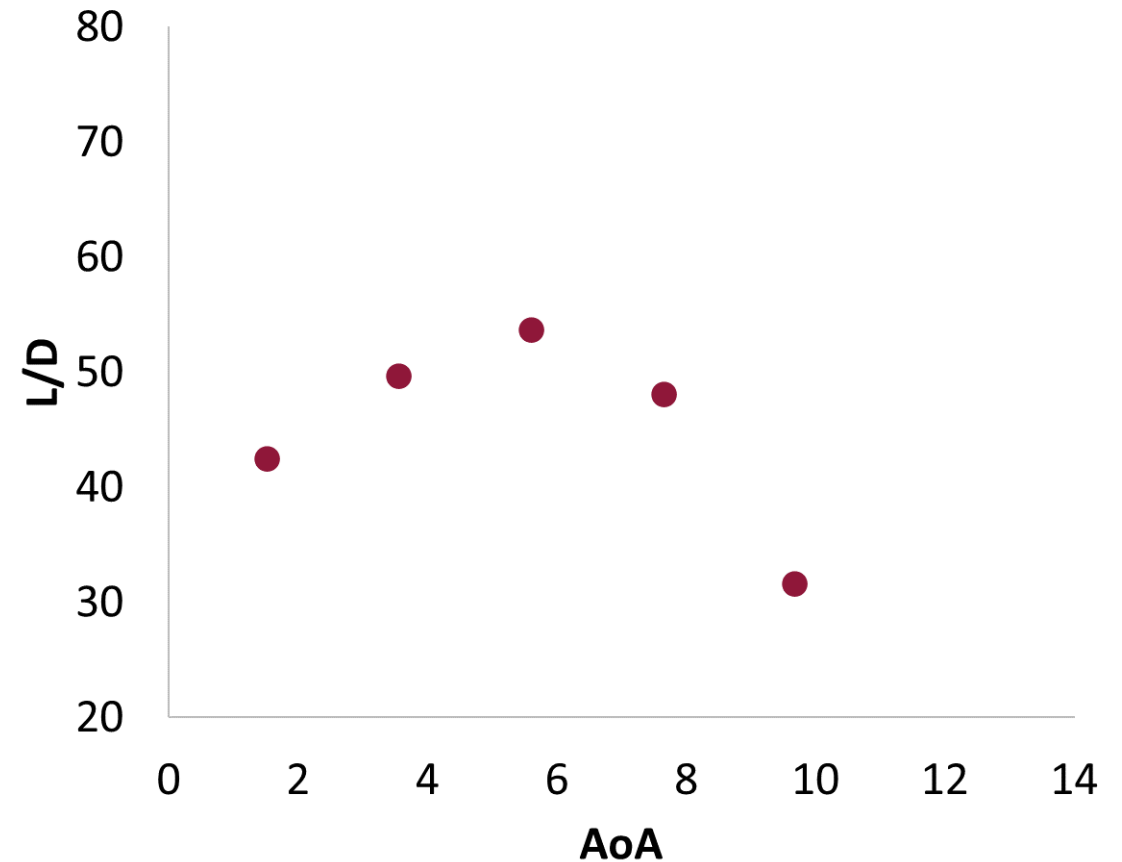
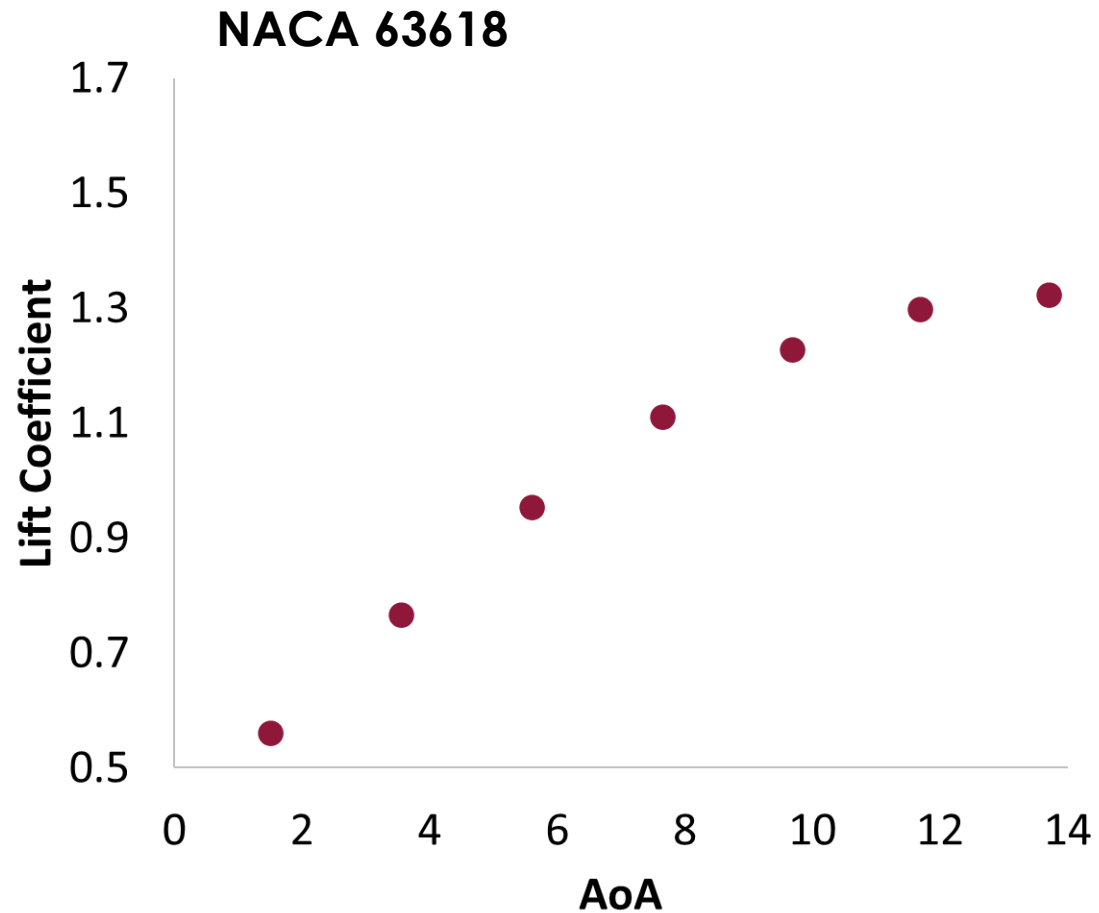
# VG Design



- A single VG design performed best for all airfoil profiles.
- Best performing shape
  - is 3D
  - has a cambered airfoil outline and
  - has variable twist
- $h_{VG} = 0.01c$
- Located at
  - $0.4c$  for the 18% thick airfoil
  - Both sides on the 35% thick airfoil

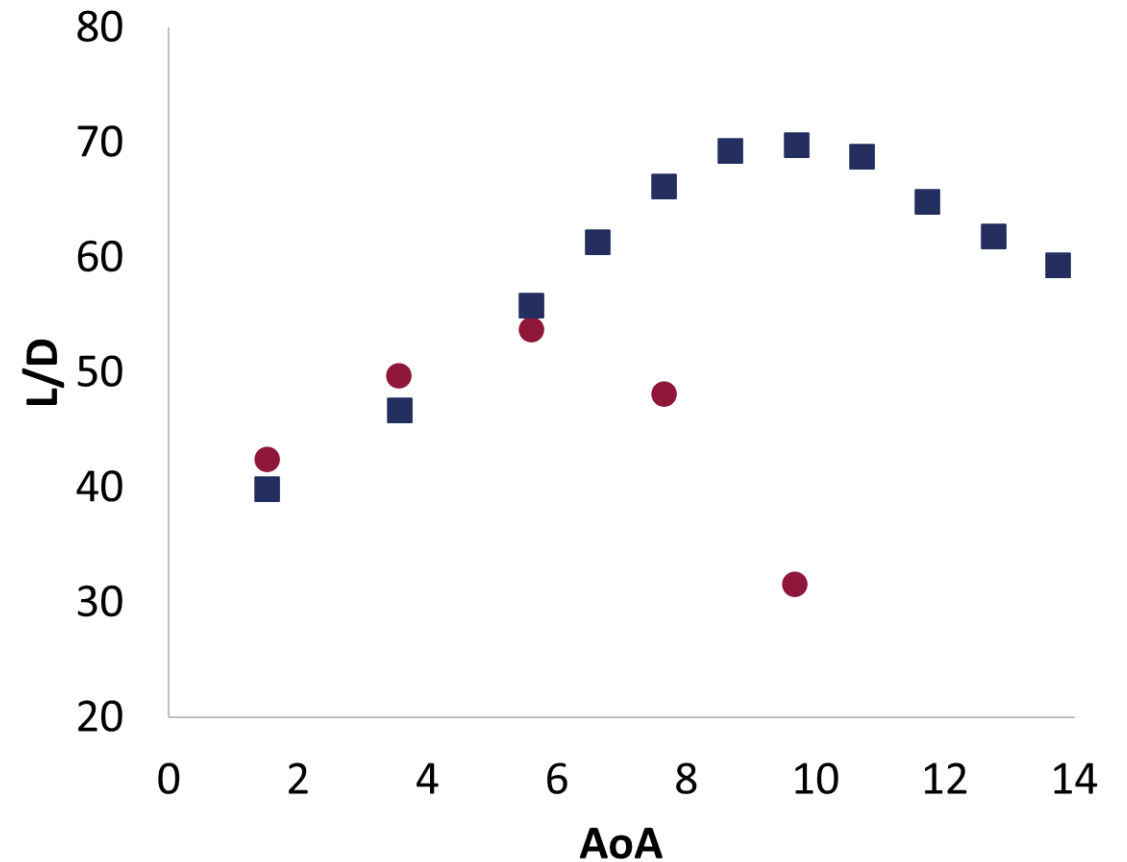
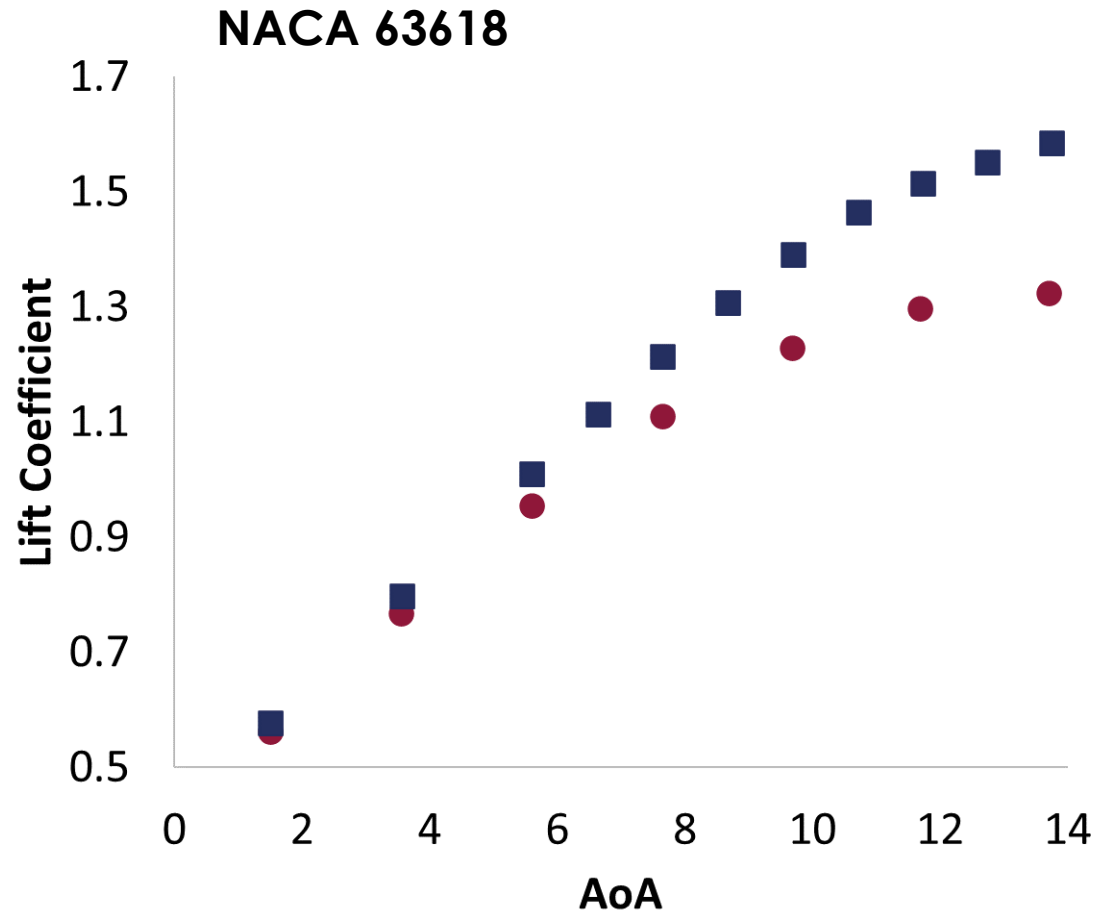
# Wind Tunnel results

● No VGs



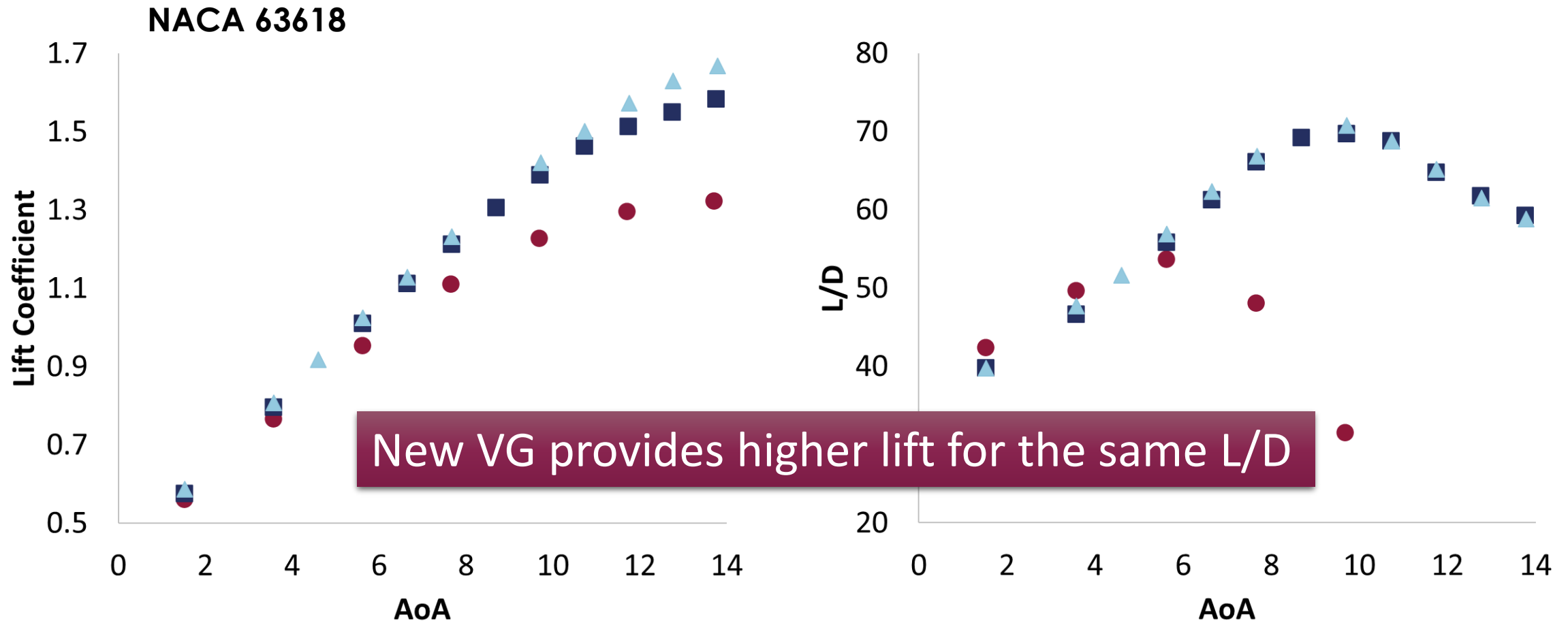
# Wind Tunnel results

- No VGs
- Vane Type VG



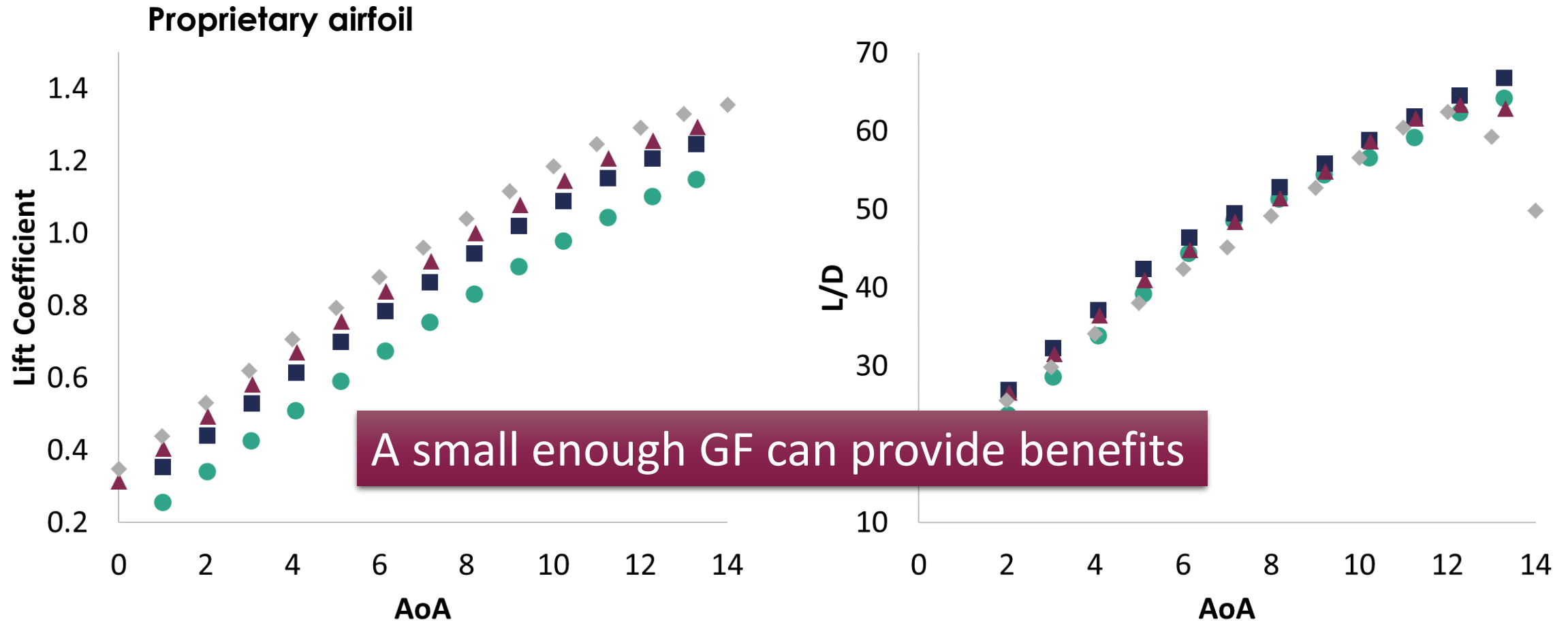
# Wind Tunnel results

- No VGs
- Vane Type VG
- ▲ Twisted 3D VG



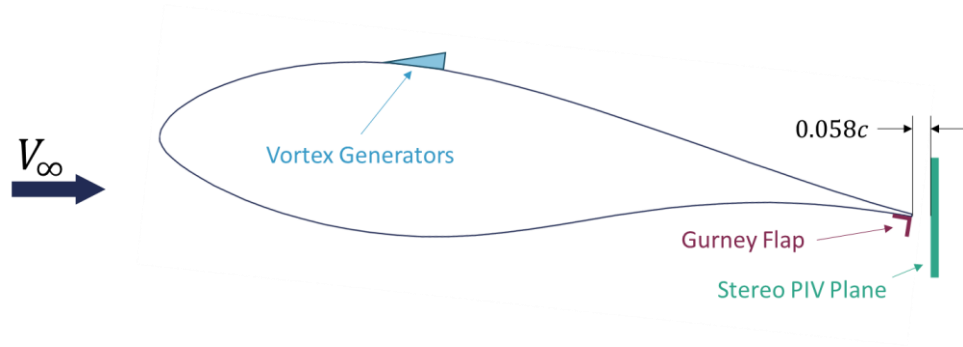
# Wind Tunnel results

- VGs only
- VGs + GF 0.004c
- ▲ VGs + GF 0.008c
- ◆ VGs + GF 0.012c

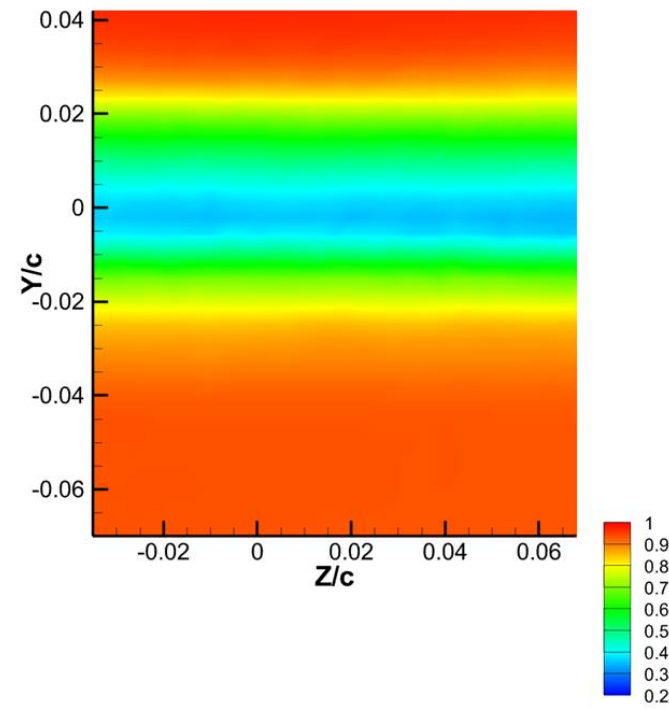


A small enough GF can provide benefits

# PIV



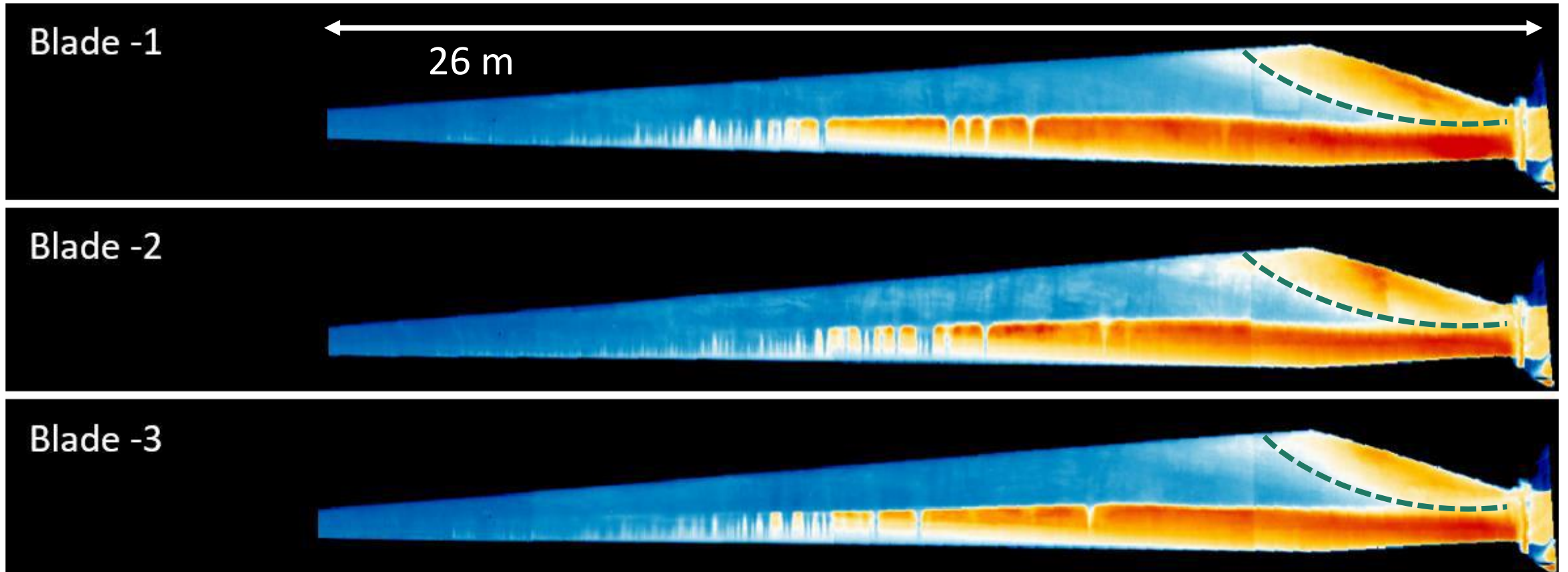
Baseline



# Wanstrow – Vestas V52 850kW

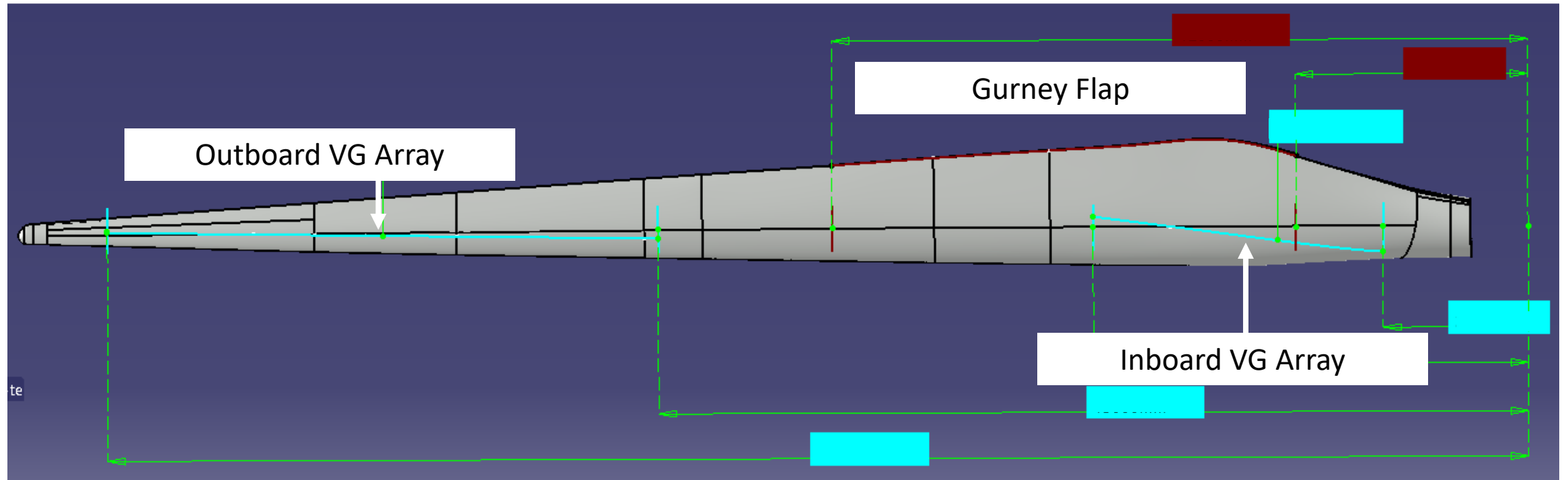


# Suction Side Features

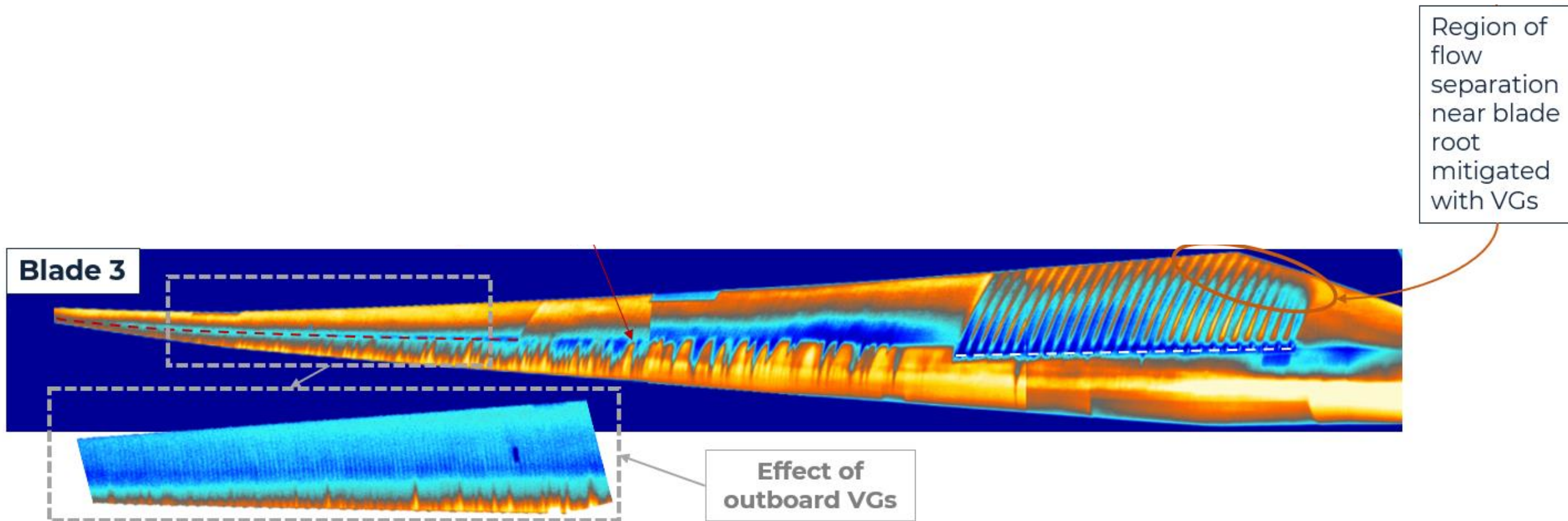




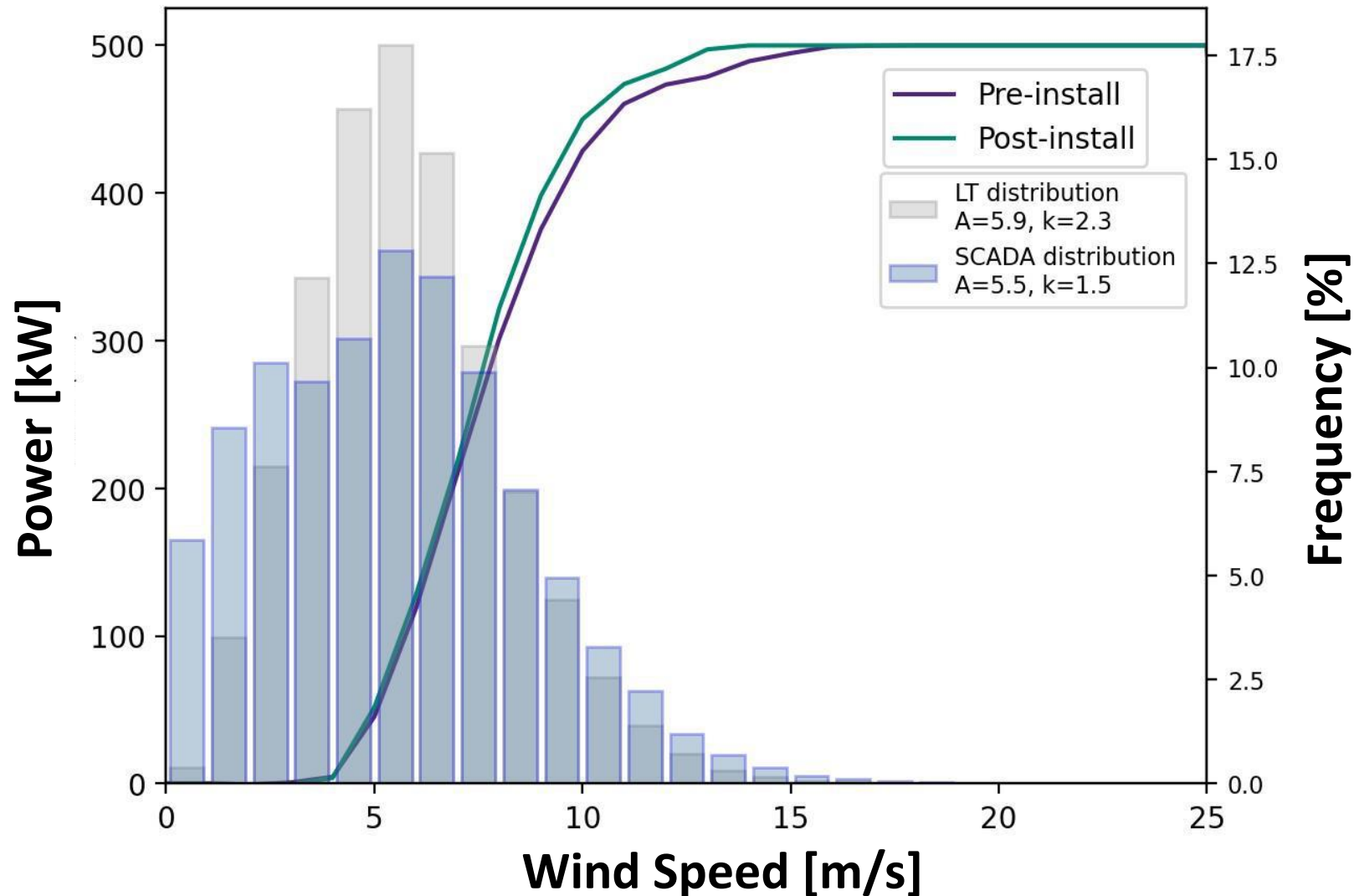
# V52 Flow control configuration



# Field Assessment



# Field Assessment **+5.4% to +5.8% AEP**

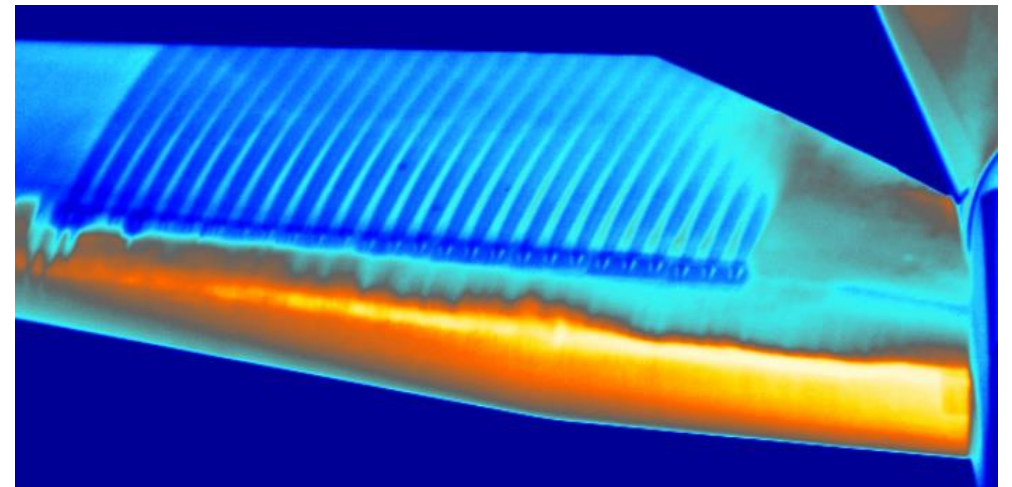
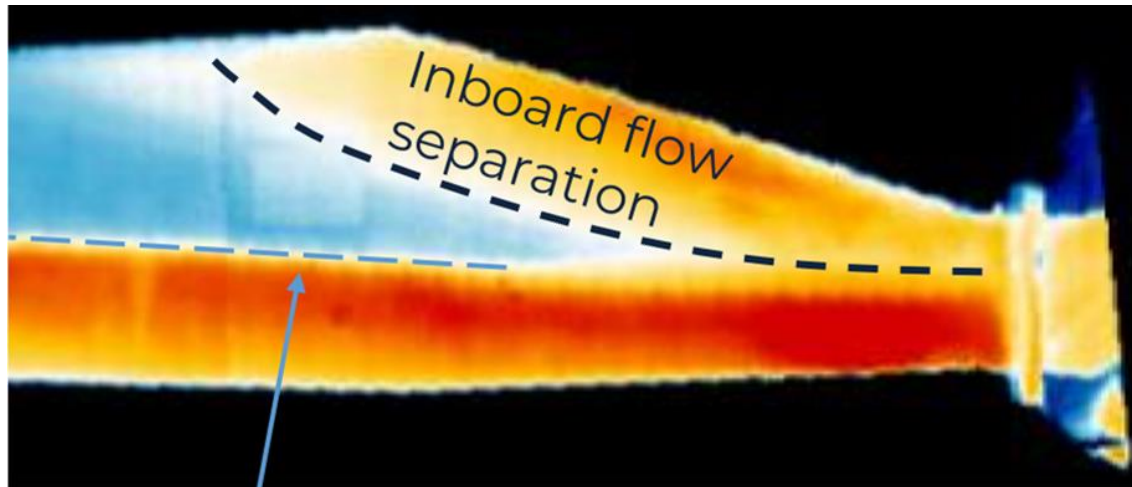
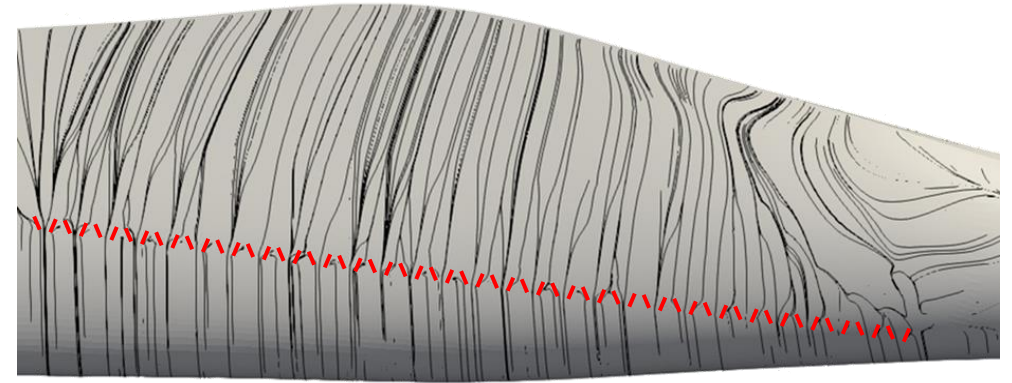


# Results – Retrofitting Flow Control Devices

**Before**



**After**





Thank you for  
your attention

Questions?

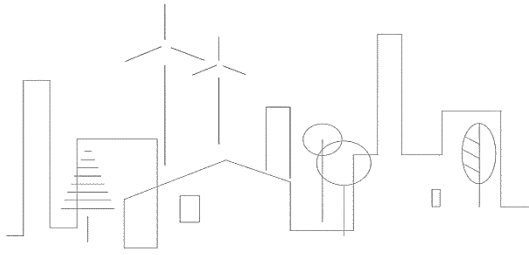
[marinos@fluid.mech.ntua.gr](mailto:marinos@fluid.mech.ntua.gr)



# References

1. Papadakis, G. and Manolesos, M.: The flow past a flatback airfoil with flow control devices: benchmarking numerical simulations against wind tunnel data, *Wind Energ. Sci.*, 5, 911–927, <https://doi.org/10.5194/wes-5-911-2020>, 2020.
2. Manolesos, M., Celik, Y., Ramsay, H., Karande, R., Wood, B., Dinwoodie, I., ... & Papadakis, G. (2024, June). Performance improvement of a Vestas V52 850kW wind turbine by retrofitting passive flow control devices. In *Journal of Physics: Conference Series* (Vol. 2767, No. 2, p. 022027). IOP Publishing.





# Aerodynamics and Wind Energy: Computational Fluid & Structural Dynamics

Vasilis Riziotis  
TWEET-IE Technical Event, 26 June 2024



Co-funded by the  
European Union

### Computational Aerodynamic & Fluid Structure Interaction

research activities related to Wind Energy applications:

- Aerodynamic and hydro-aero-elastic design/optimization and analysis of wind turbines
- Wind turbines noise emission and propagation
- Wind farms layout optimization and wake effects assessment

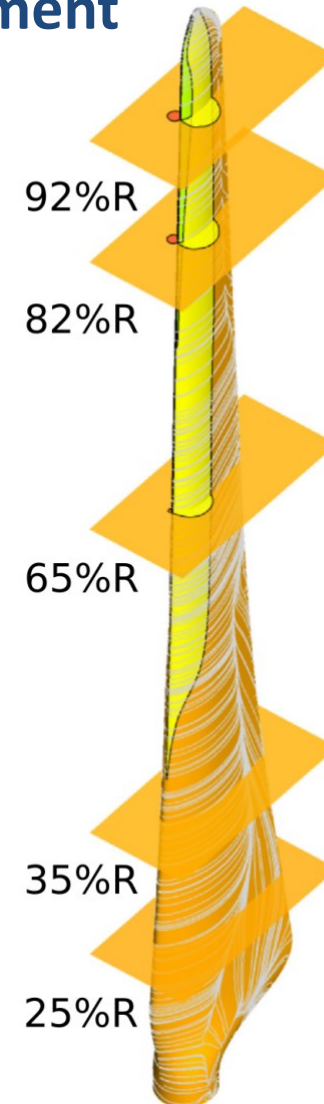
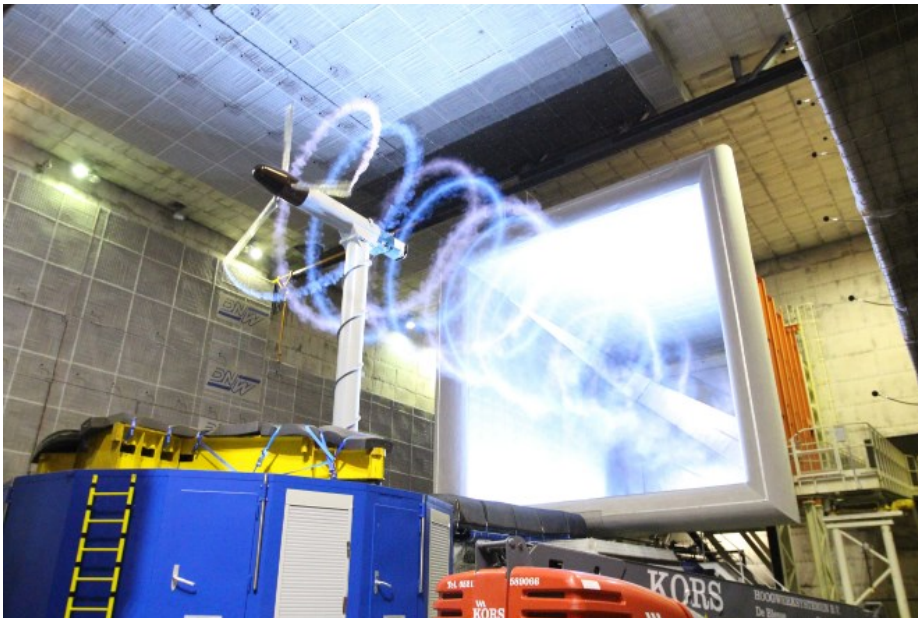
A wide range of in-house aerodynamics analysis tools have been developed over the years:

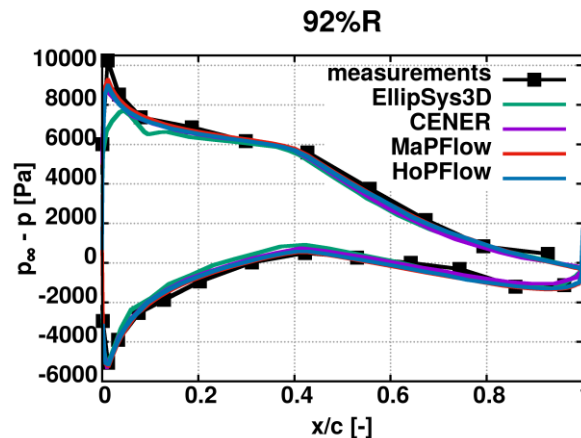
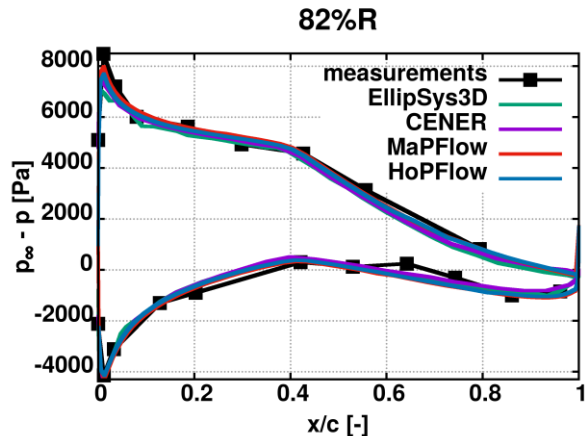
- Enhanced Blade Element Momentum (BEM) models
- Lifting line / lifting surface / panel free wake vortex models
- Actuator disk and line CFD URANS/DDES/LES models
- Fully resolved CFD URANS/DDES/LES models
- Hybrid CFD models (Eulerian – Lagrangian)

## Modelling of New MEXIXO experiment

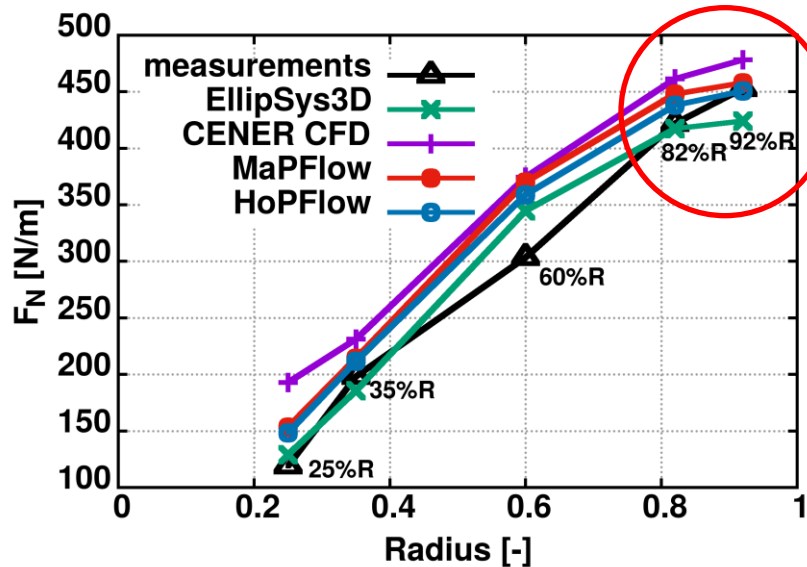
MEXICO 2012 → New MEXICO 2015

$U_{\infty}$	14.7 m/s	20-45%R	DU91-W2-250
$\Omega$	425 rpm	55-65%R	RISØ A1-21
Yaw, Tilt	0°	75-100%R	NACA 64418

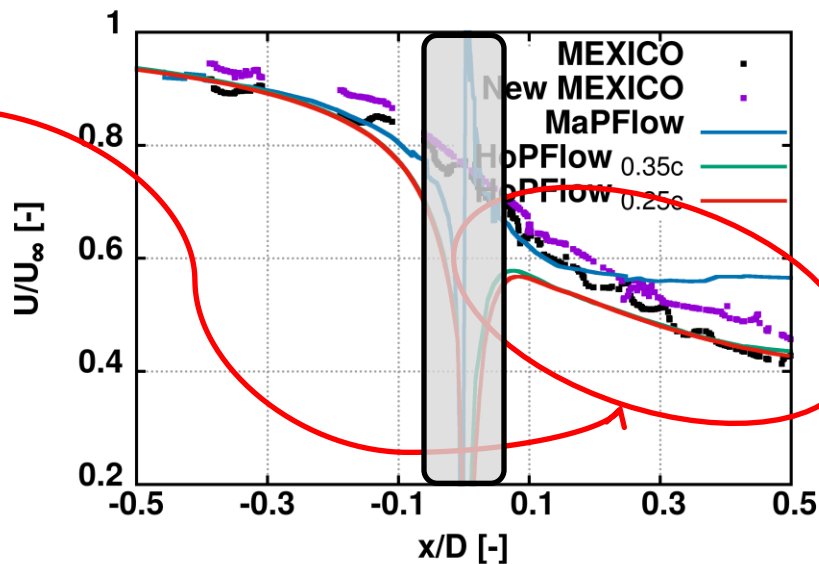




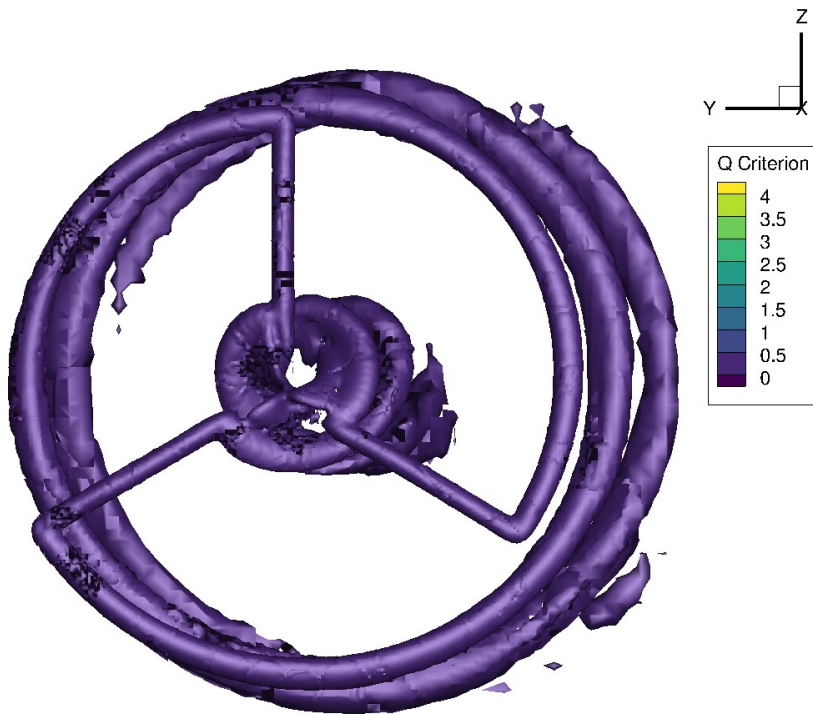
## Normal Force Radial Distribution



## 425 rpm. -- 14.7 m/s -- $r=1.8m$

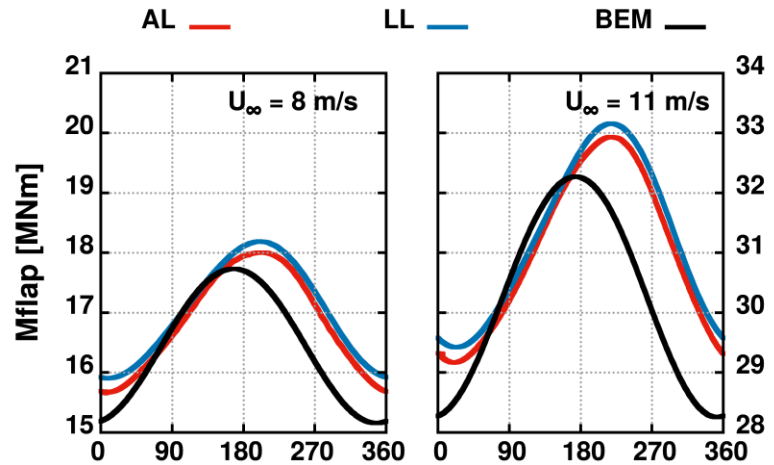


## Modelling of yawed flow conditions

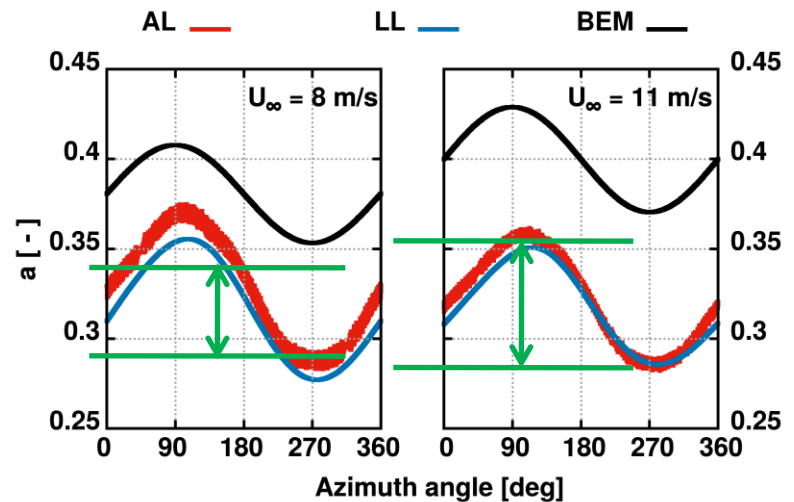


**Yaw 15°**

### Root Flapwise Moment

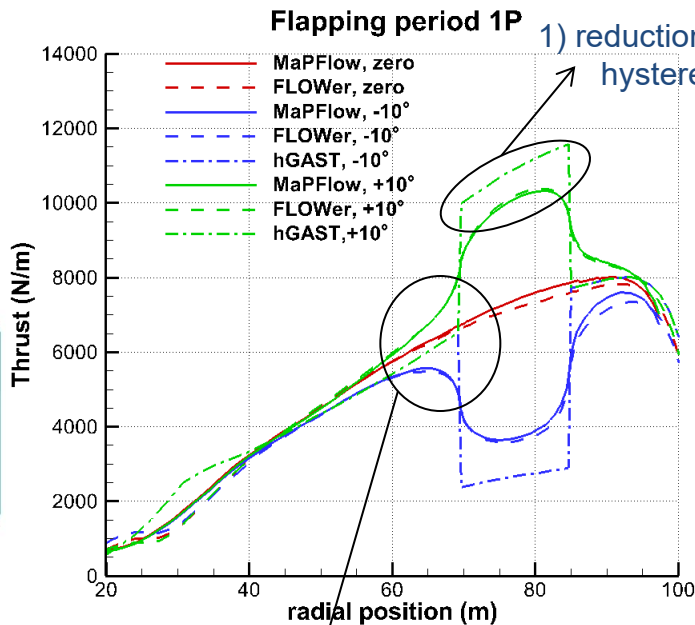
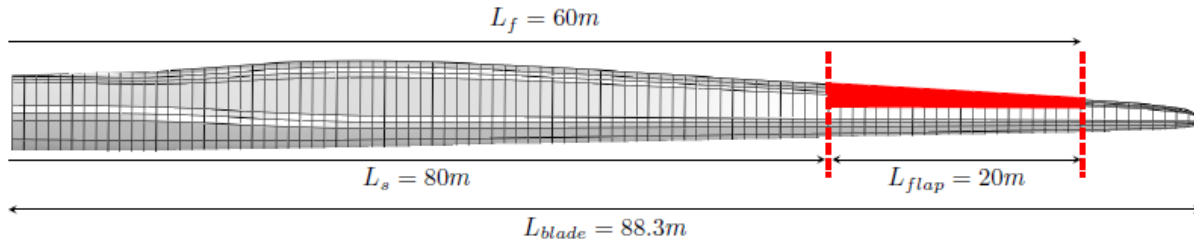


### Axial Induction Factor at 75%R

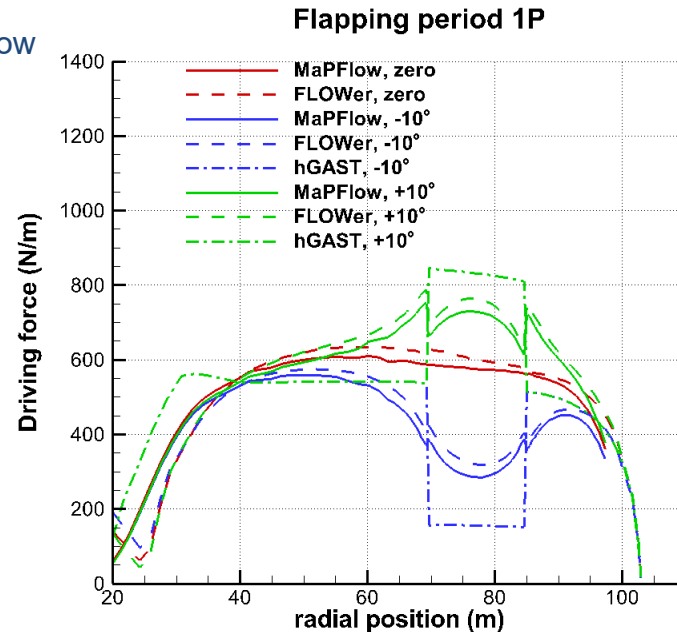




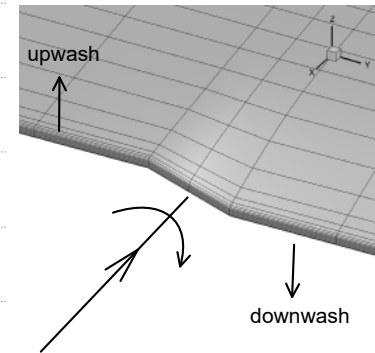
## 1P flap deflection - radial distribution of loads for $\pm 10^\circ$ flap deflection

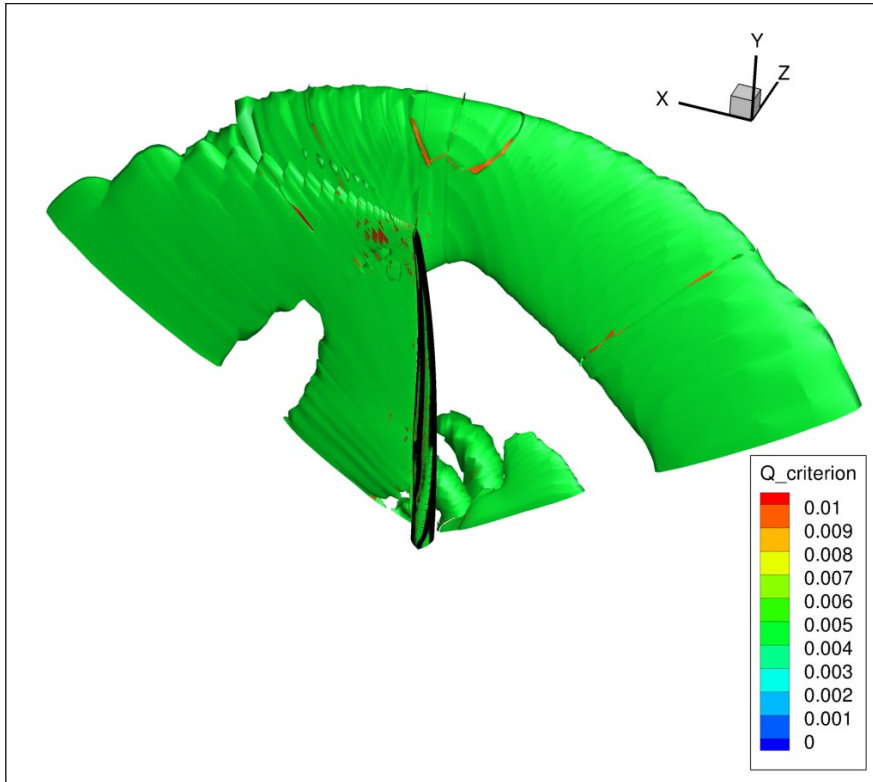


2) expansion of flap influence beyond the flap region

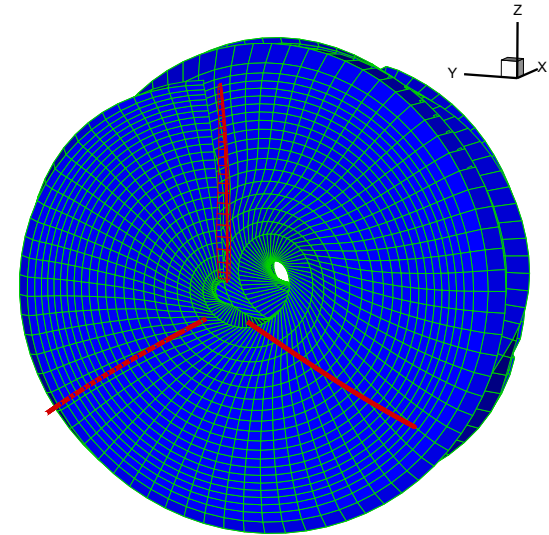


3) off-center appearance of maximum

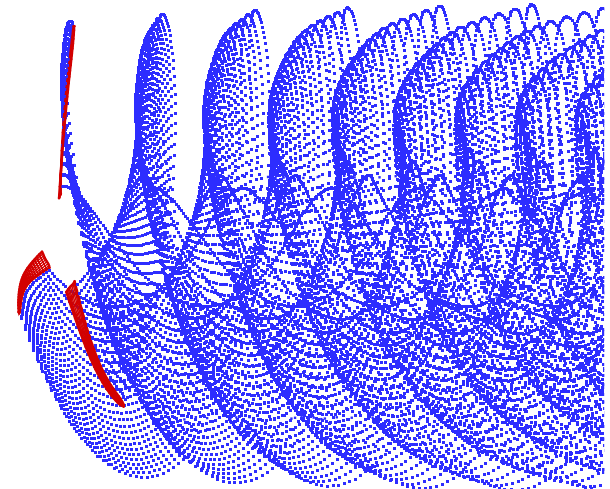




**3D CFD (Q criterion)**

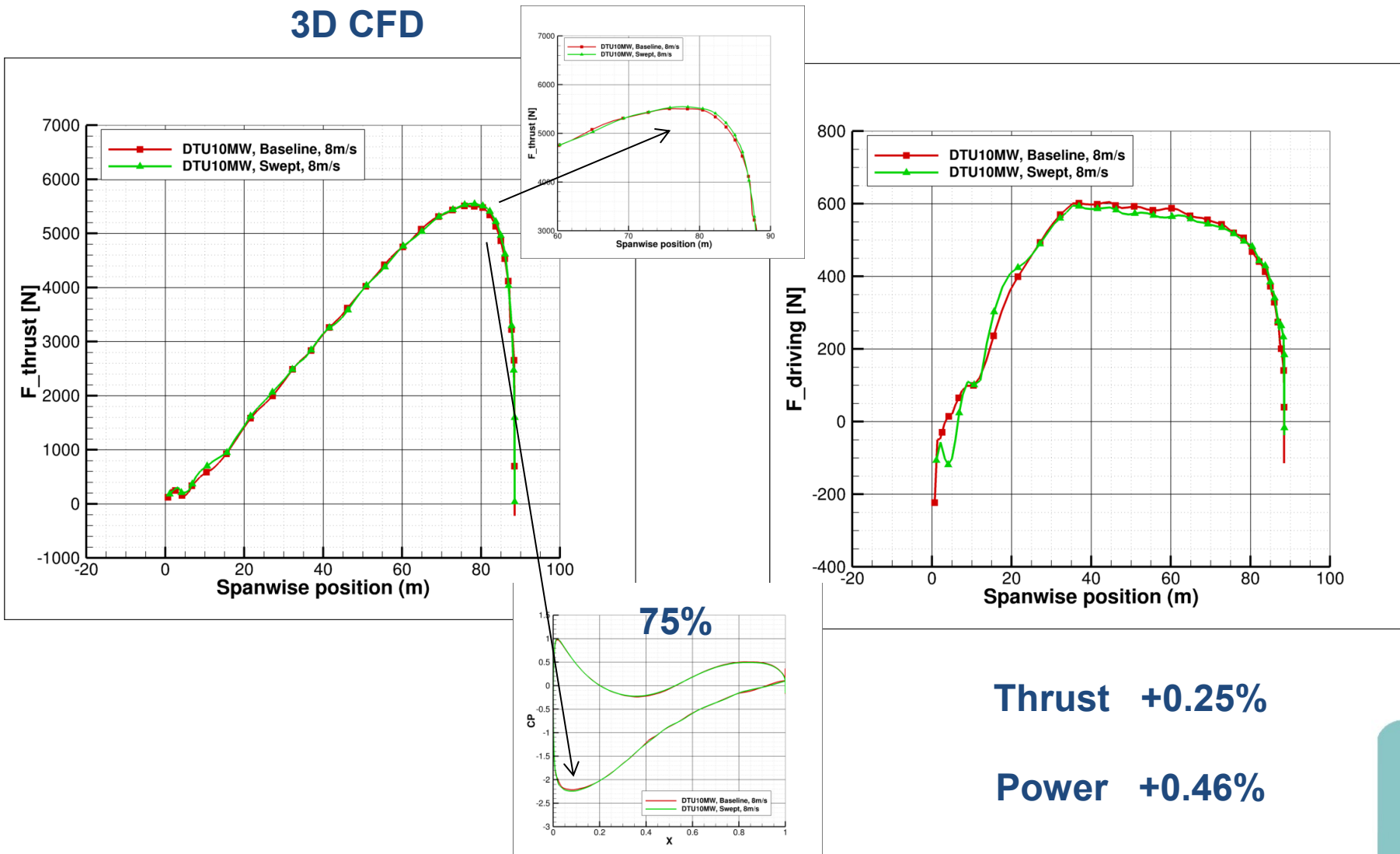


**3D free wake lifting line (wake pattern)**



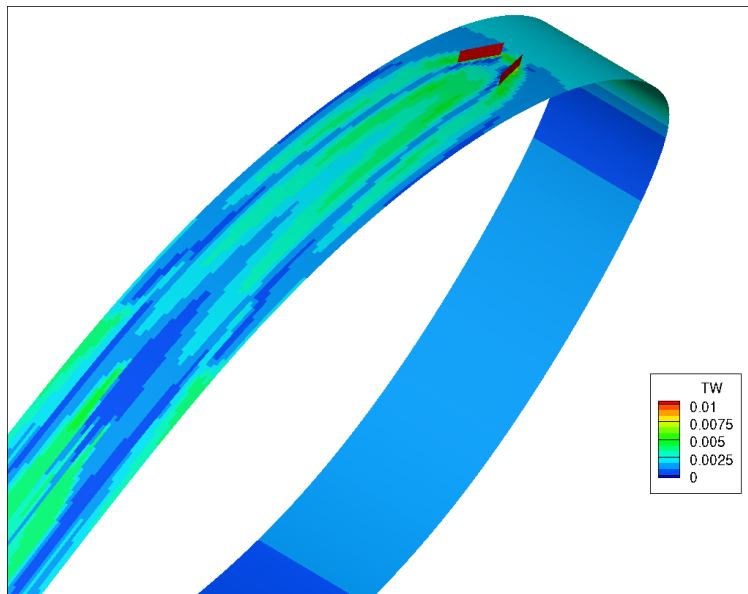
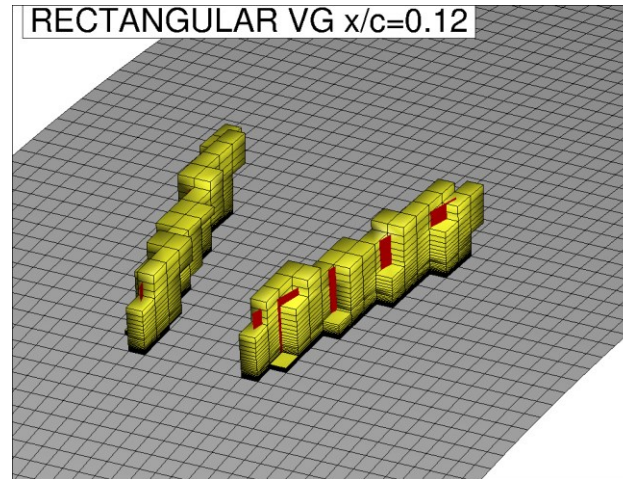
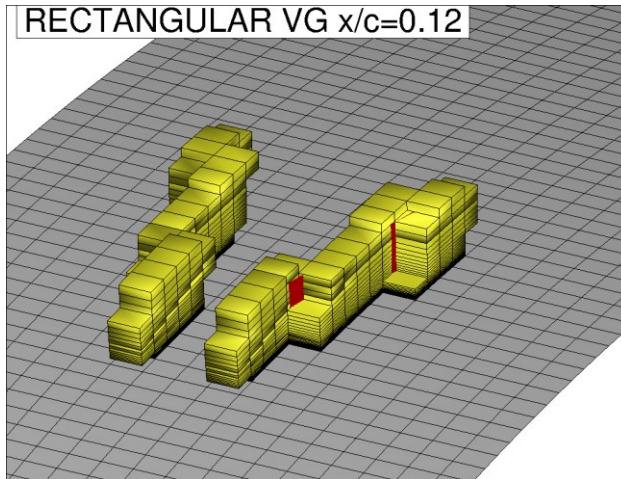
**3D free wake aeroelastic (wake pattern)**

## 3D CFD

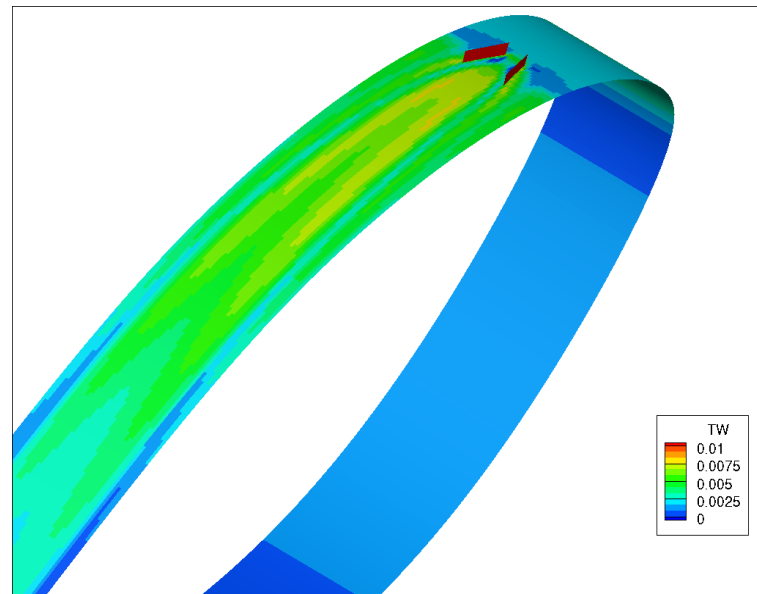


**Thrust +0.25%**

**Power +0.46%**

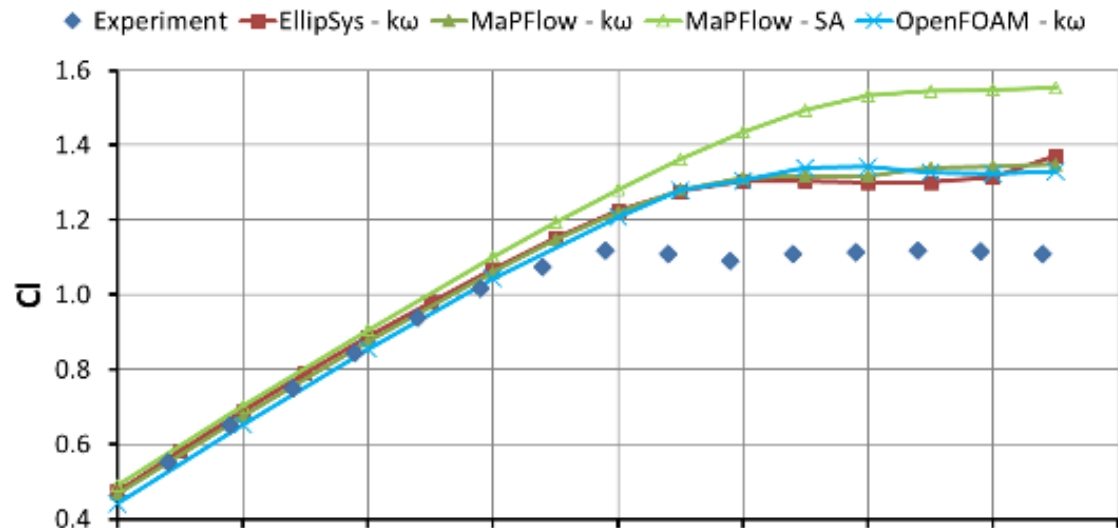


Free transition

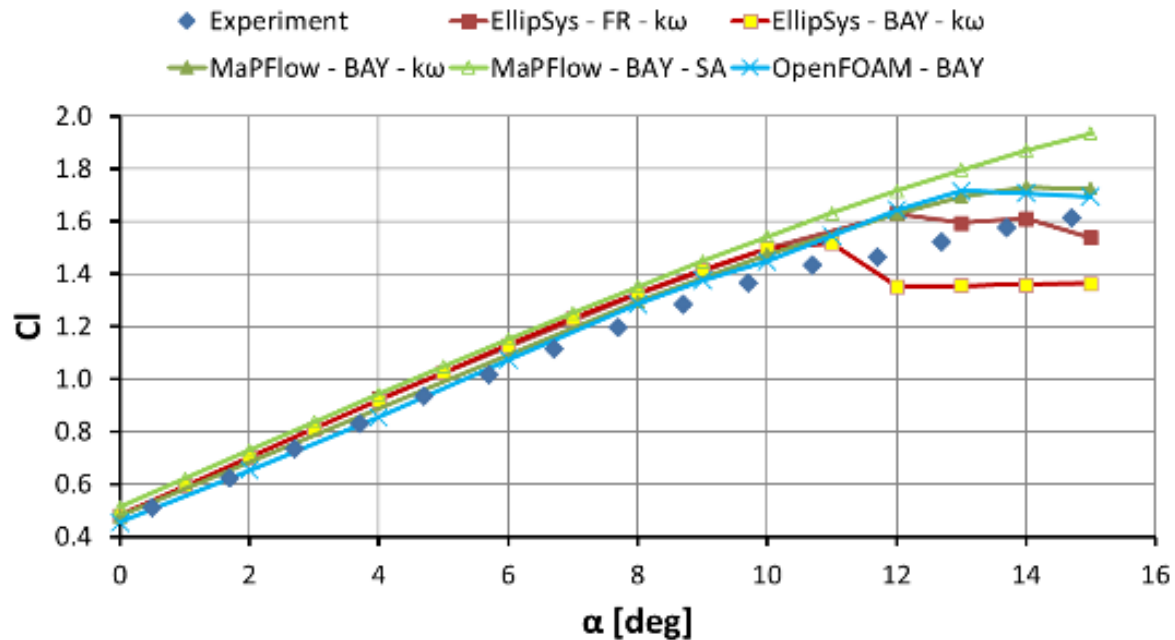


Forced transition

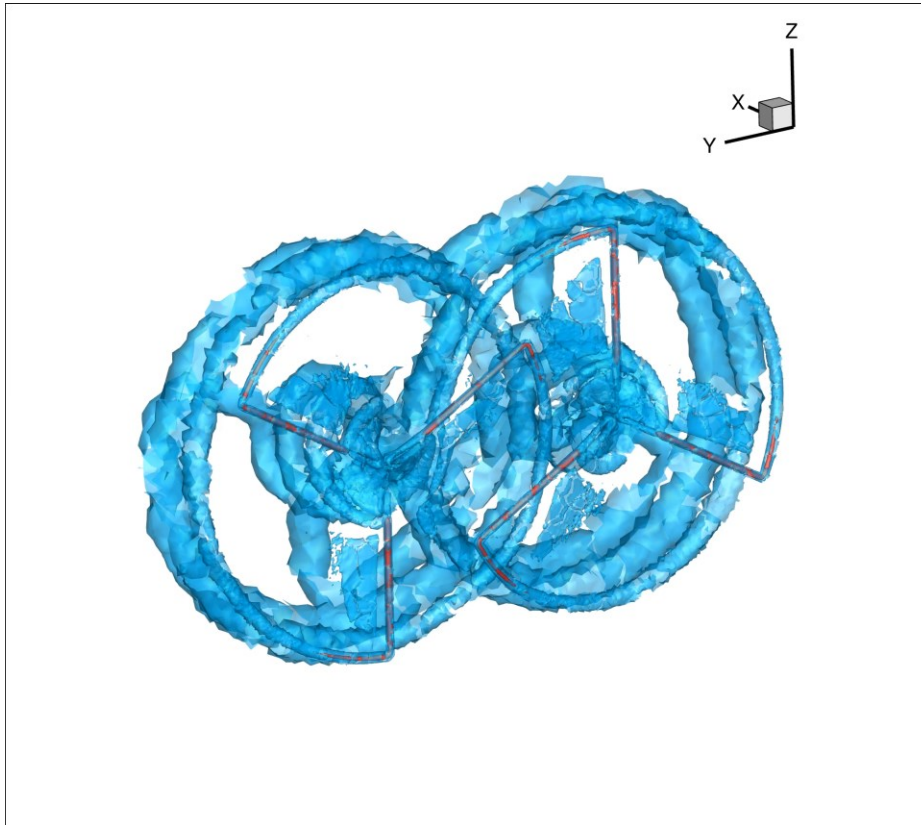
## No Vortex Generators



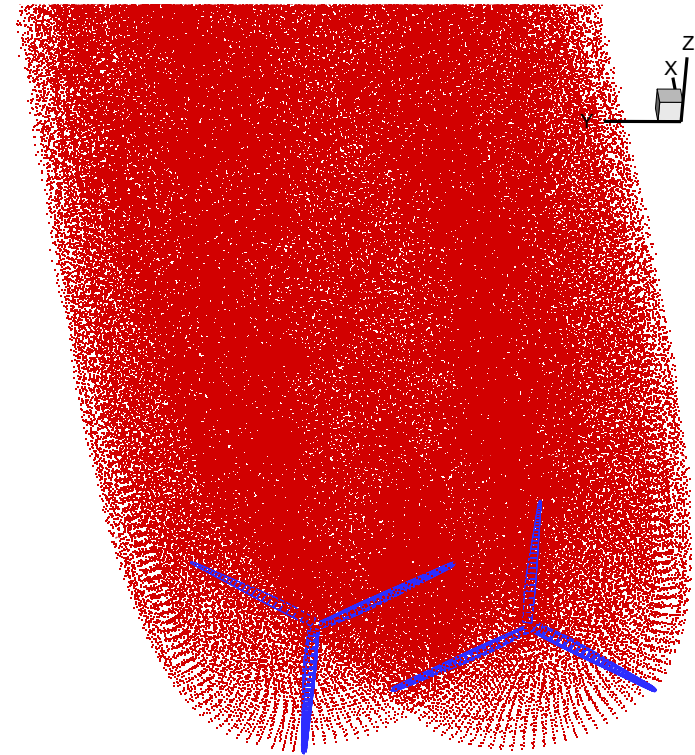
## With Vortex Generators



## Multi-Rotors



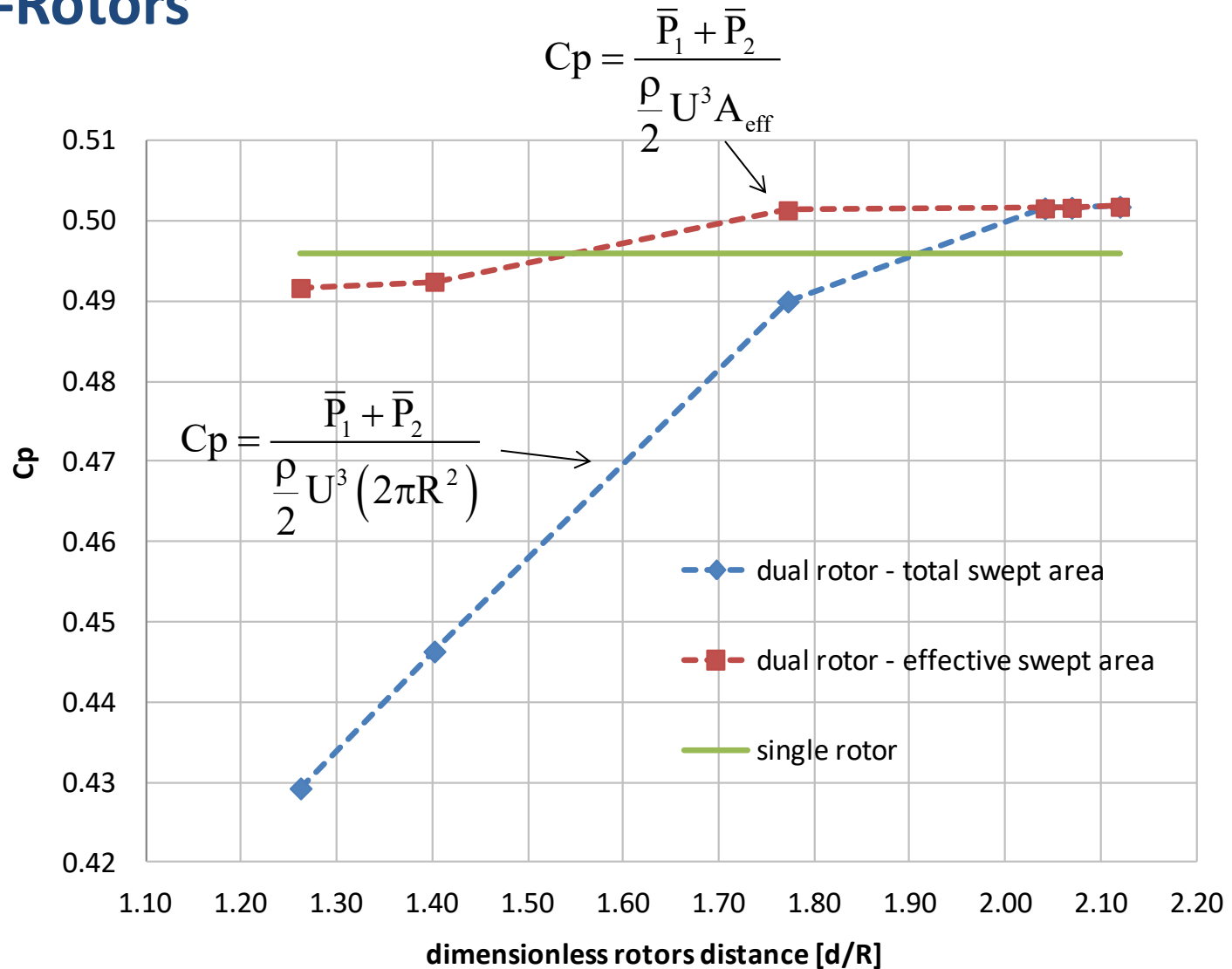
**Actuator Line CFD simulations**



**Vortex Free Wake  
simulation**

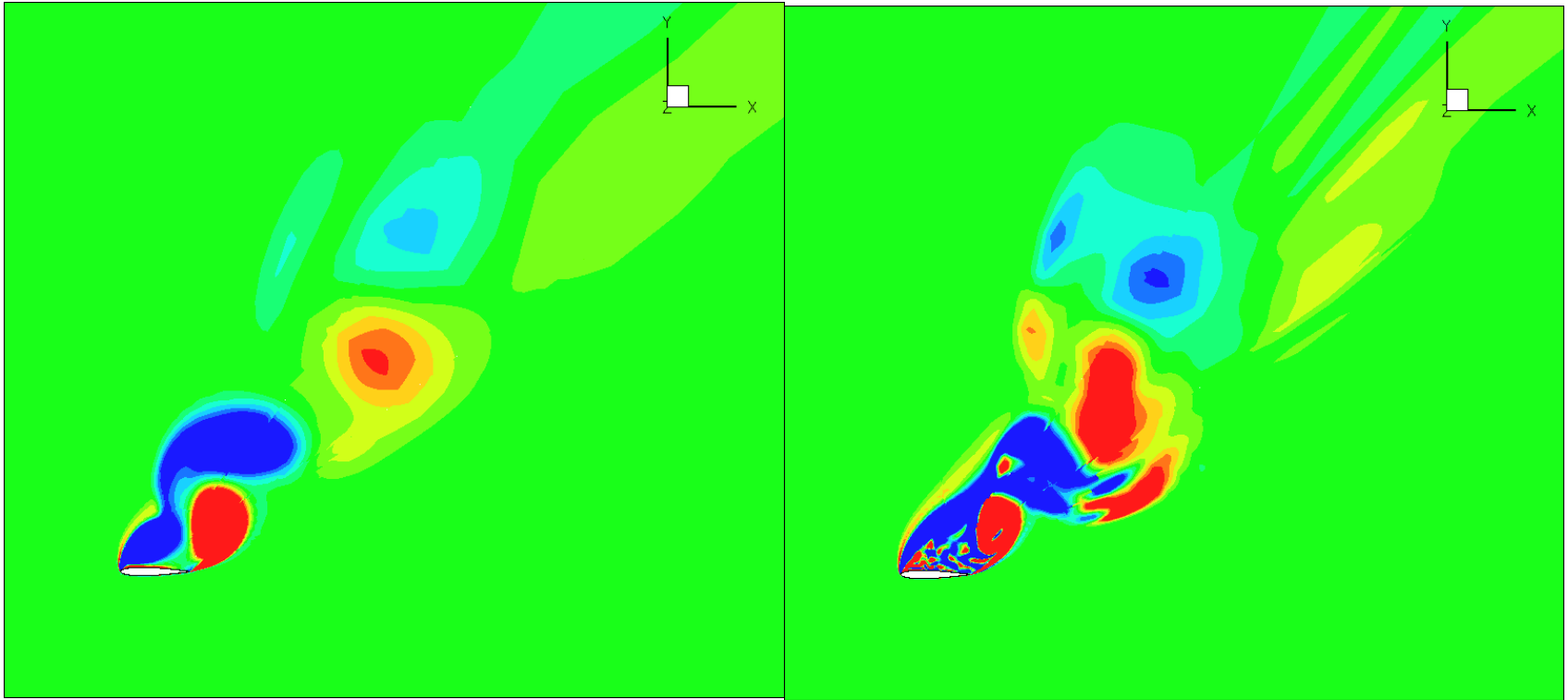


## Multi-Rotors



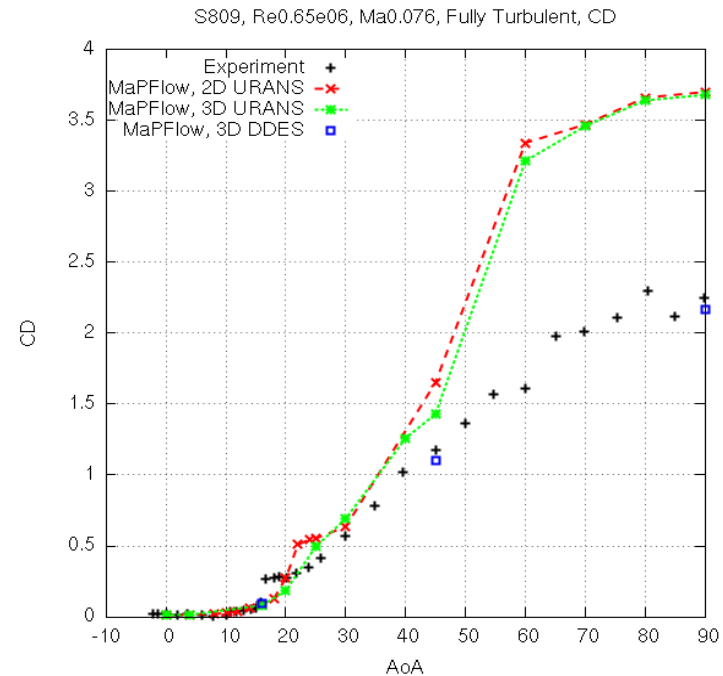
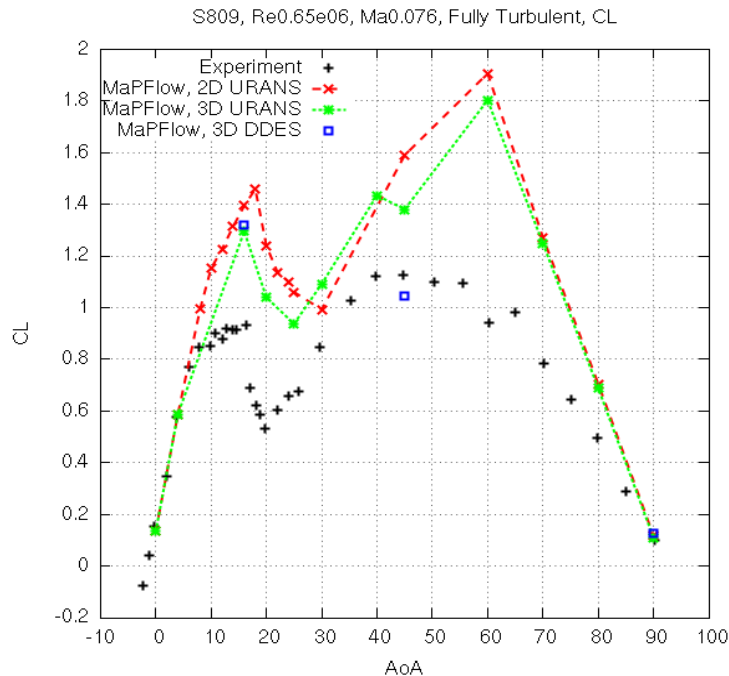
# vortex shedding past an extruded airfoil section

## URANS vs DES



# vortex shedding past an extruded airfoil section

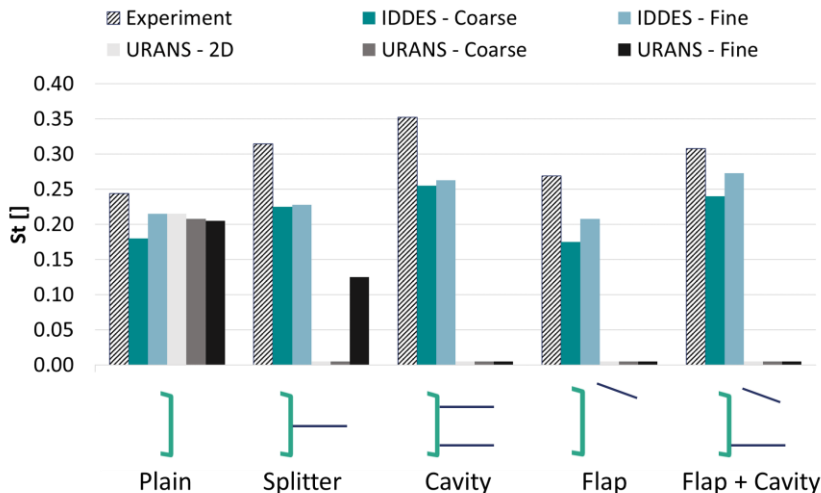
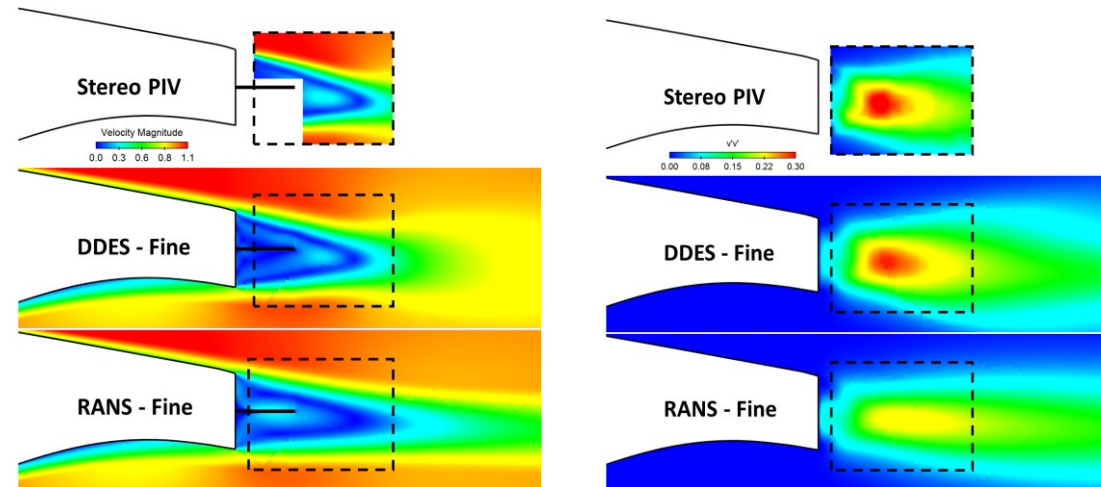
## URANS vs DES



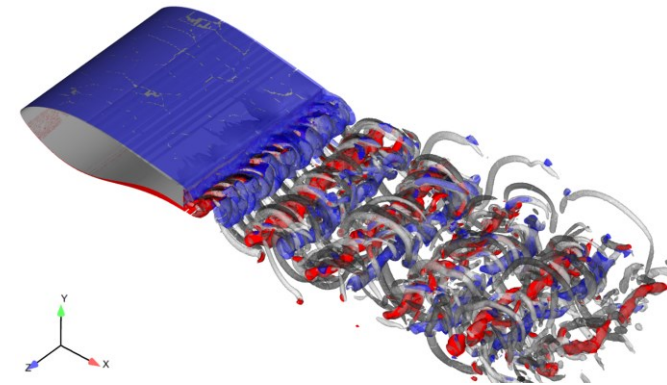
### S809 airfoil

- Reynolds 0.65e06, Mach 0.076
- SA turbulence model, original and DDES versions
- 200x100 2D grid, 200x100x25 3D grid with AR=1

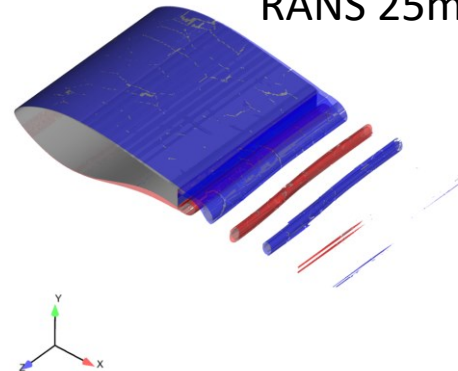
## Modelling of flatback airfoils



IDDES 25m cells



RANS 25m cells

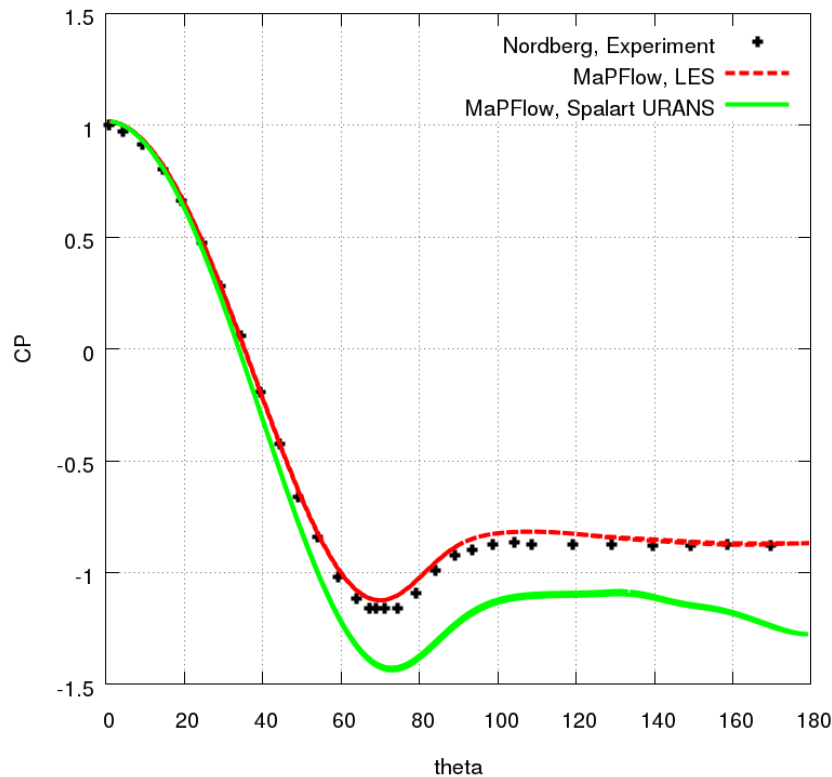


Different Unsteady Characteristics between the IDDES and RANS

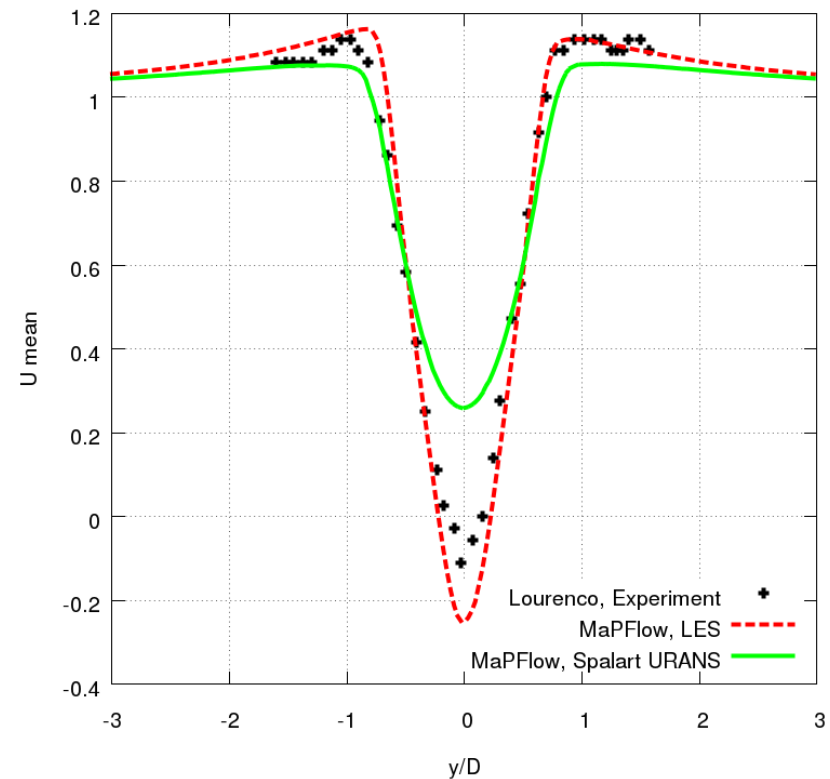
# Cylinder pressure distribution and wake velocities

## LES vs URANS

Cylinder, Reynolds 3900, Mach 0.08, CP

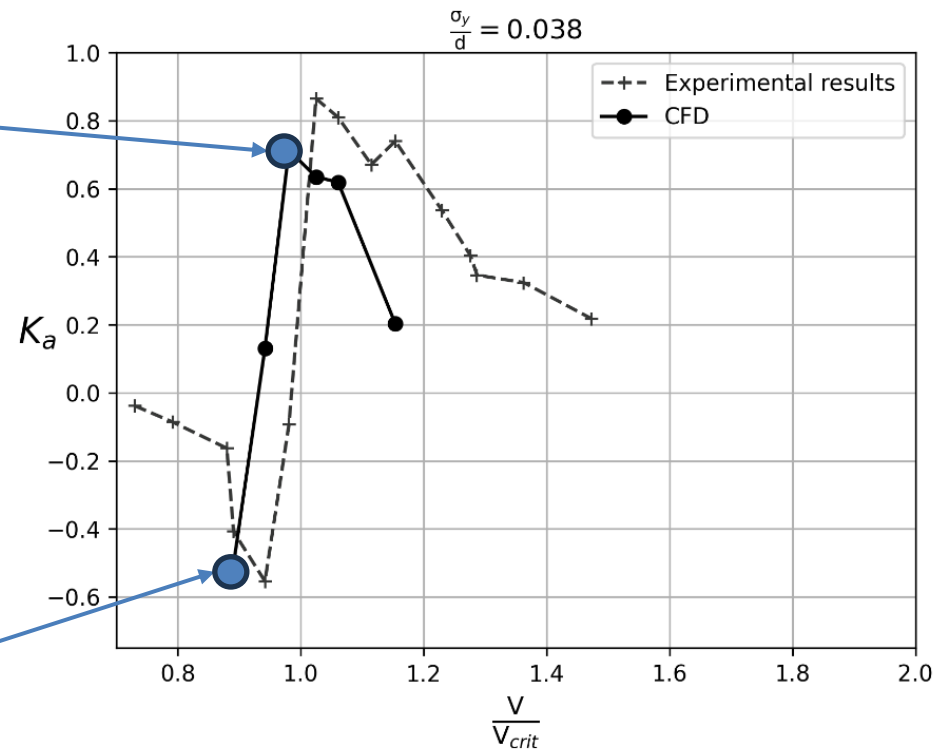
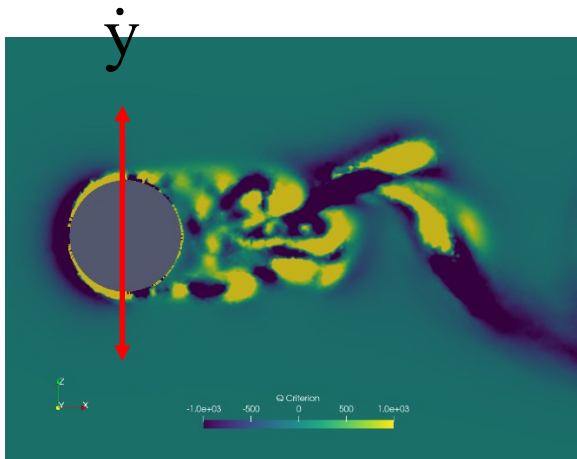
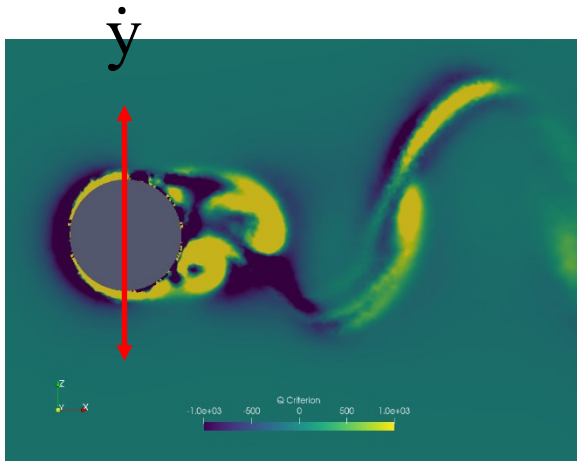


Cylinder, Reynolds 3900, Mach 0.08, U mean at  $x/D=1.54$



Cylinder wake, Re3900

## Prediction of Vortex Induced Vibrations on Wind Turbine Towers

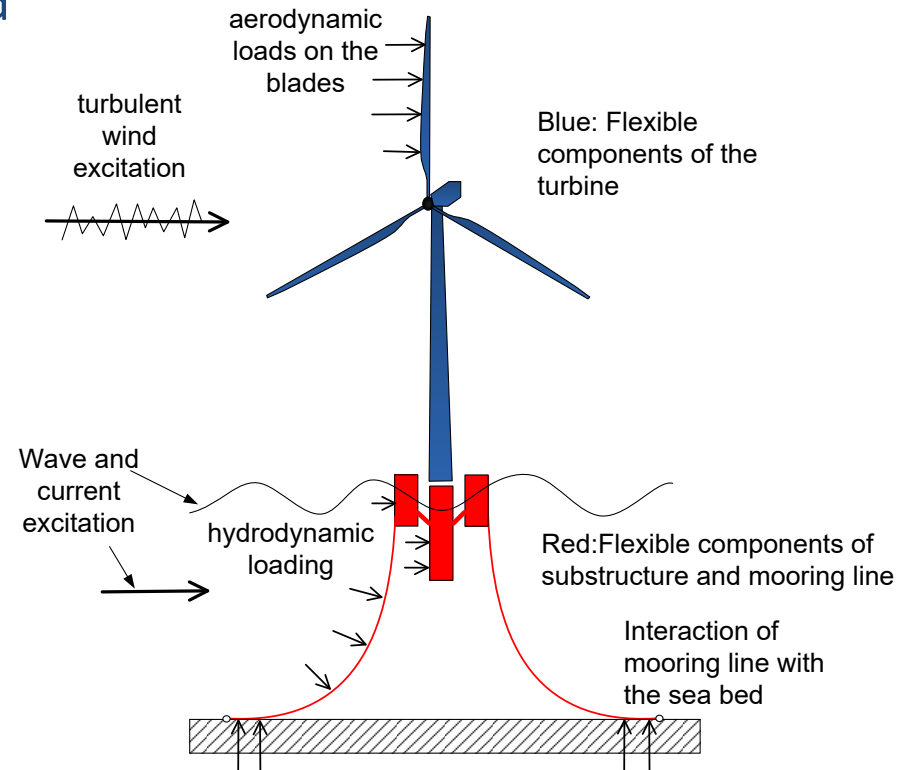


$$P = \int_0^T F_a(t) \dot{y}(t) dt$$

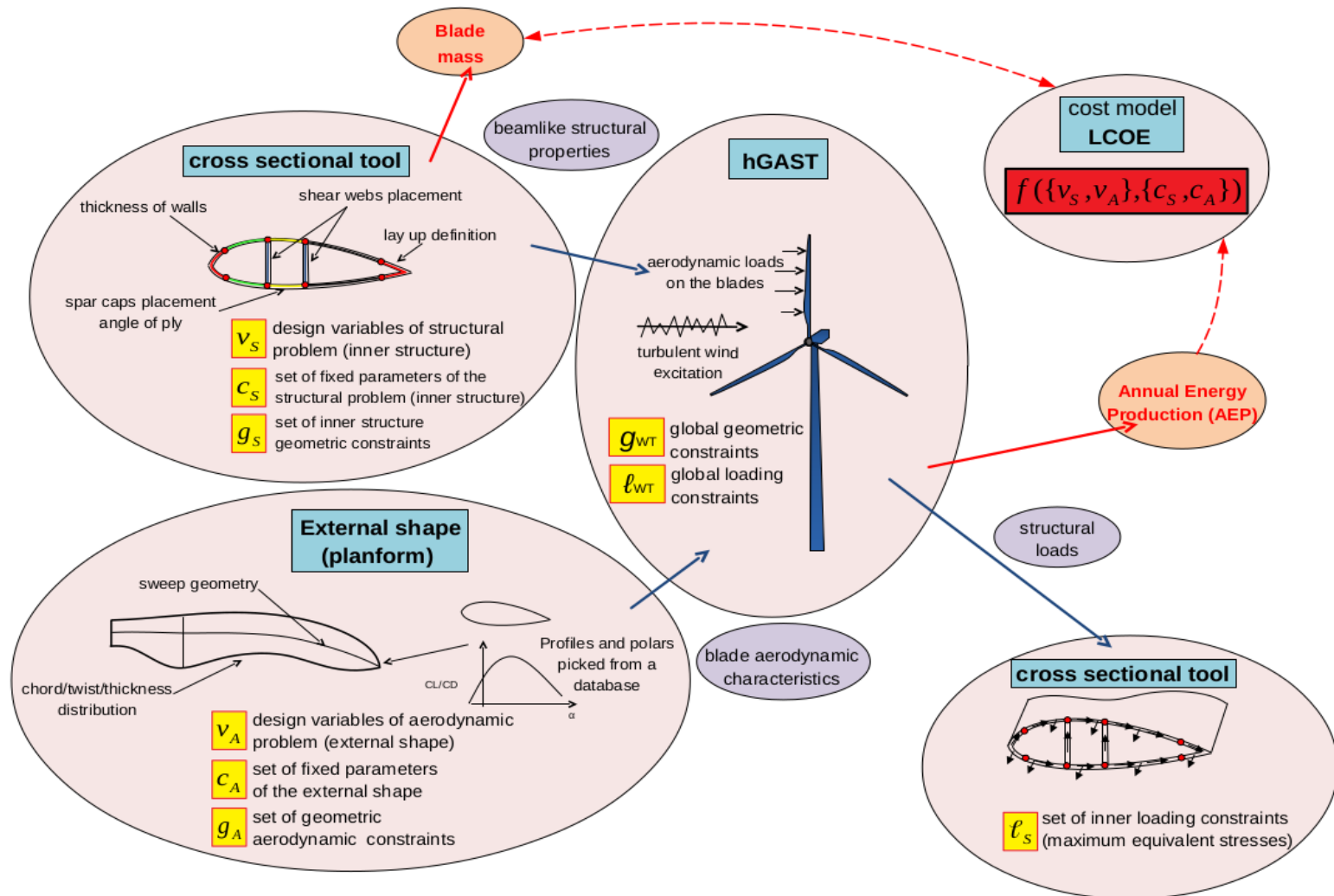


The in-house code **hGAST** is used in the aeroelastic analysis of rotor and combined rotor-pylon problems.

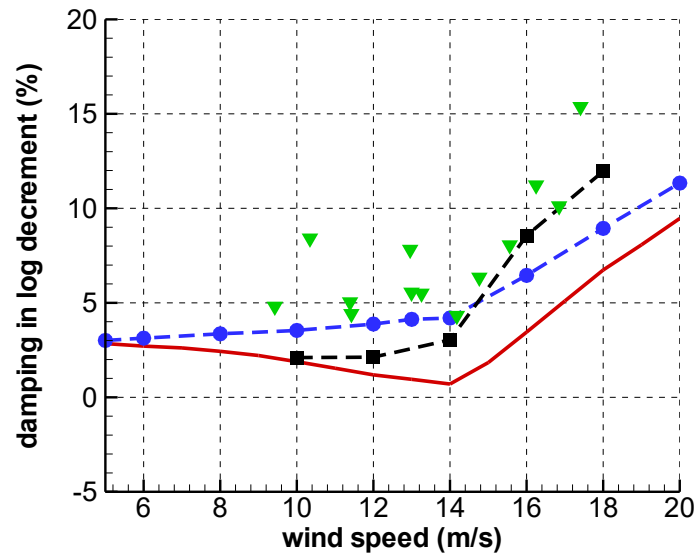
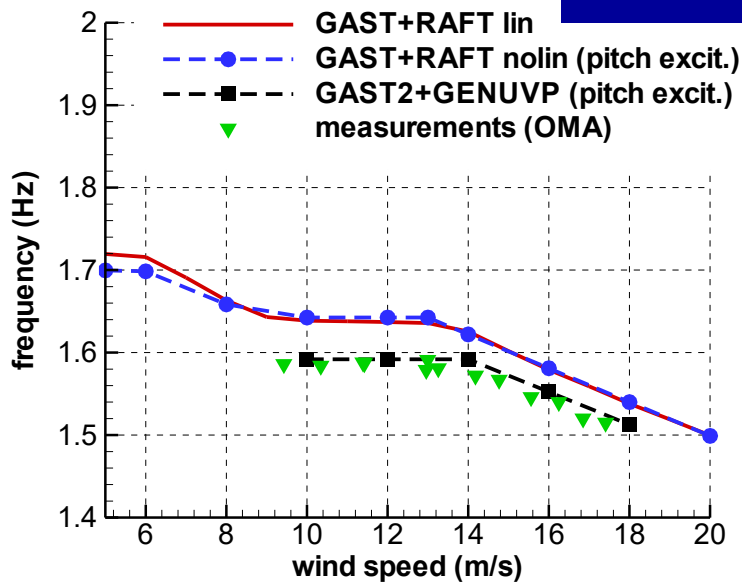
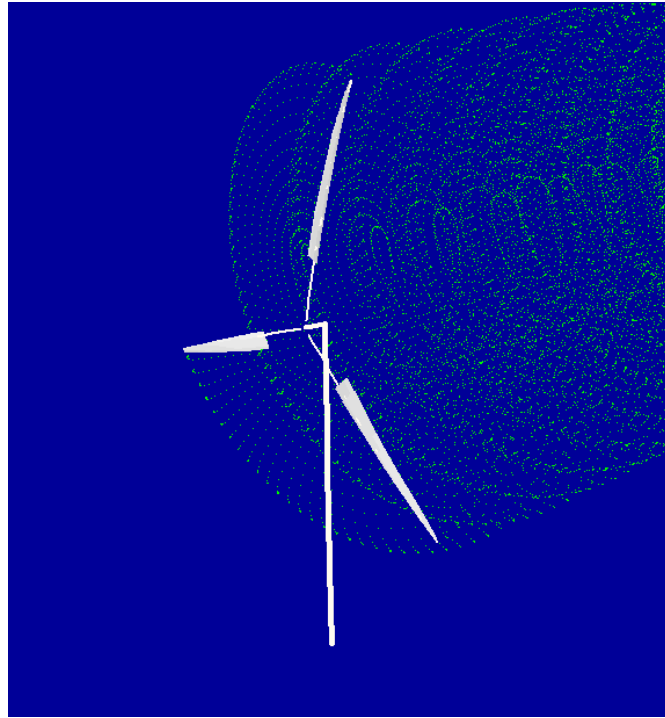
- hydro-servo-aero-elastic tool which besides the structural dynamics is also able to model the controls of the system and the wave loading for offshore (including floating WTs).
- In the modeling of the flexible bodies higher order beam models and multibody dynamics are applied using a FEM discretization.
- It can be used for non linear time domain analysis and linear eigenvalue stability analysis



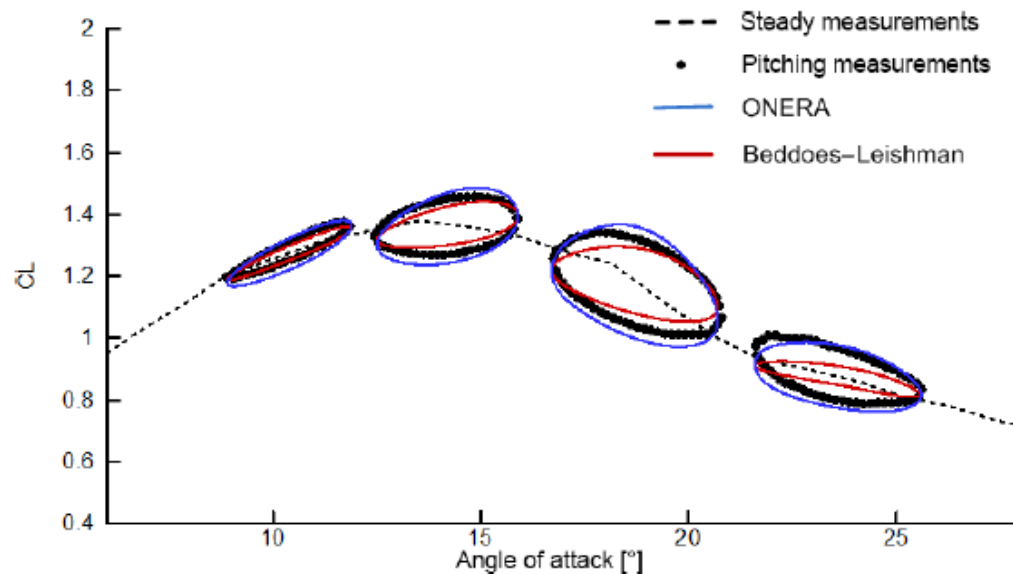
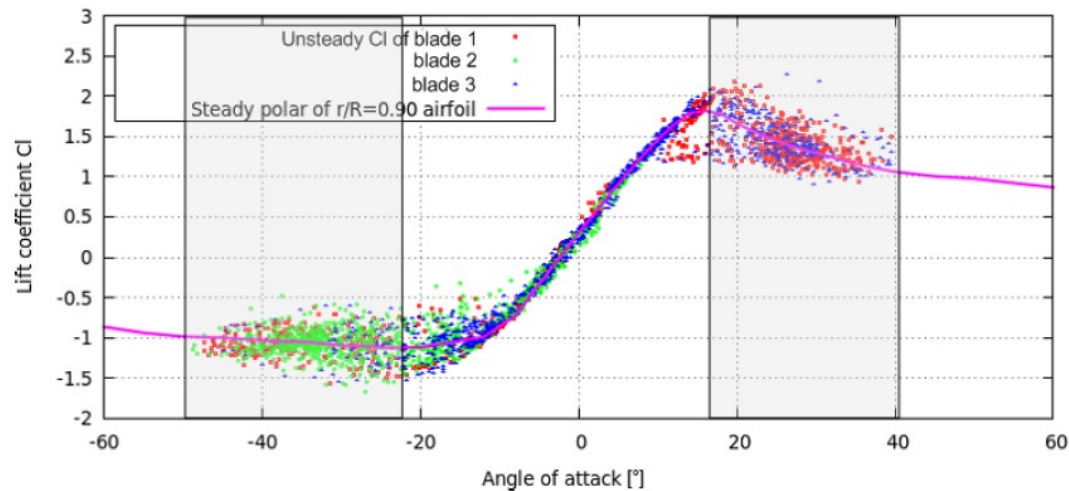
## Multi-Disciplinary-Optimization tools



## Stability Analysis Tools



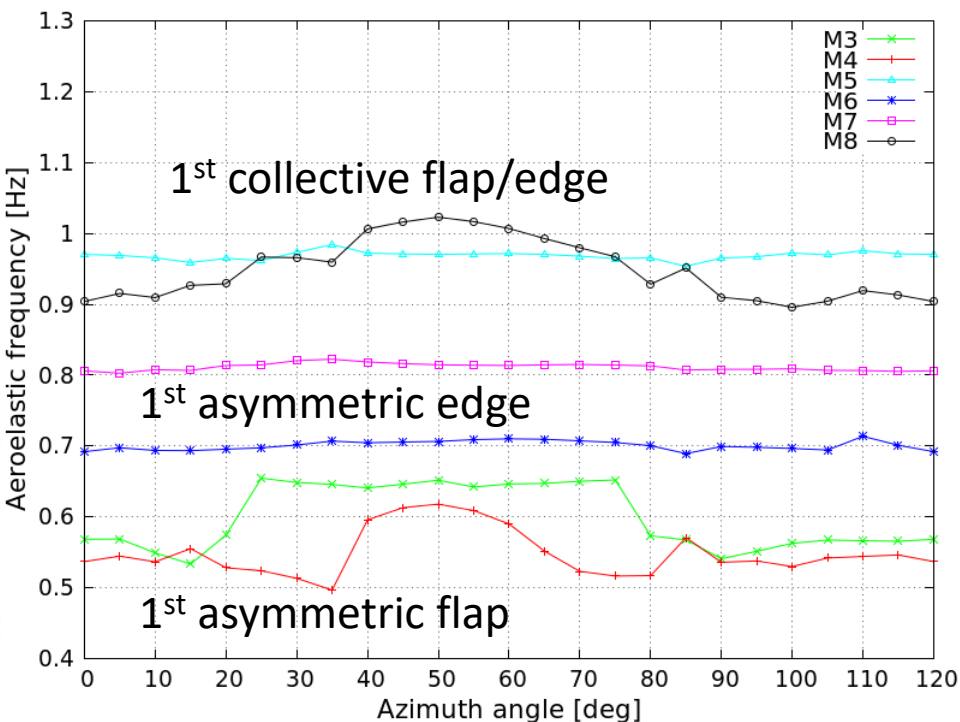
## Stall Induced Vibrations during idling or parked state



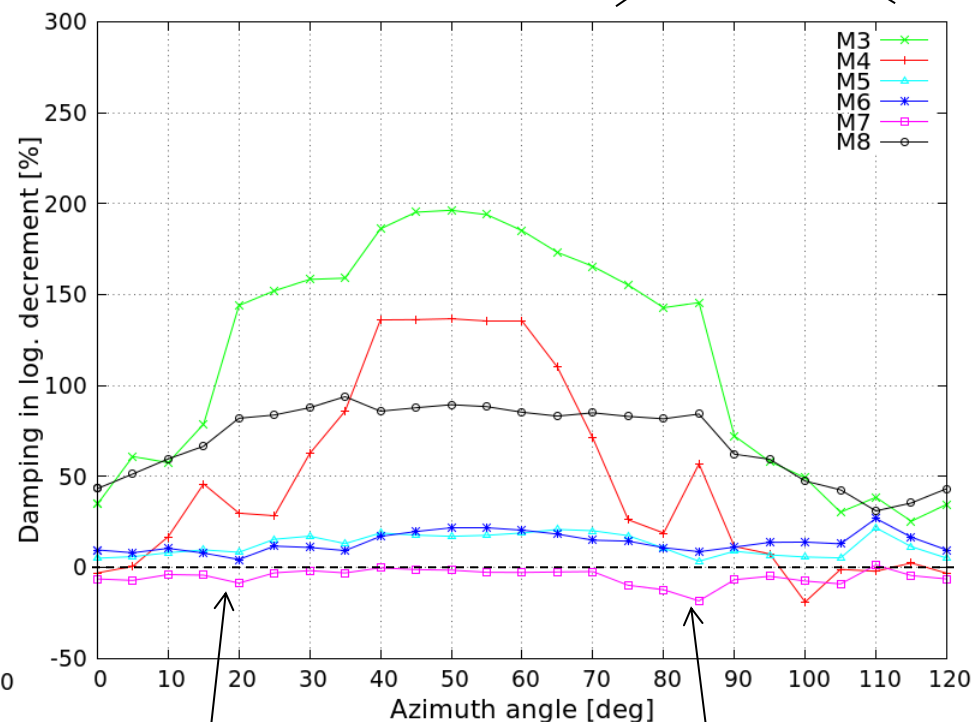
## Stall Induced Vibrations during idling or parked state

Wind speed=42.5 m/s, yaw=30deg

modal frequency

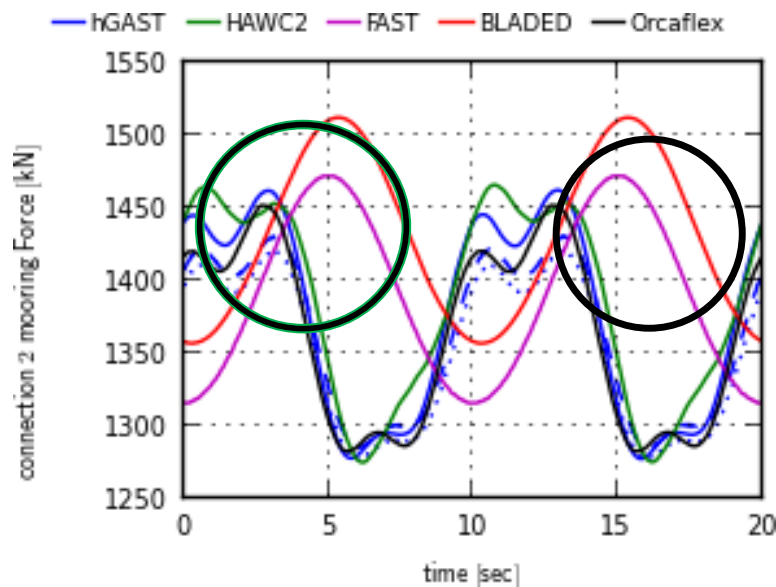
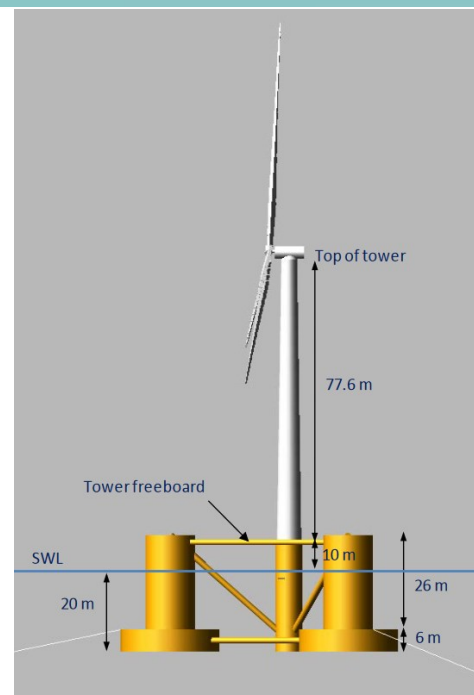
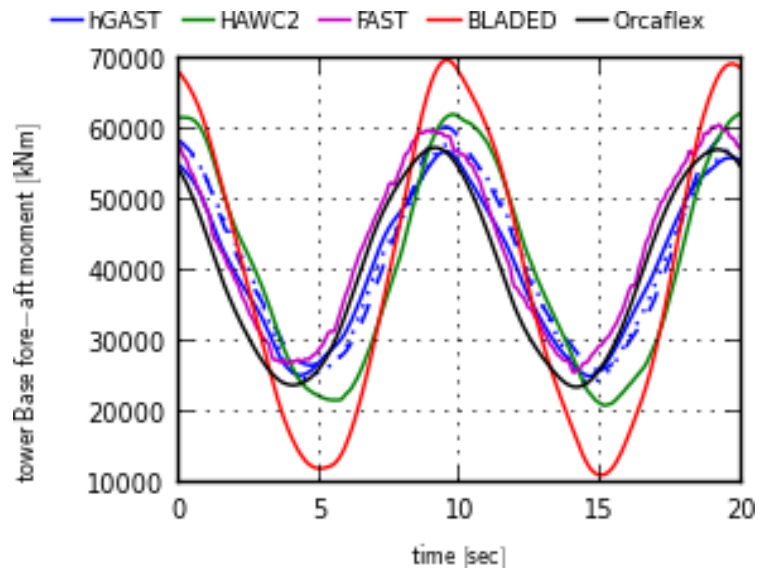
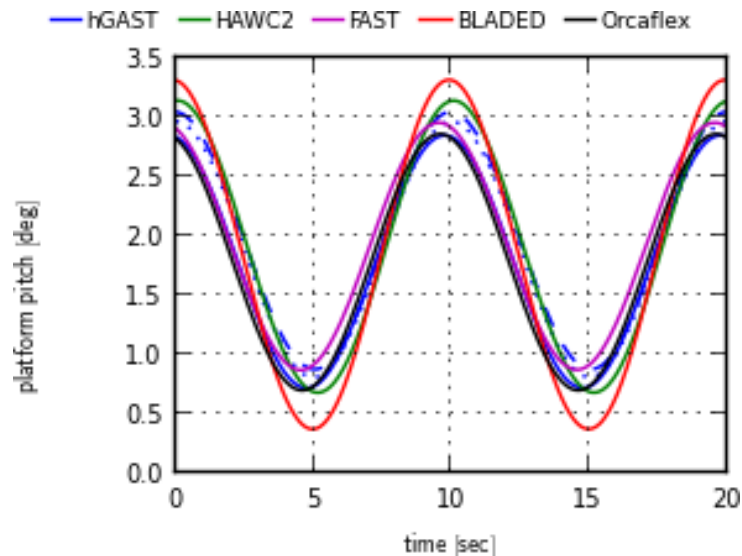


modal damping

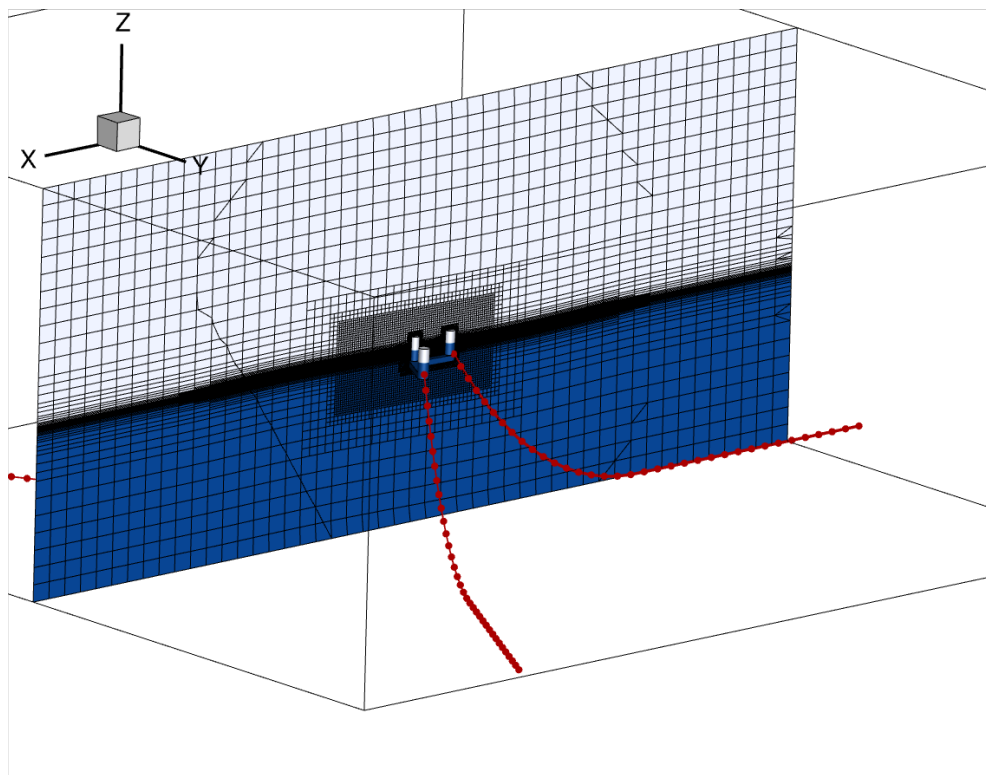


negative damping values of M7 in the vicinity of 20deg and 80deg azimuth angles

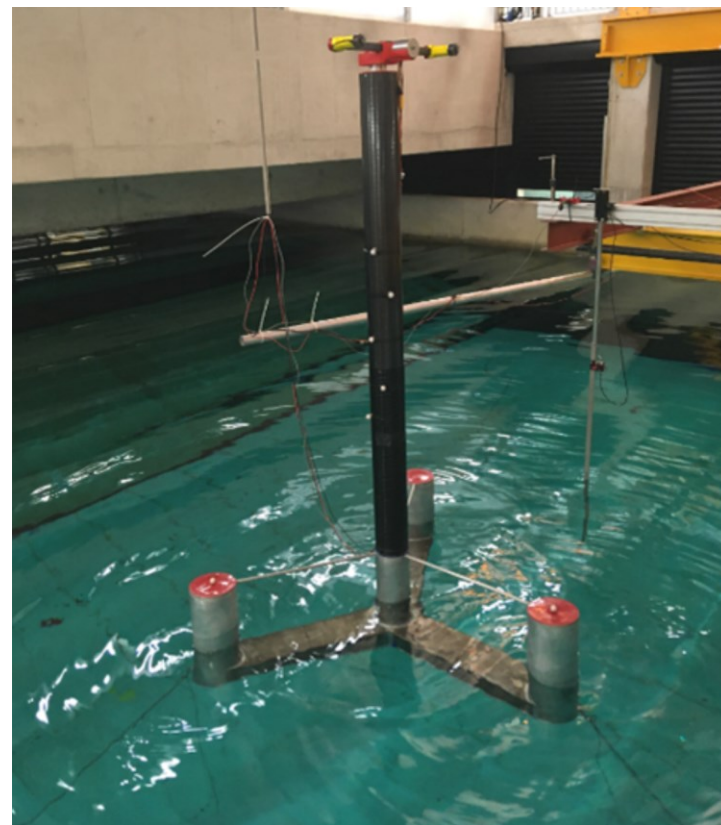
## IEA - OC4





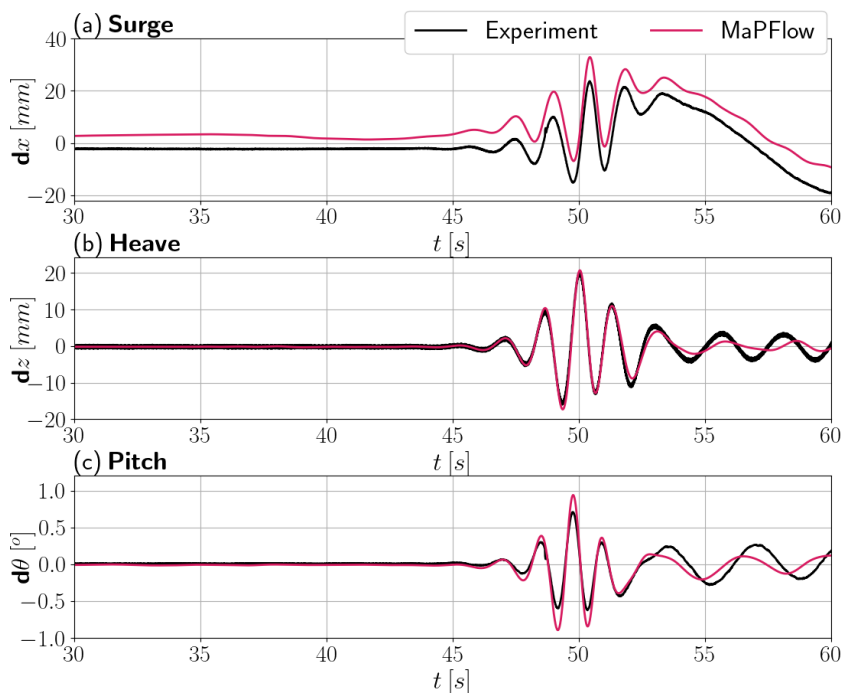


Numerical wave tank

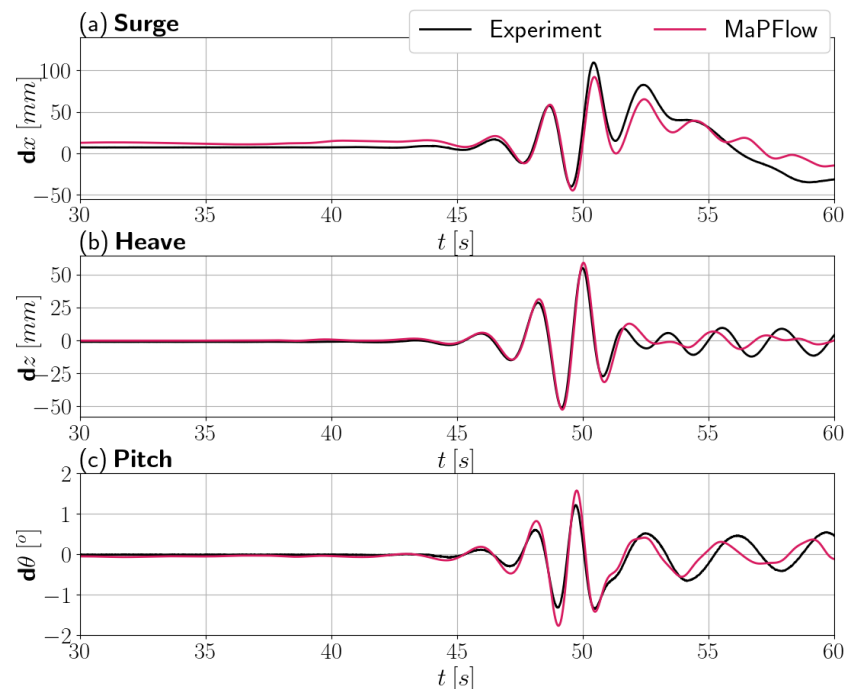


Floater UMaine VoltturnUS-S  
Scale 1:70

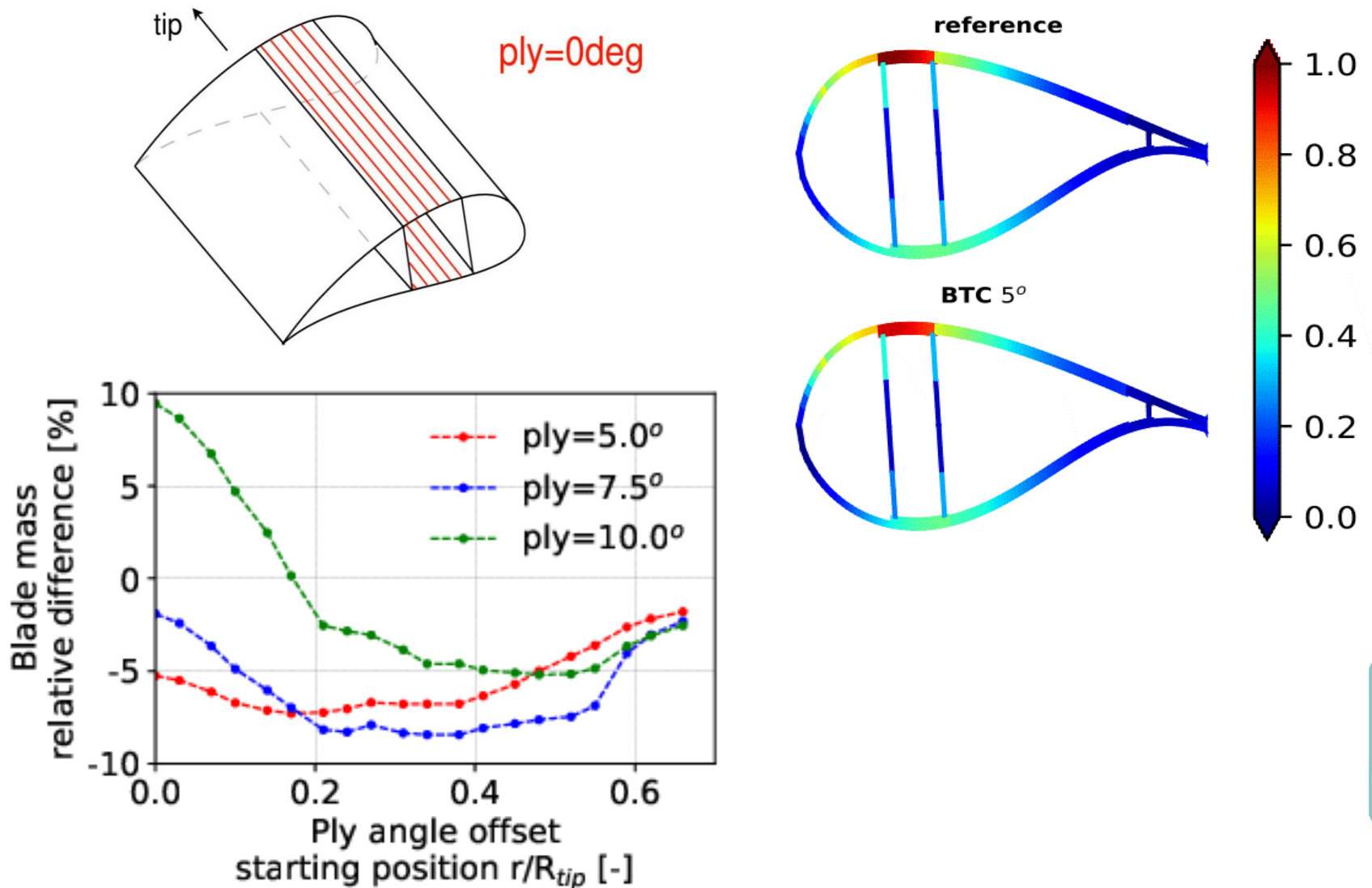
## Benign wavetrain



## Extreme wavetrain

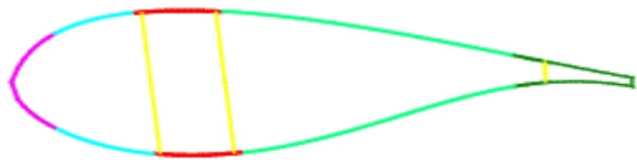


## Optimization of the use of passive control of loads – Bend Twist Coupling

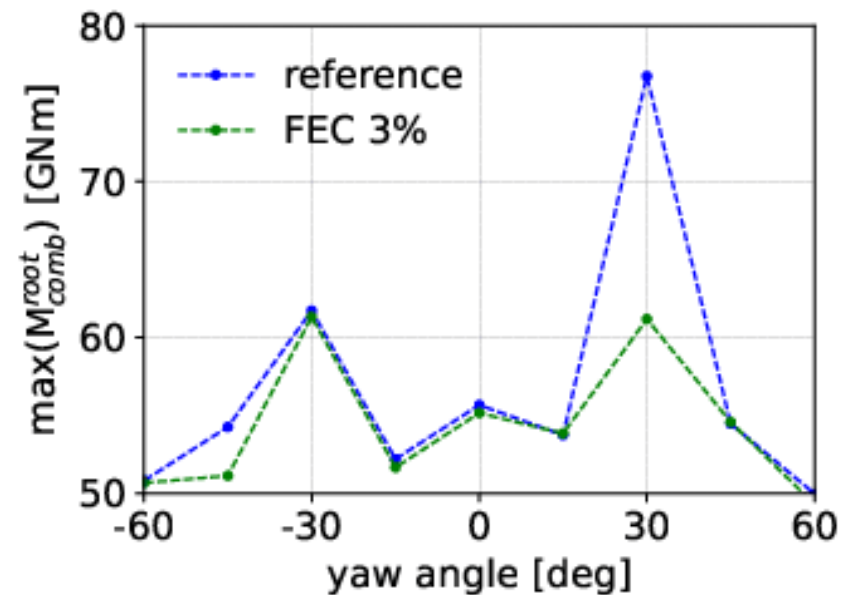
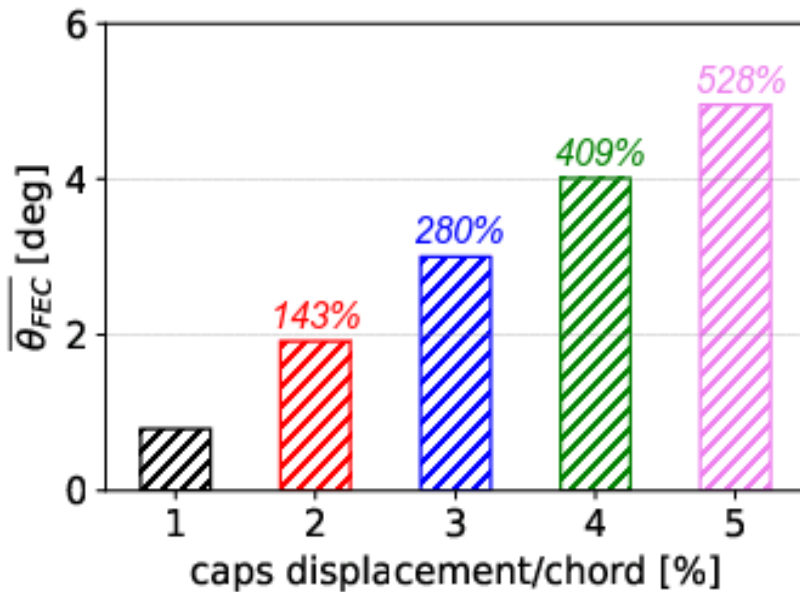


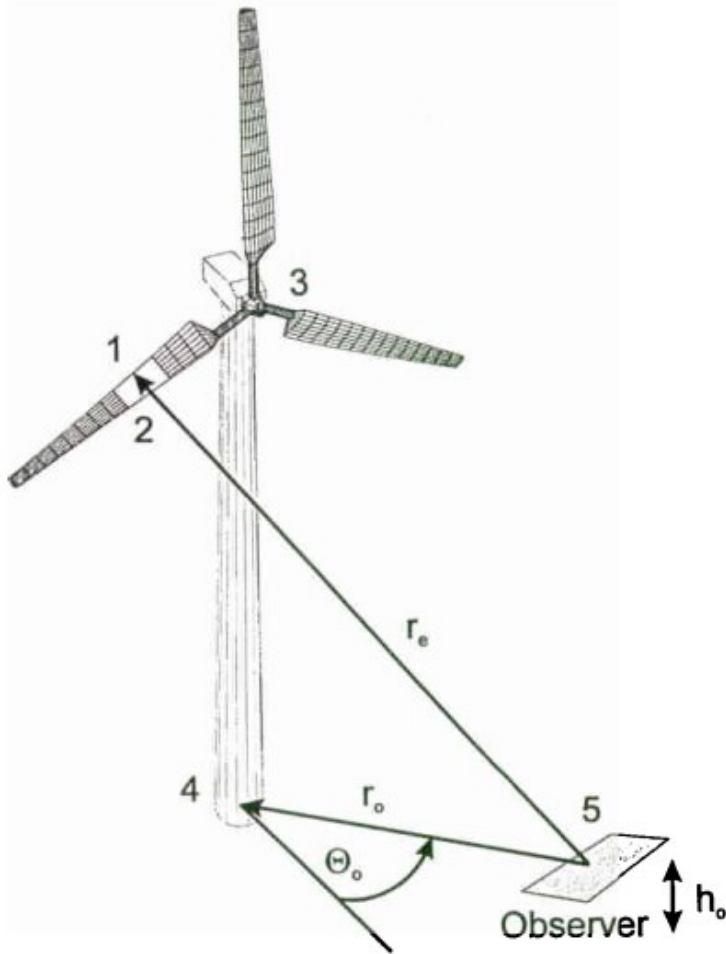
## Optimization of the use of passive control of loads – Flap Edge Coupling

geometry FEC 0%



trailing    nose    caps  
leading    tail    webs

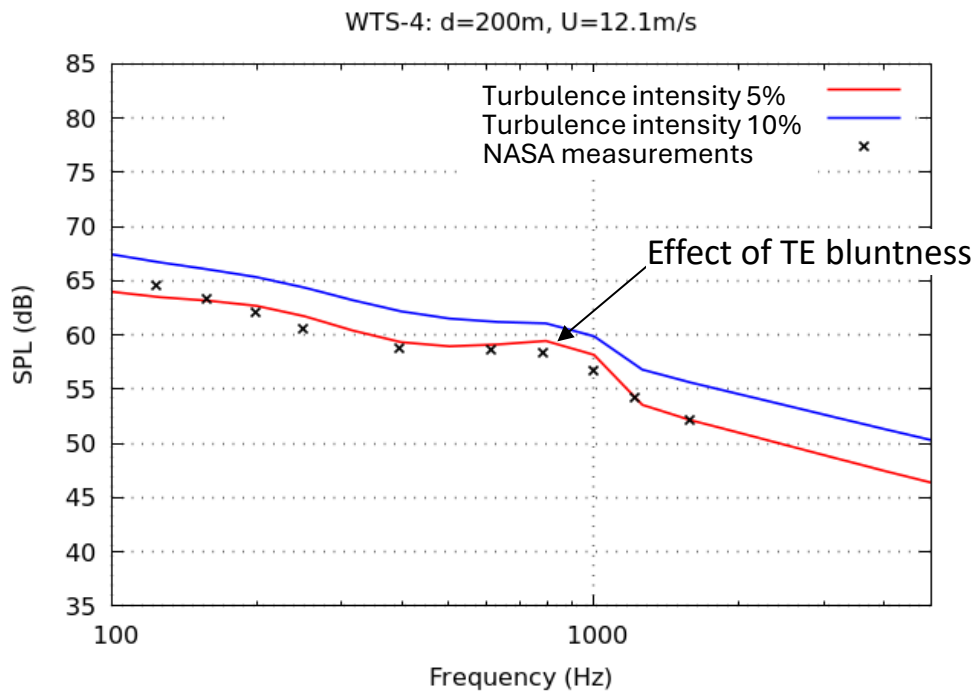




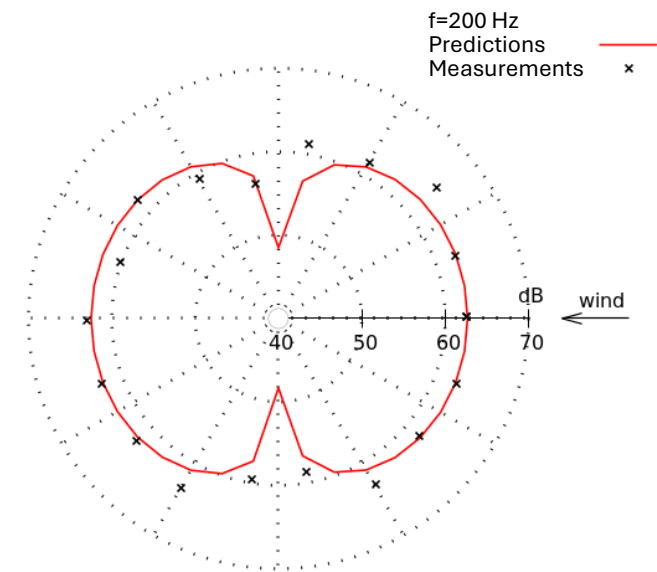
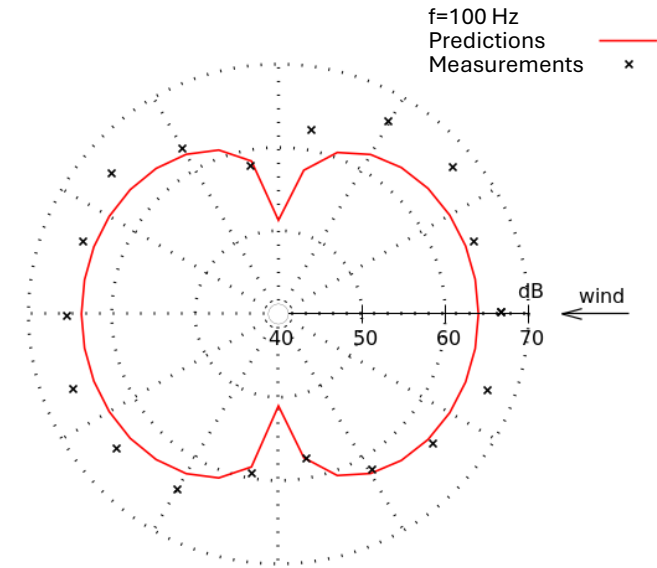
- Quasi 3-D approach
- Blades are discretized into sections
- Section by section 2D aerodynamic simulations using viscous-inviscid interaction code Foil2w →  $\delta$ ,  $\delta^*$  at trailing edge
- Each blade section is treated as an acoustic source
- Aeroacoustic calculations using semi-empirical relationships (Brooks, Pope, Marcolini)
  - ✓ Turbulent boundary layer noise
  - ✓ Turbulent inflow noise
  - ✓ Trailing edge bluntness noise
  - ✓ Laminar boundary layer noise
  - ✓ Tip noise
  - ✓ Stall and separation noise
- Total SPL is obtained by summing up all contributions:

WTS-4 turbine, 4.2MW,  
Diameter, hub height = 80m  
Rotational speed = 30rpm

SPL spectrum, d=200m



Directivity polars, d=200m





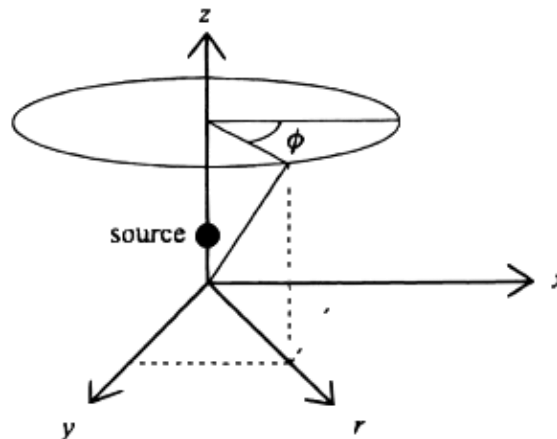
## Axisymmetric approximations in frequency domain

### Ray theory

- Trace eigenrays
- Calculate attenuation loss along the eigenrays through simulation of physical attenuation mechanisms
- Synthesize sound pressure level at the receiver for one or more frequencies

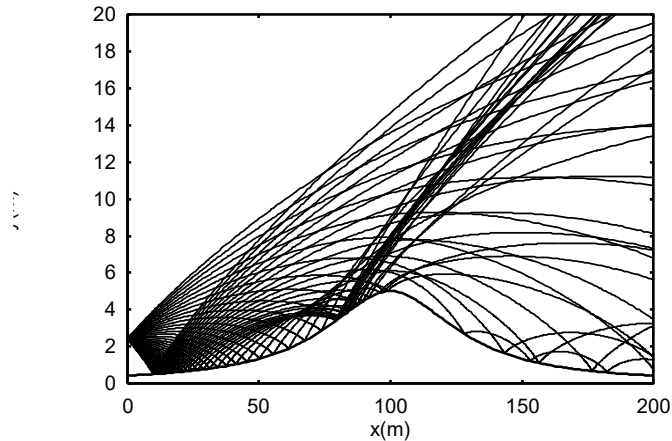
### Parabolic equation

- Simplified form of the Helmholtz equation for small propagation angles
- Back scattering is ignored



## Axisymmetric approximations in frequency domain

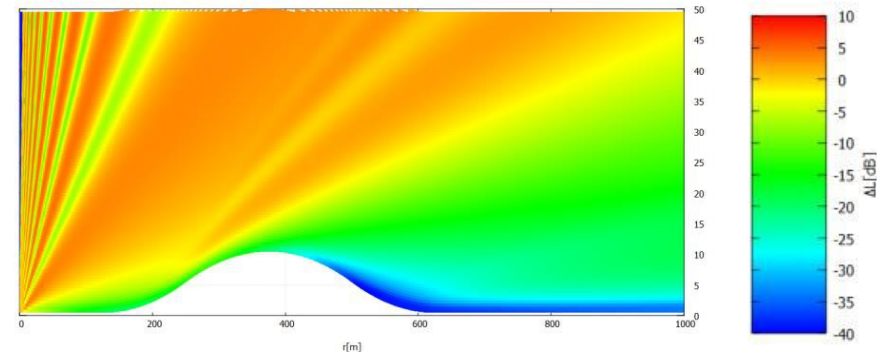
### Ray theory



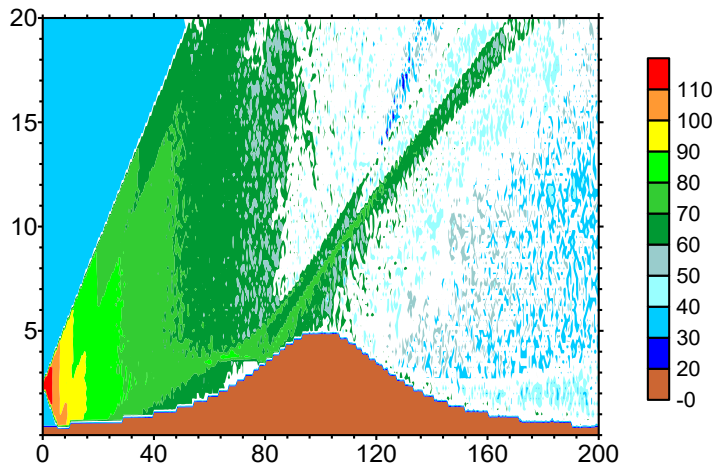
### Parabolic equation

Contours of relative SPL (wrt free spherical propagation)

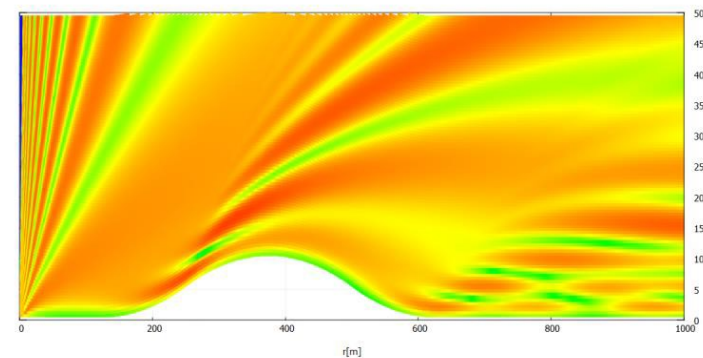
### No wind



### SPL contours

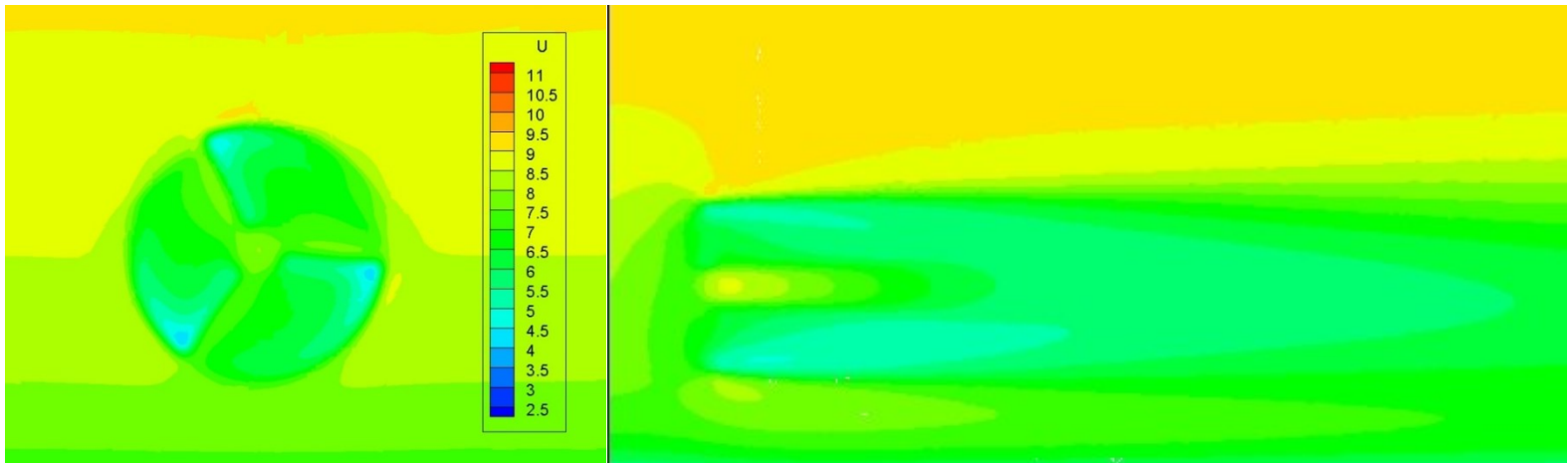


### Downwind conditions

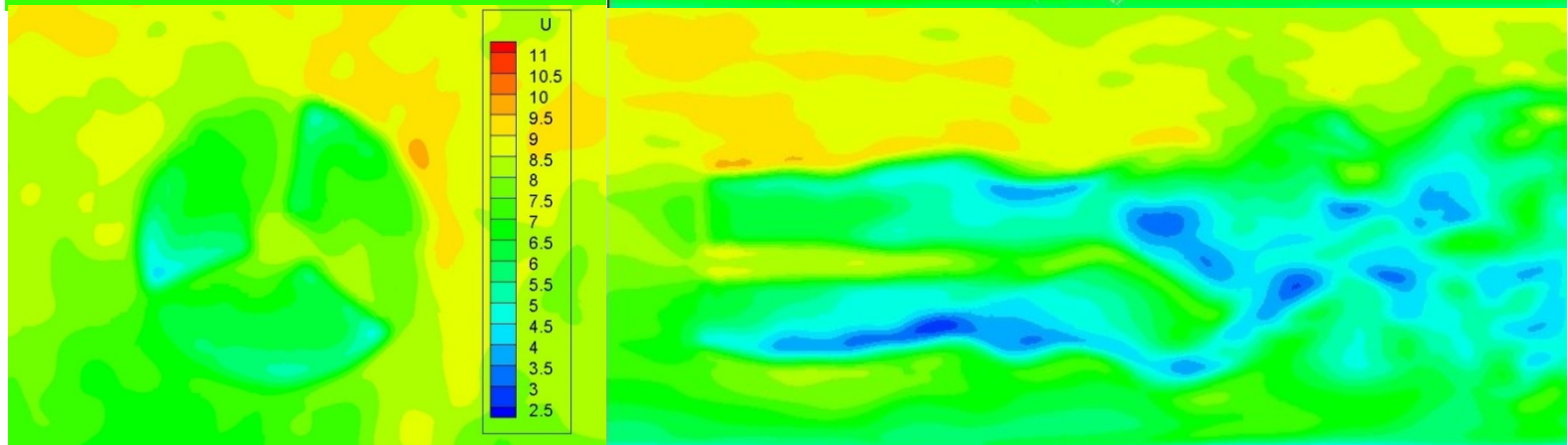


Actuator line - Velocity contours

RANS

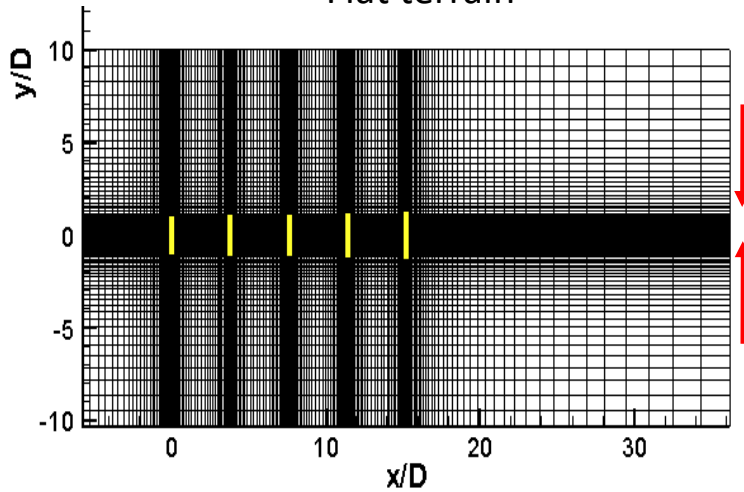


LES

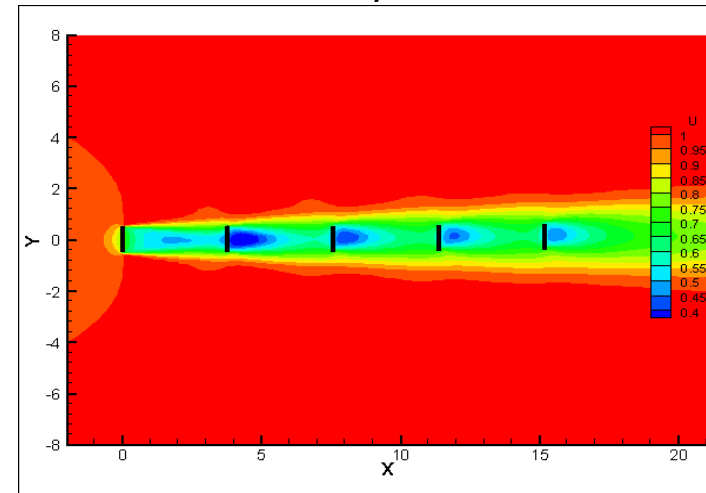


## Actuator disk – Wind farms

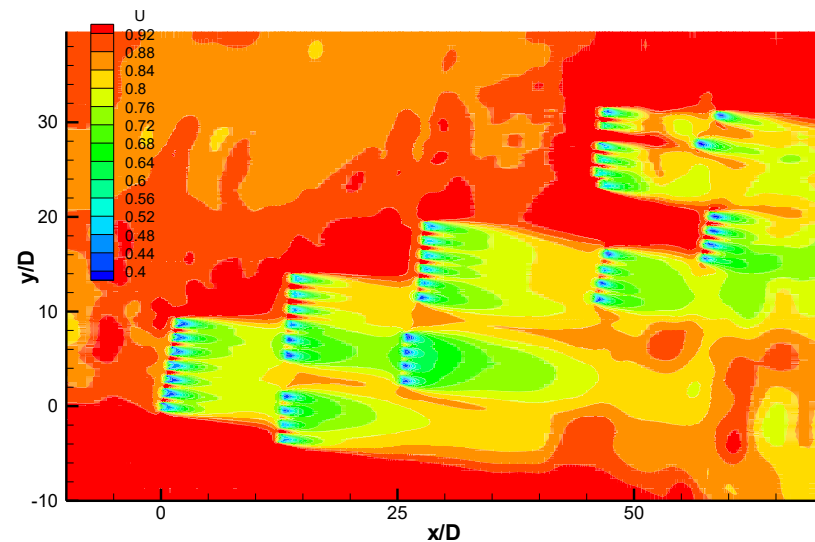
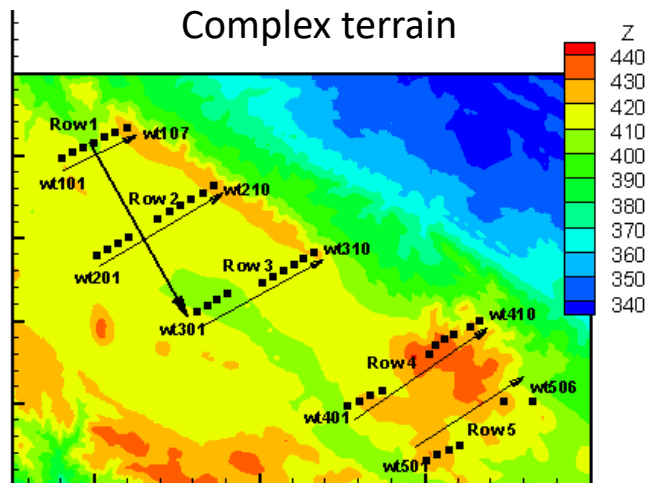
Flat terrain

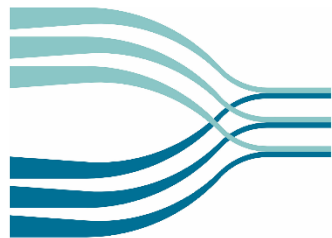


Velocity contours



Complex terrain





# WIND TUNNEL

NATIONAL TECHNICAL UNIVERSITY OF ATHENS

[www.wt.fluid.mech.ntua.gr](http://www.wt.fluid.mech.ntua.gr)



windtunnel.ntua

# Environmental Flows and Buildings

## Wind Tunnel Testing and CFD

Prof. Demetri Bouris

Technical Info Day, NTUA Wind Tunnel Facility – Capacity and Services

June 25<sup>th</sup>, NTUA, Athens

**TWEET-IE / Twin Wind tunnels for Energy and the Environment –  
Innovations and Excellence**

**HORIZON-WIDERA-2021-ACCESS-03-01 / PR# 101079125**





- Wind engineering studies are performed for numerous applications
  - Wind turbines
  - Wind farms
  - Buildings
  - Building complexes
  - Pollutant dispersion etc

- Field measurements are difficult to control
- Computer-aided simulation (CFD) is a more recent and powerful tool in modeling the ABL. It provides valuable insight but many unexplained complexities remain
- Testing in a wind tunnel is repeatable, more accessible and accurate.  
Challenges exist

## Examples :

### Wind Tunnel Testing – Atmospheric Flows

### Computational Fluid Dynamics – Atmospheric Flows

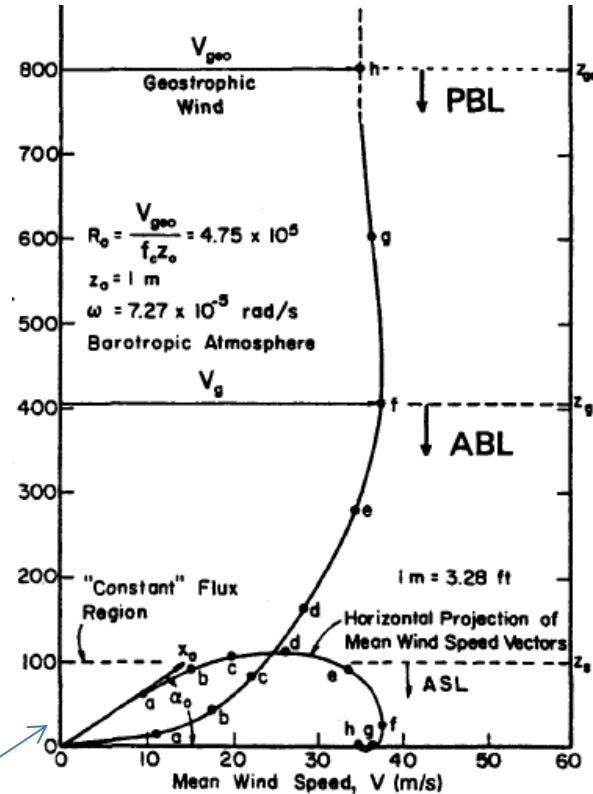
- Modelling the Atmospheric Boundary Layer
- Flows past buildings
- Flows past photovoltaic panel arrays
- Flows past vegetation

## Structure of the ABL

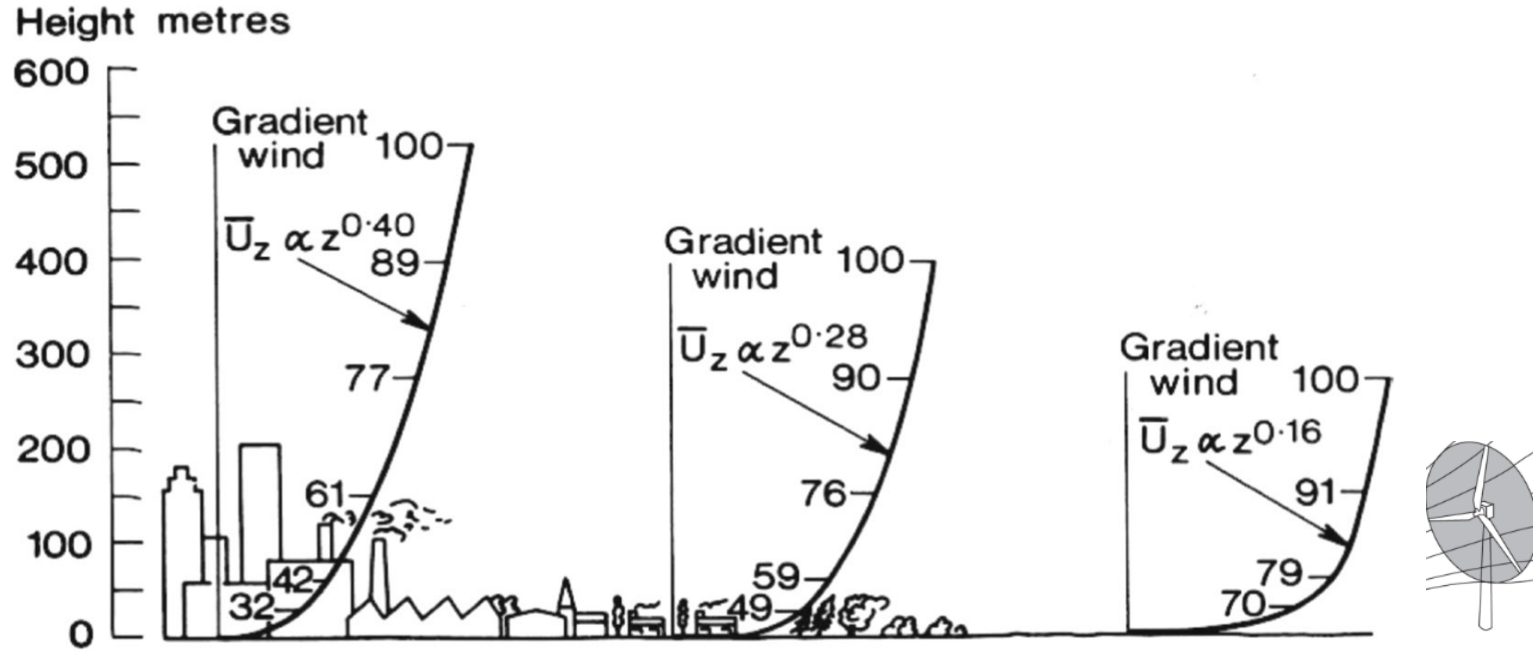
for FUR in neutral atm.  
(Flat Uniformly Rough Terrain)

PBL	Planetary Boundary Layer
ABL	Atmospheric Boundary Layer (surface shear stress negligible)
ASL	Atmospheric Surface Layer (turbulent fluxes vary $\pm 10\%$ )
$z_{geo}$	geostrophic height
$z_g$	gradient height
$z_s$	surface layer height

Eckman  
Layer



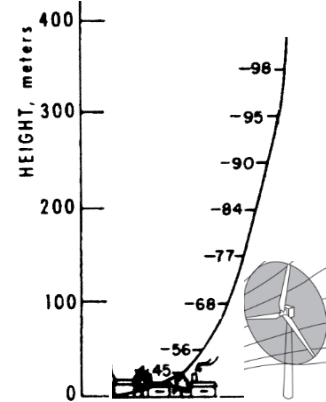
~ logarithmic profile



Effect of surface roughness ( $z_o$ )

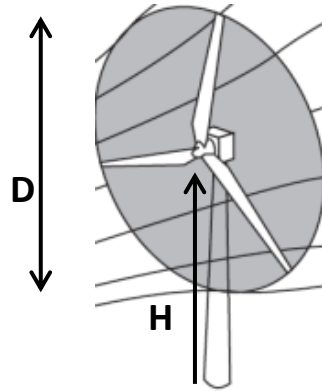
When creating a scale model to study,  
the following criteria must be fulfilled

- **Geometric Similarity:** Similarity of dimensions (length)
- **Kinematic Similarity:** Similarity of time and length scales
- **Dynamic Similarity:** Similarity of forces

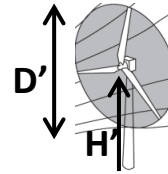




## Geometric Similarity: Similarity of dimensions

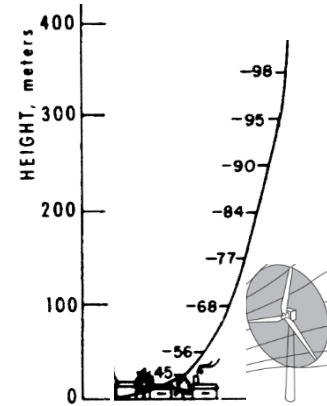


p: prototype



m: model

$$\left(\frac{D}{H}\right)_p = \left(\frac{D'}{H'}\right)_m$$

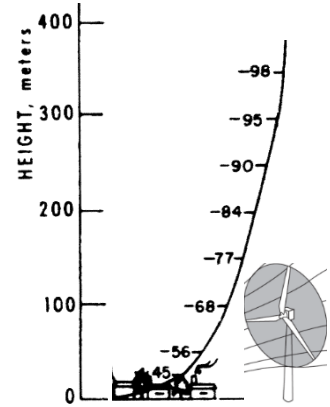


## Geometric Similarity: Similarity of dimensions

### Jensen number

Ratio of structure's length scale to terrain roughness

$$\left(\frac{L_b}{z_0}\right)_m = \left(\frac{L_b}{z_0}\right)_p$$



### Ratio of structure's length scale to gradient height

(large scale motion)

$$\left(\frac{L_b}{z_g}\right)_m = \left(\frac{L_b}{z_g}\right)_p$$

### Ratio of structure's length scale to turbulence length scale

(small scale motion)

$$\left(\frac{L_b}{L_t}\right)_m = \left(\frac{L_b}{L_t}\right)_p$$

## Dynamic Similarity: Similarity of forces

### Reynolds number

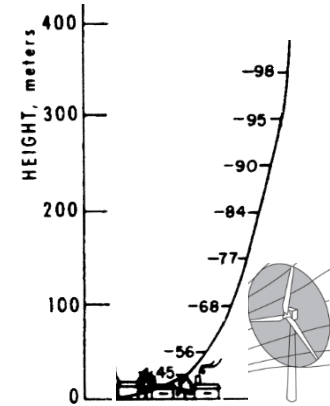
the ratio of inertial to viscous forces

$$Re = \frac{UL\rho}{\mu}$$

### Rosby number

the ratio of inertial to Coriolis forces

$$Ro = \frac{U}{Lf}, \quad f = 2\Omega\sin\varphi$$

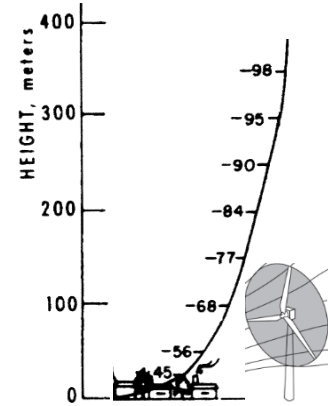


- When modeling the ABL ( $z < 300\text{m}$ ), Rosby numbers are usually large (i.e. Coriolis insignificant)
- Reynolds effects usually neglected  $>$ critical Reynolds number ( $\sim 10^4$  for buildings)
- For geometries with sharp edges (e.g. buildings, wind turbines etc) the flow pattern remains the same  $>$ critical Reynolds number

## Kinematic Similarity:

### Similarity of velocity time and length scales

- This mainly refers to incoming flow (upstream boundary layer)
- Difficult to achieve for both mean velocity and turbulence
- Boundary layers can be considered independent of Re number after a critical surface Re number (surface roughness) has been exceeded  $u_* z_o / \nu > 2.5$



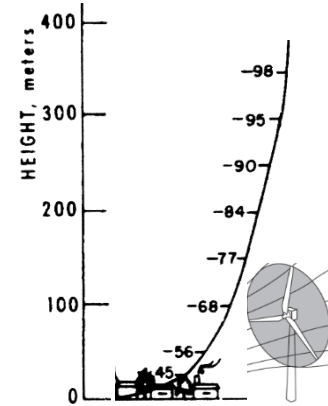
- Vibrations, fluctuations, unsteady motion : 
$$\left(\frac{TV}{L}\right)_m = \left(\frac{TV}{L}\right)_p \rightarrow \left(\frac{T_m}{T_p}\right) = \left(\frac{L_m}{L_p}\right) \left(\frac{V_p}{V_m}\right)$$

e.g.  $T_m = T_p (1/300)(1) = T_p / 300$

## Scaling factor

The most demanding geometric similarity parameter is the ratio of structure's length scale to turbulence length scale since it is a function of height

$$\left(\frac{L_b}{L_t}\right)_m = \left(\frac{L_b}{L_t}\right)_p$$



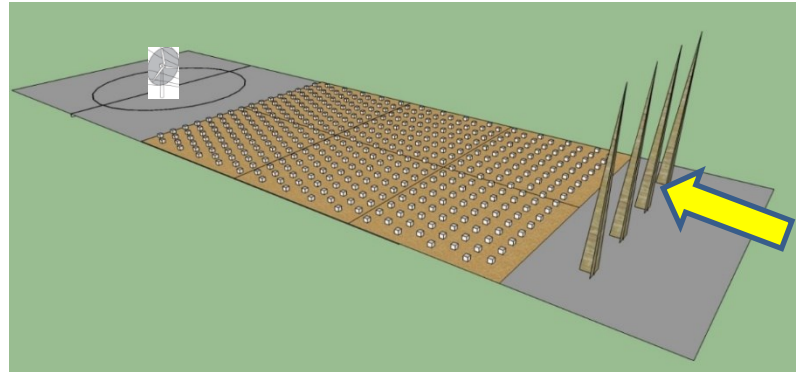
Cooke (1978), proposed using the ESDU profile for  $L_{ux}$  in order to find the scaling factor

$$S = \frac{L_{bp}}{L_{bm}}$$

$$\left. \begin{aligned} L_{X_u} &= 25(z - d)^{0.35} z_0^{-0.063} \\ S L_{X_{uM}} &= 25[S(z - d)_M]^{0.35} [S z_{0M}]^{-0.063} \end{aligned} \right\} S = \frac{91.3(z - d)_M^{0.491}}{L_{X_{uM}}^{1.403} z_{0M}^{0.088}}$$

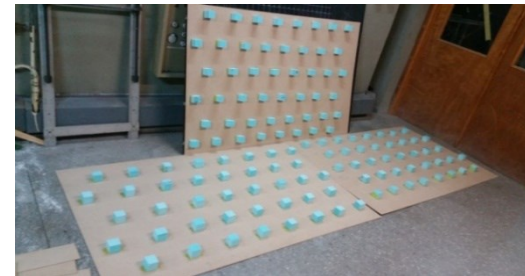
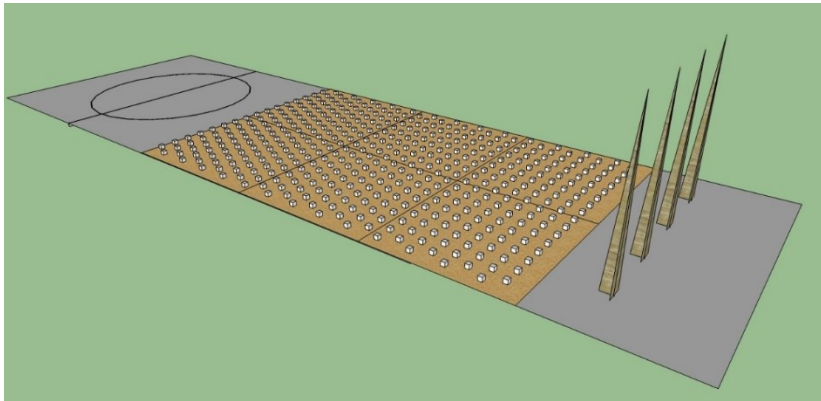
## Boundary Layer Development (augmentation) Devices

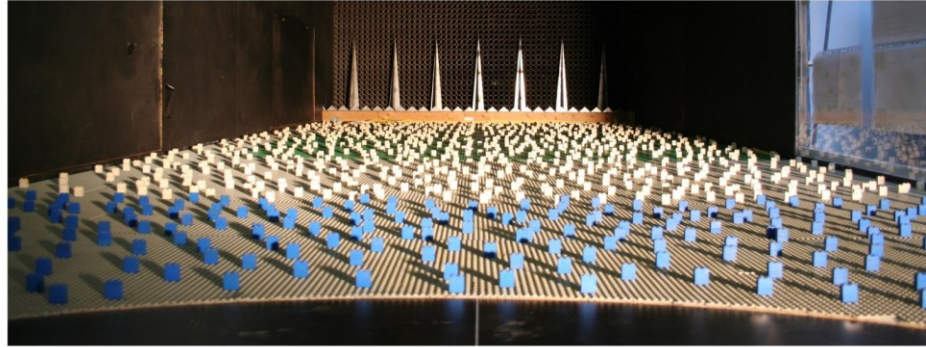
- Even with long wind tunnels, the boundary layer depth may not be sufficient to study models of adequate scale e.g.  $< 1:200$
- Augmentation of the boundary layer height or the use of shorter wind tunnels is possible using special devices:
  - Fences
  - Surface roughness elements
  - Spires





Scale factor	1:300
Spire	$h=170\text{cm}$ ,
$b=30\text{cm}$	
Roughness cubes	height $k=5\text{cm}$
Roughness cubes spacing	$D=20\text{cm}$

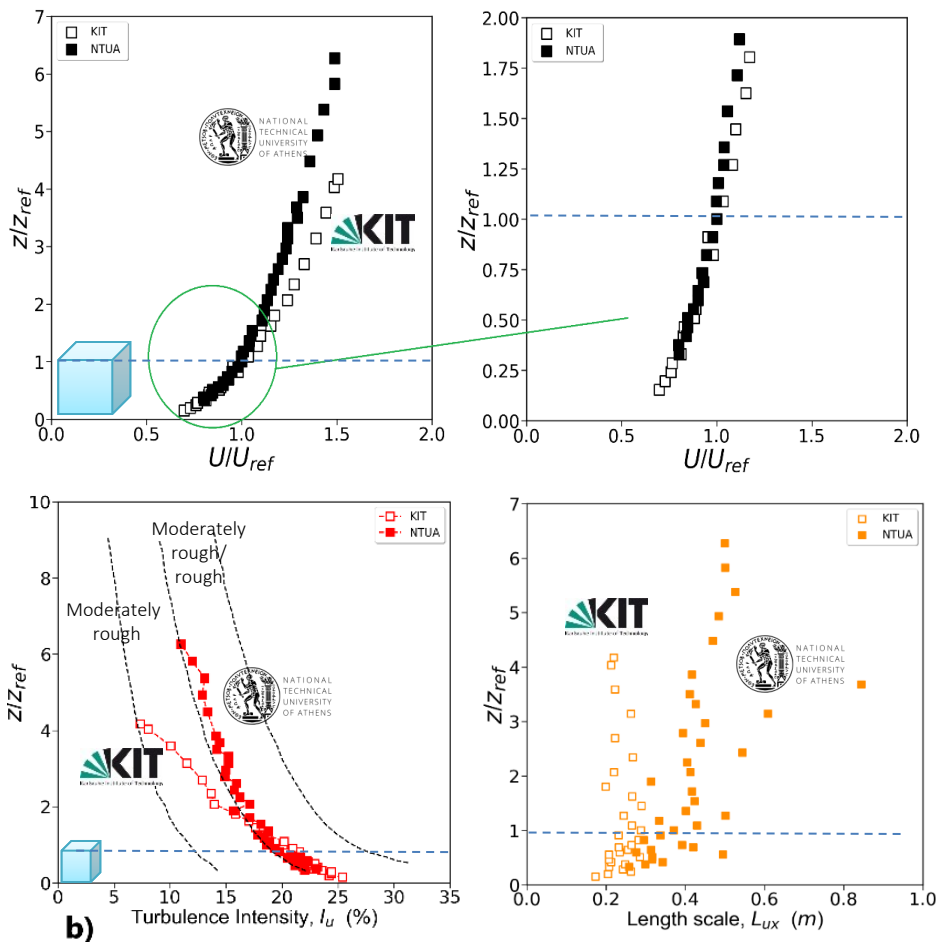




(width × height): 2.0 m × 1.0 m

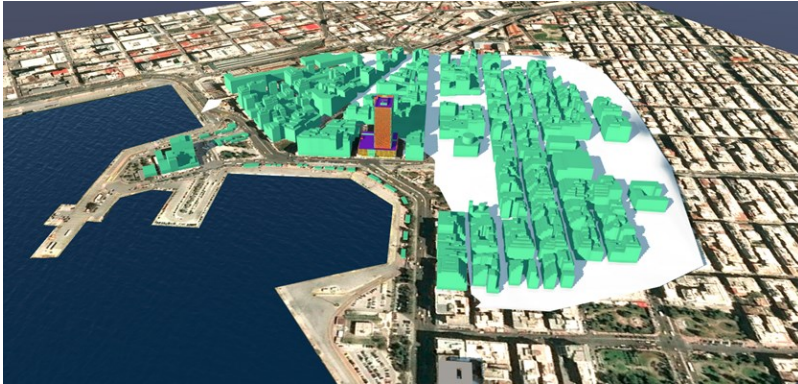


(width × height): 3.5 m × 2.5 m



- Piraeus Tower
- PV panels – static
- PV panels – dynamic
- Trees

An NTUA\_WT study contributed to the design of the new façade for the “Piraeus Tower” building in Piraeus, Greece

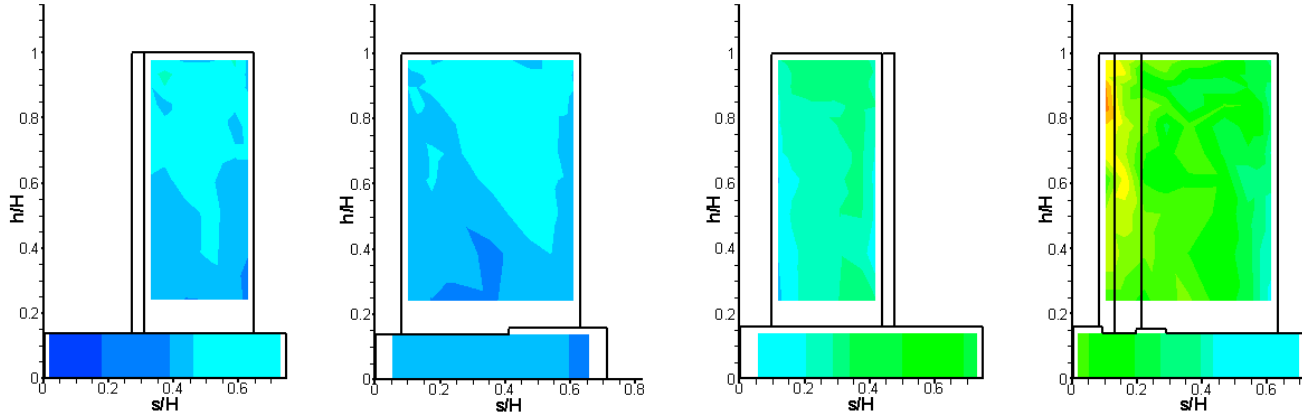
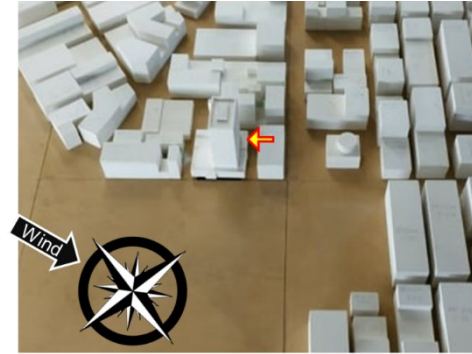
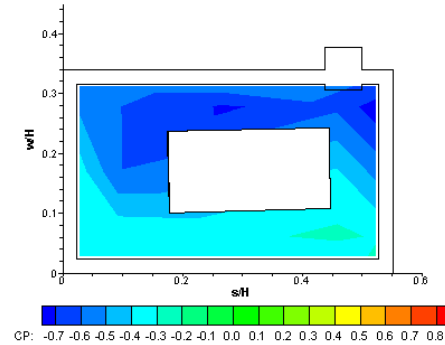




- 3D printing for the building
- 350 m radius around tower
- 1:300 scale for building and wind
- 350 pressure points
- 8 different wind directions
- Urban terrain ABL conditions

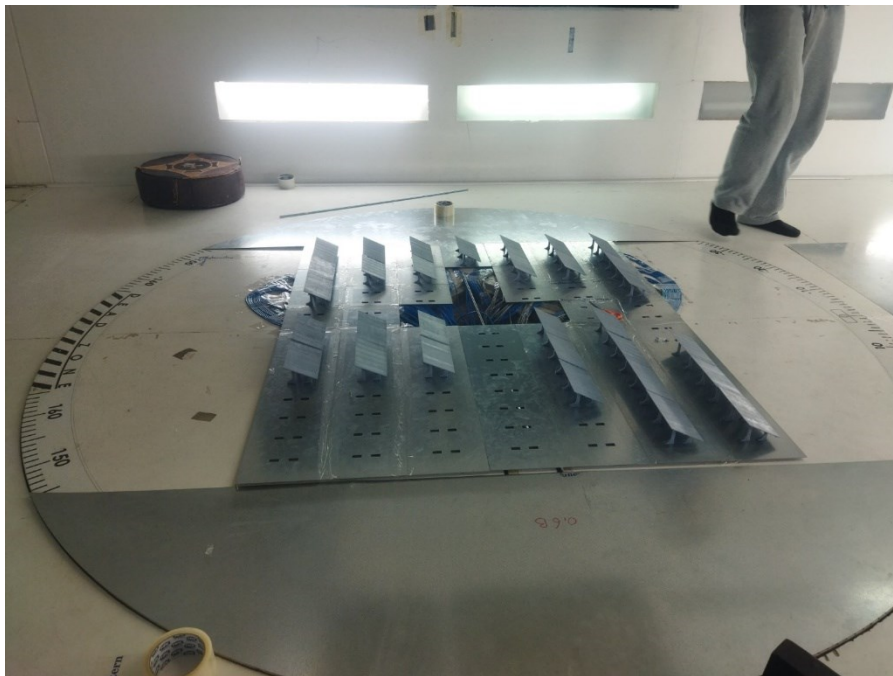


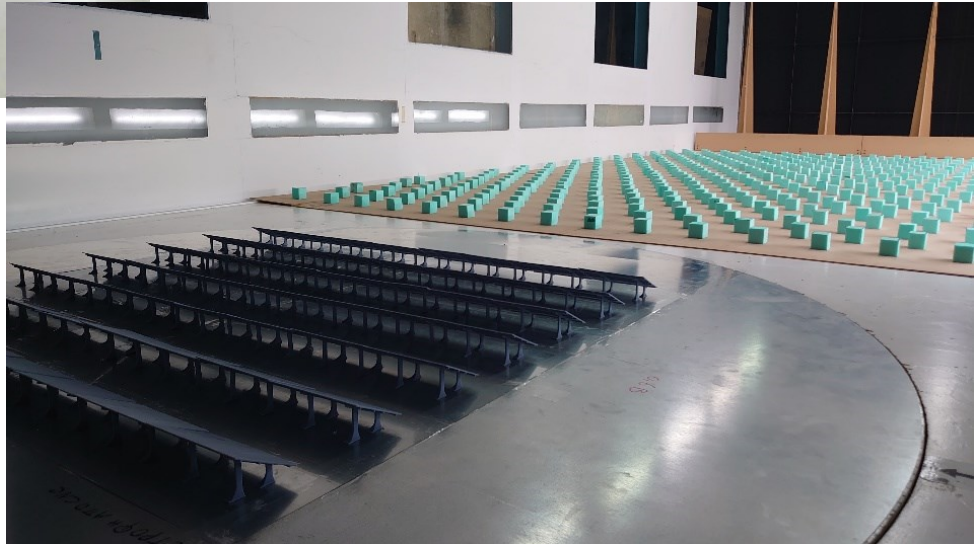
# North NW



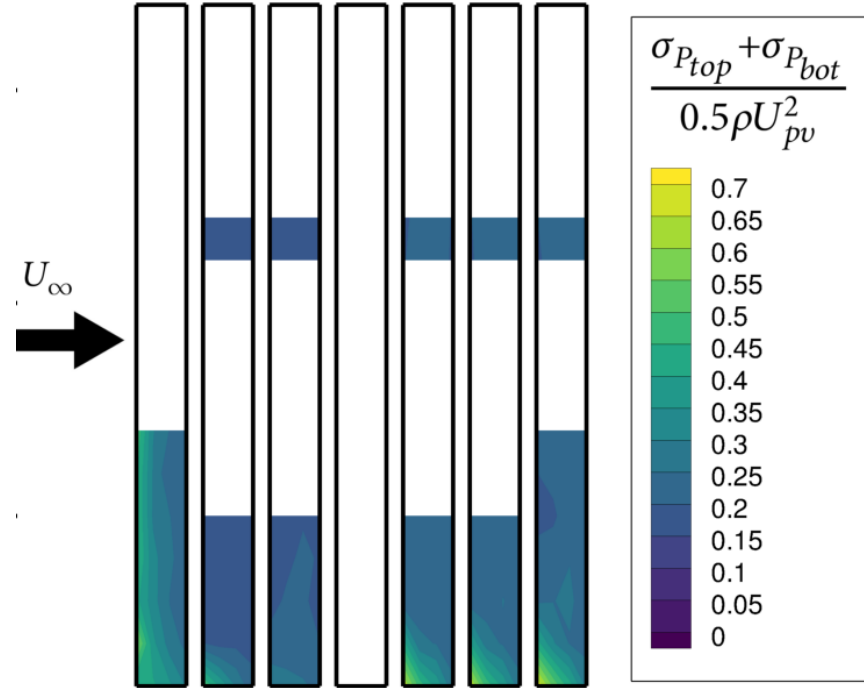
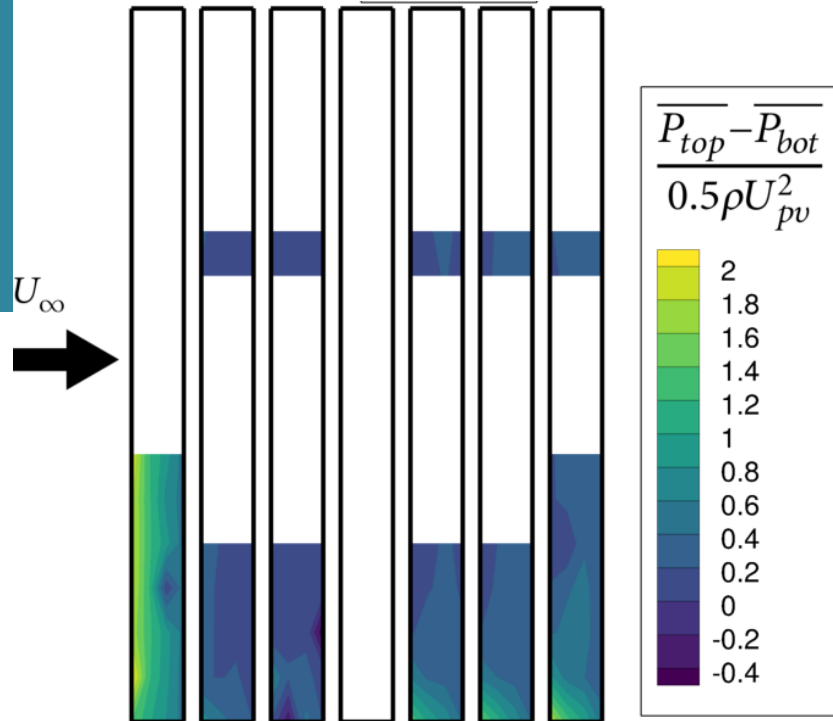


# PV panels – static loads

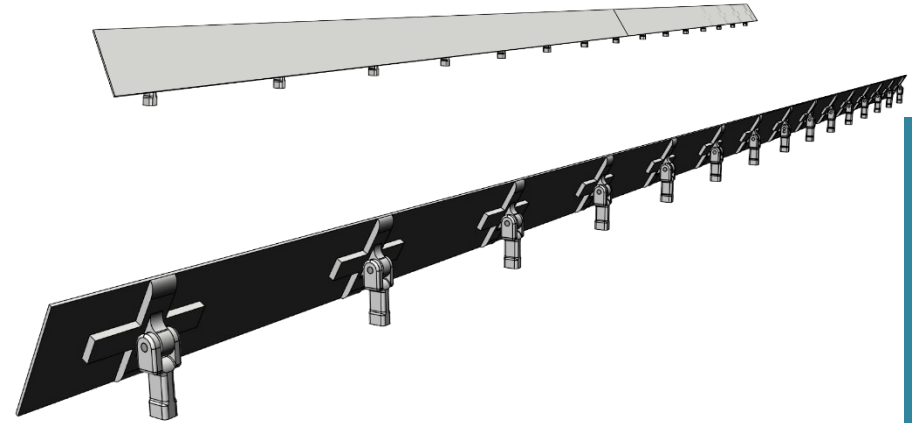
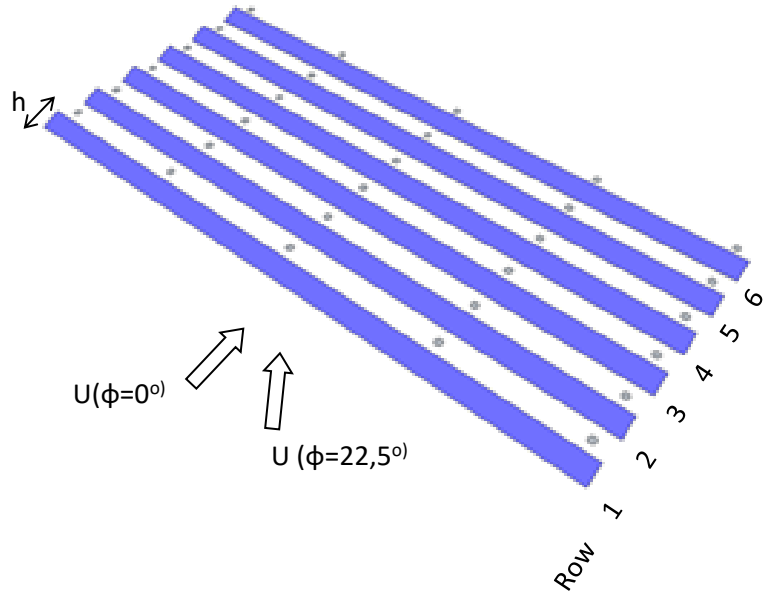


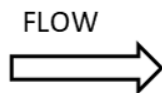
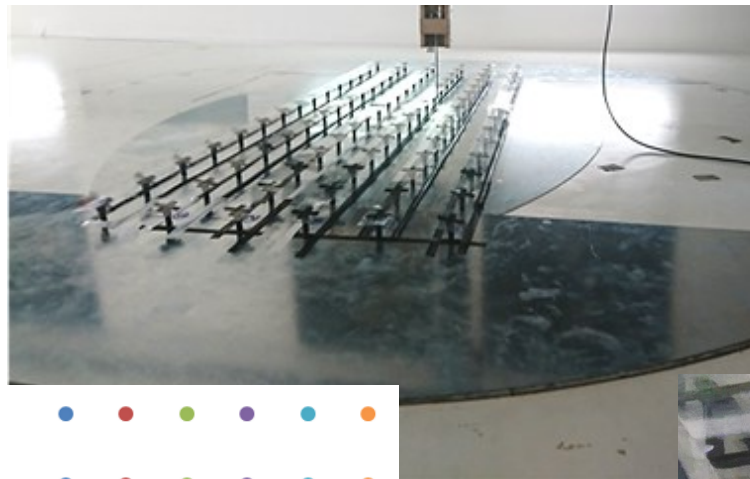


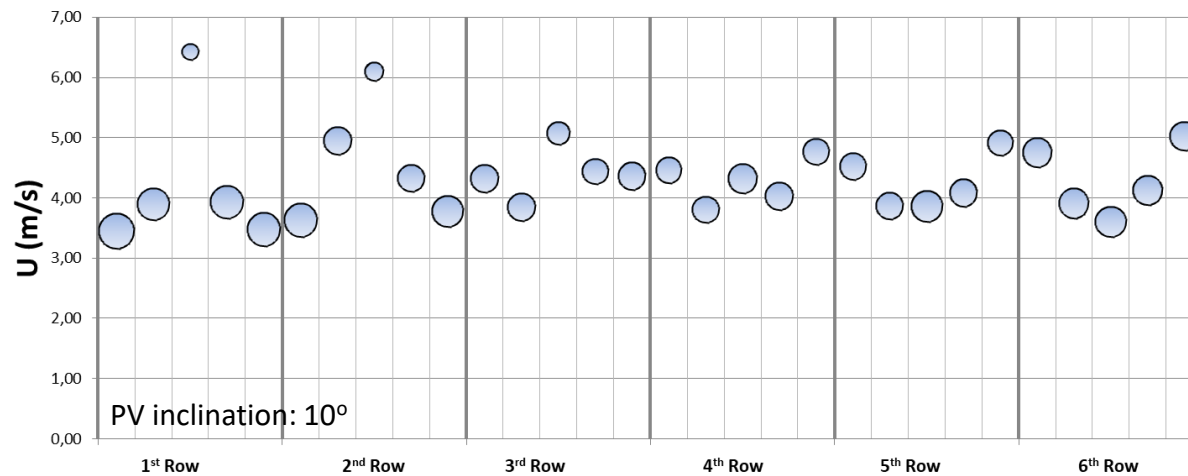
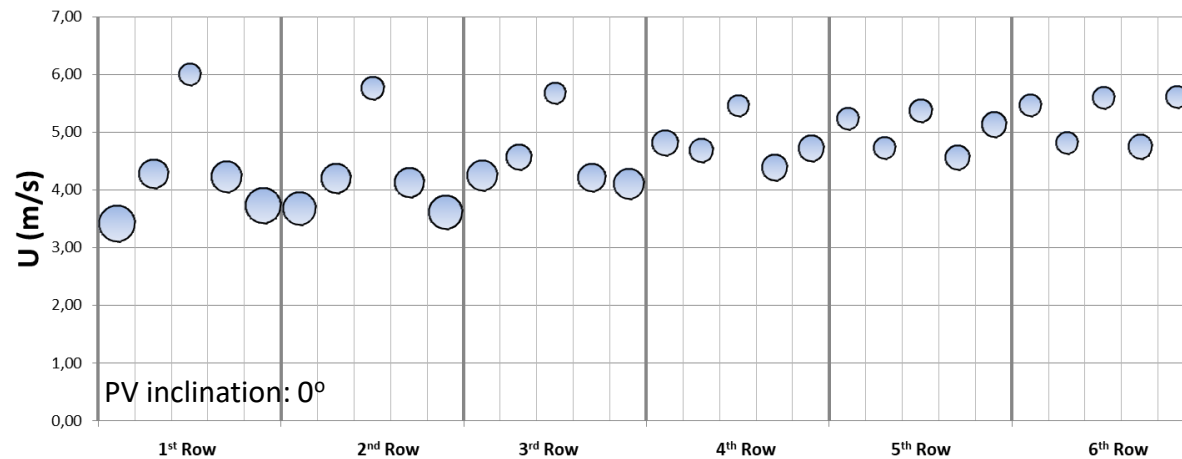
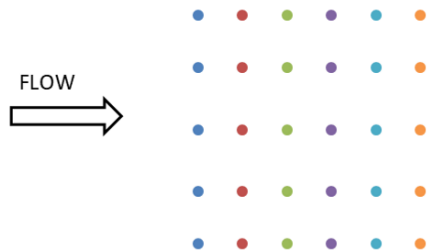
Open terrain ABL conditions















Vegetation-air-buildings interactions,  
affect thermal comfort and indoor natural ventilation.

Heat and mass exchange between  
vegetation elements and the environment,  
depends on trees' aerodynamical, geometrical and thermal properties.



## Methodology - Definition of trees' aerodynamic parameters

### Wind tunnel measurements in the NTUA Test section (2.5m x 3.5m)

- Influence of wind speed on drag coefficient, measured for 6  $U_{ref}$  : 1.5m/s- 10m/s
- For each reference wind speed, **measurement of the drag force** in the direction of wind speed (F)
- For all trees, **definition of the frontal area (A)** through image processing



- Drag coefficient  $C_d$  of the trees for a given reference wind speed  $U_{ref}$  (m/s)

$$C_d = \frac{2F}{\rho U_{ref}^2 A}$$

## Methodology – Definition of trees' aerodynamic parameters

**Citrus sample trees** – common tree type in Greek cities, increased drought tolerance, low irrigation needs - Young trees of different heights, tested: **1.3m, 1.5m, 1.8m**

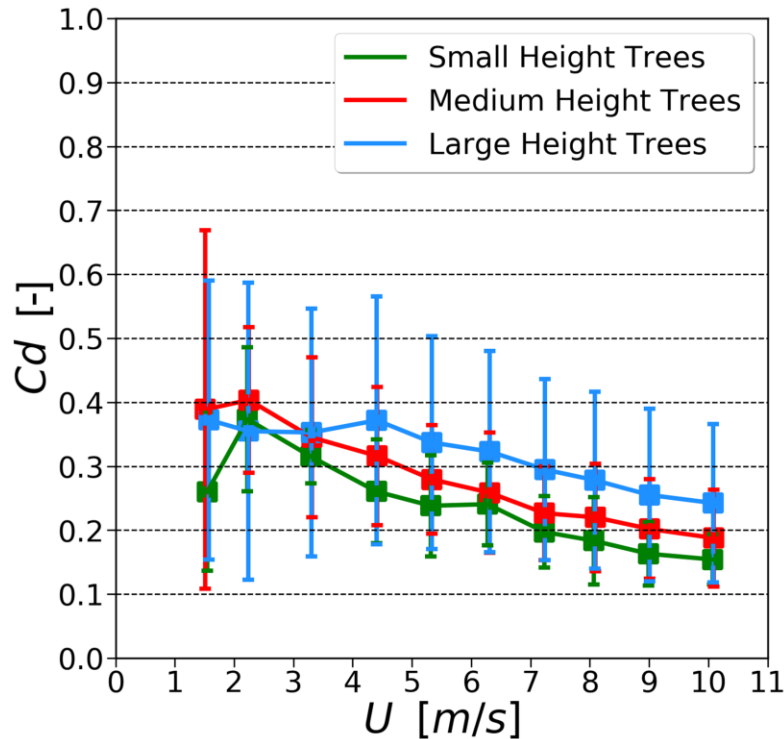
**Wind tunnel measurements** in the NTUA Test section (2.5m x 3.5m)



Kistler  
Multicomponent Force



## Results – Drag coefficient ( $C_d$ )



### ➤ Small height test trees

Mean  $C_d$ =0.15-0.37

St.Dev =0.039-0.12

### ➤ Medium height trees

Mean  $C_d$ =0.18-0.40

St.Dev =0.073-0.28

### ➤ Large height trees

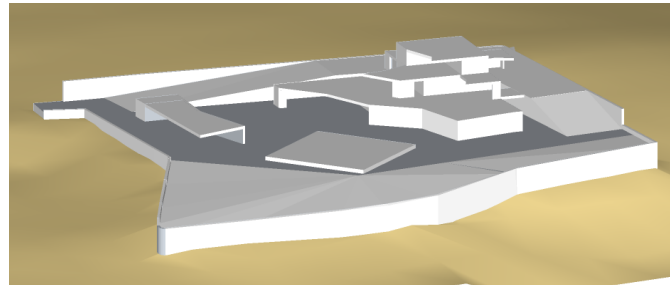
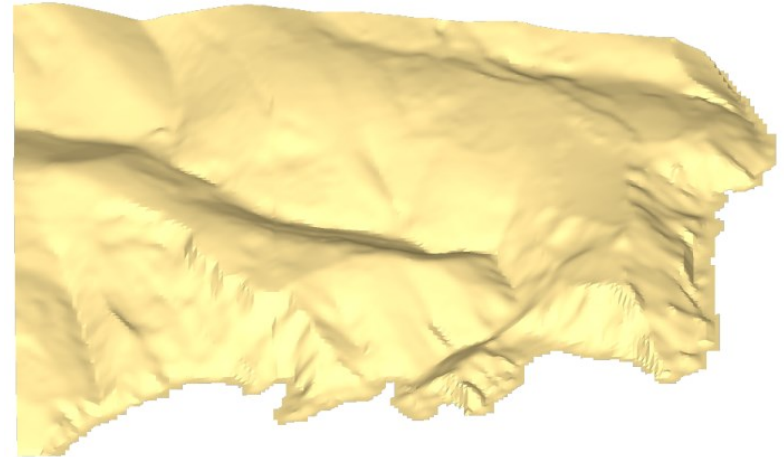
Mean  $C_d$ =0.24-0.37

St.Dev =0.12-0.22

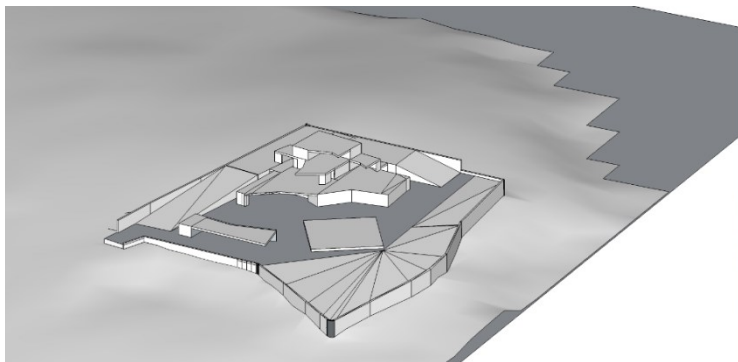
- Wind flow past buildings
  - In complex terrain
  - With vegetation and wind barriers

## Challenges

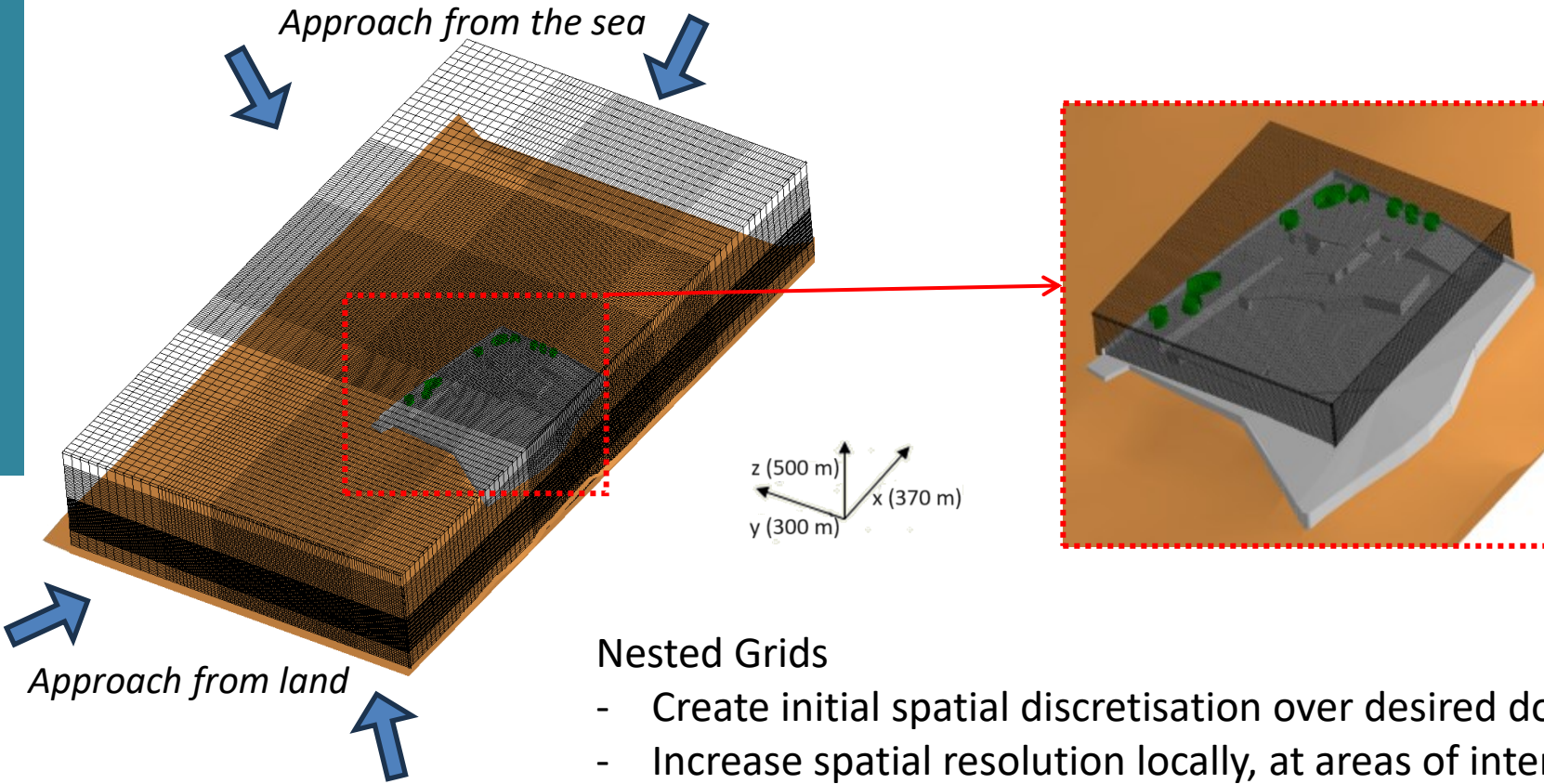
- Representative computational model
- Adequate spatial resolution
- Flexibility in wind direction calculation
- Application of appropriate ABL BCs
- Determine areas of discomfort
- Provide protection from wind





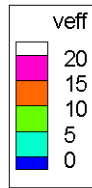
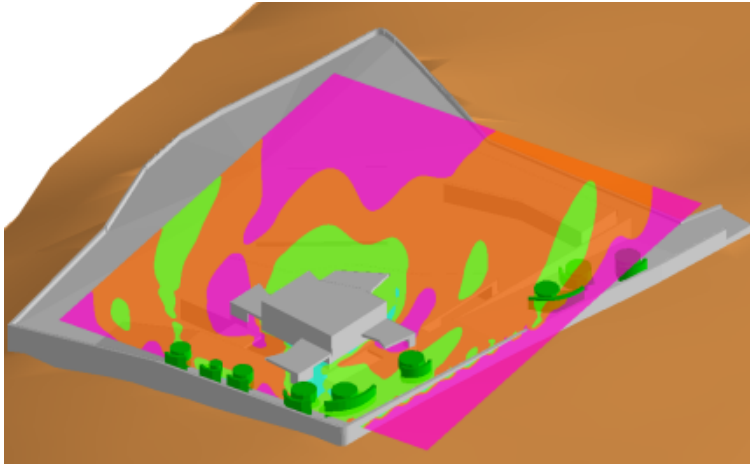




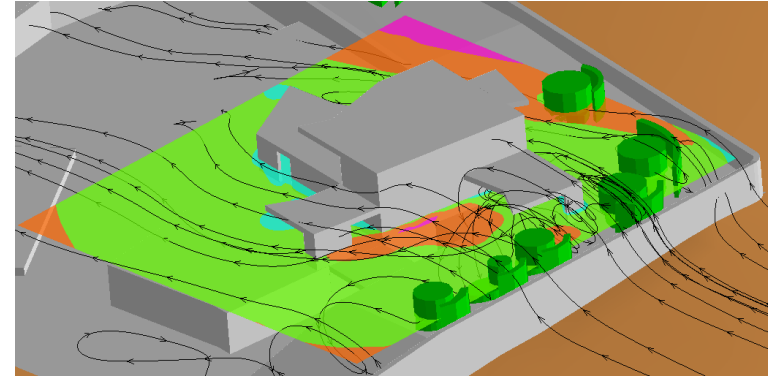
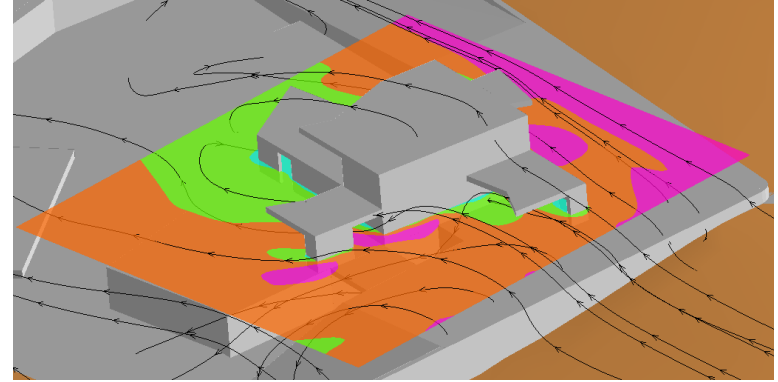


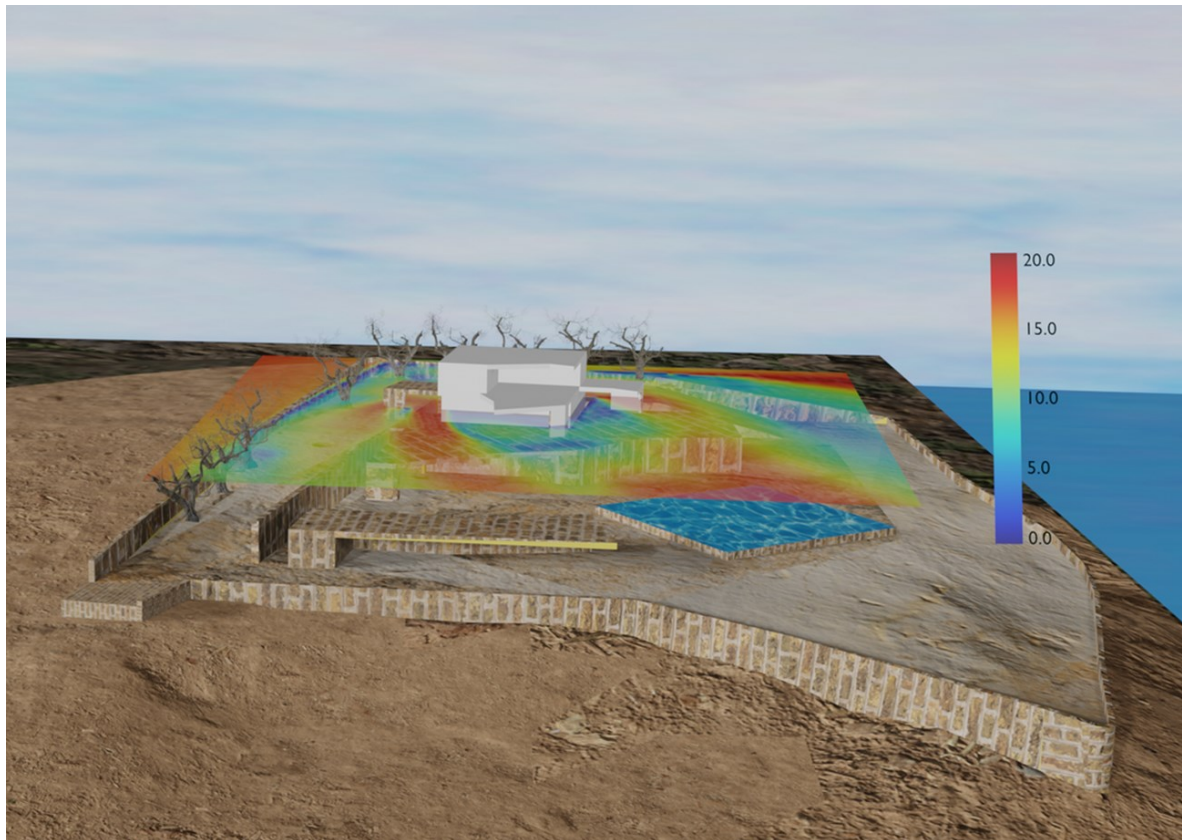
### Nested Grids

- Create initial spatial discretisation over desired domain
- Increase spatial resolution locally, at areas of interest.
- Apply ABL BC according to land type. Mesoscale theory

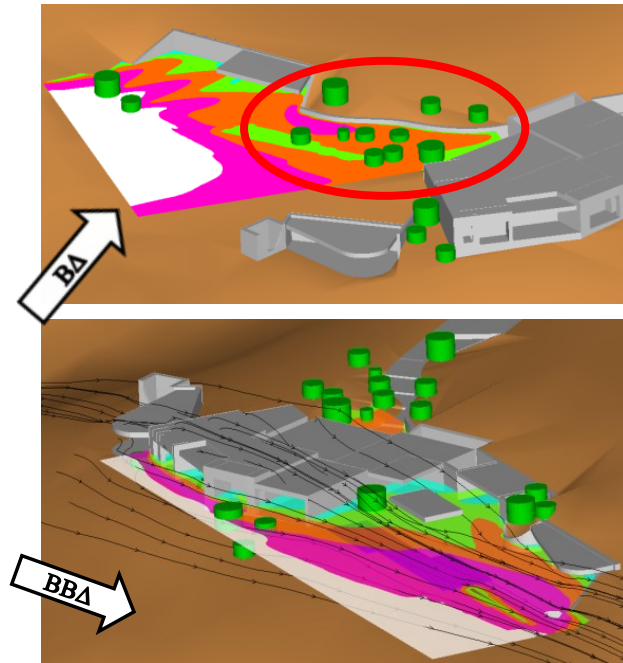


$$v_{eff} = V \times \left(1 + C \times \frac{V_{rms}}{V}\right)$$

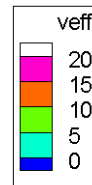
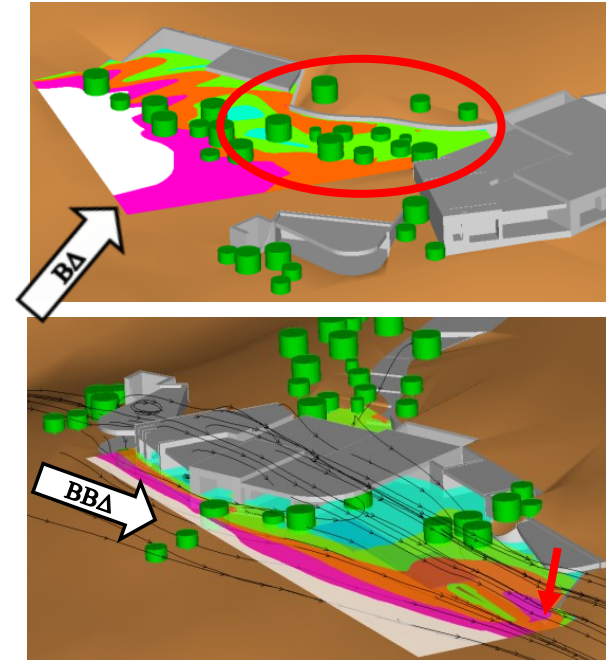




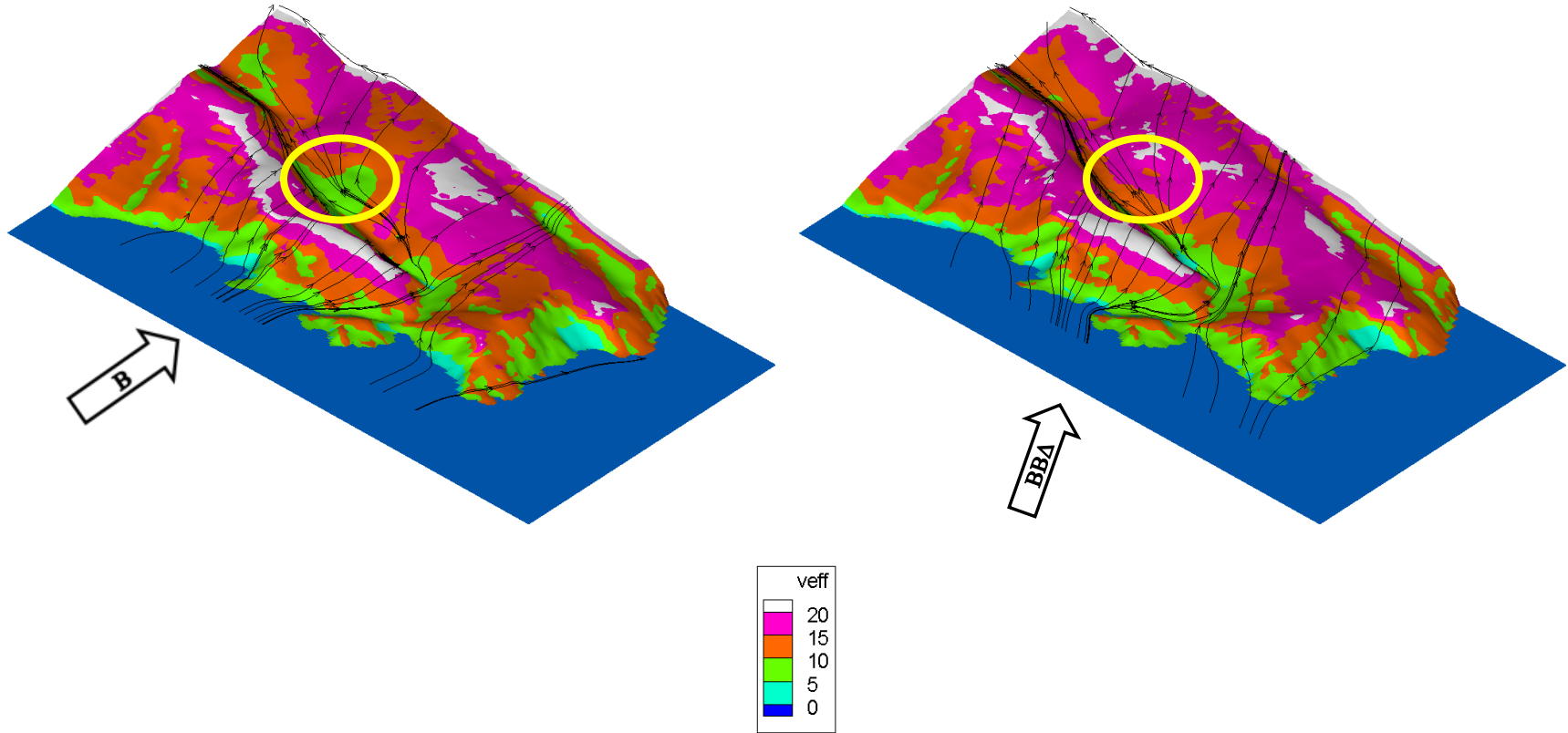
## Initial Design



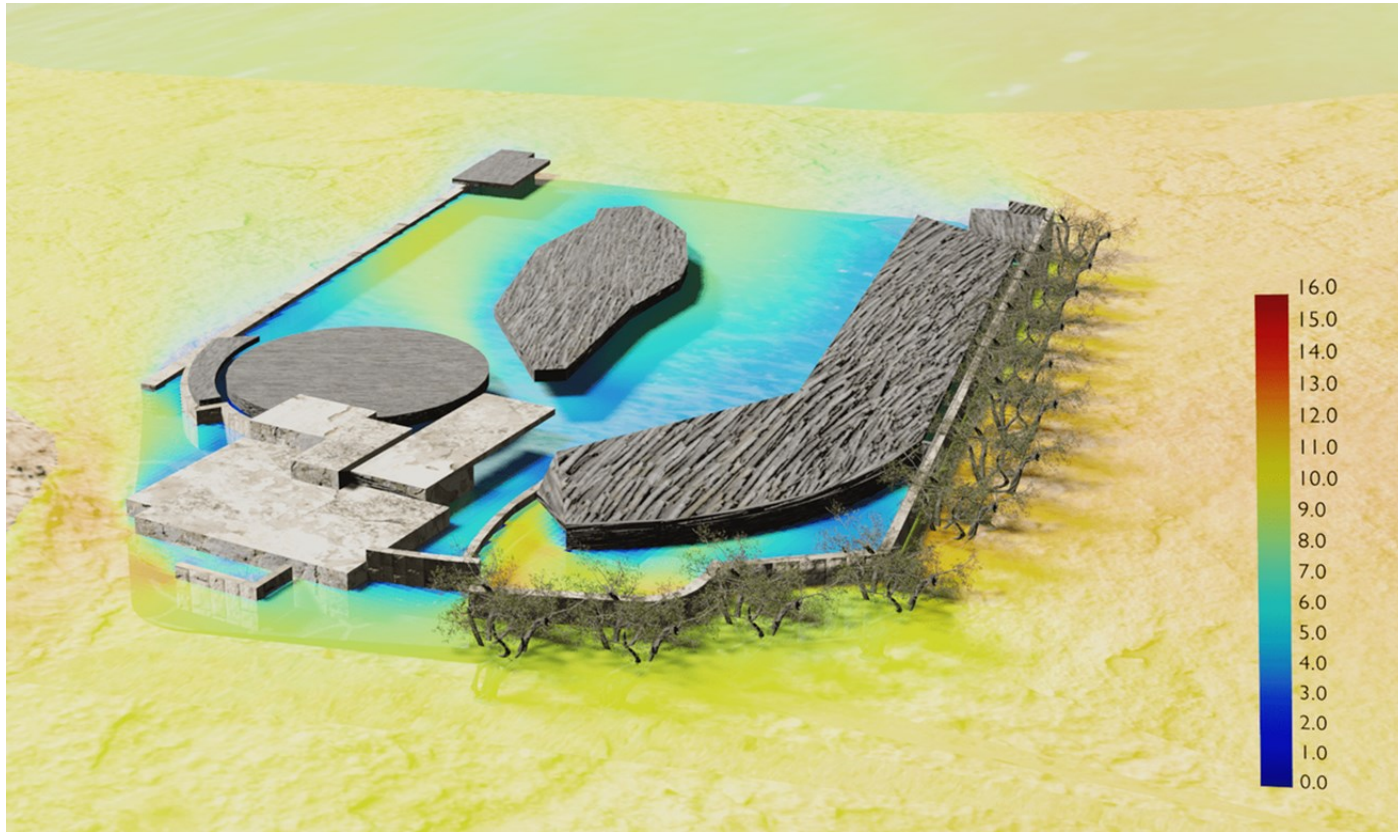
## Final Design



# Effect of wind direction







Thank you !



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